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Predictive Incremental Vector Control for DFIG With Weighted-Dynamic Objective Constraint-Handling Method-PSO Weighting Matrices Design

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ABSTRACT This paper proposes a Particle Swarm Optimization (PSO) based method, the Weighted-Dynamic-Objective Constraint-Handling PSO Method (WDOCHM-PSO). This was used to design the weighting matrices of an incremental Model-Based Predictive Controller (MBPC) for a Doubly Fed Induction Generator (DFIG) applied in a small-scale wind energy system. In contrast to the original PSO, the proposed method has an inner mechanism for dealing with constraints and an adaptive search factor. Additionally, the proposed incremental MBPC implementation does not need the flux information, since the intrinsic integral action rejects the constant flux disturbance. Finally, experimental results show that the proposed controller with the new constraint handling design method is nearly two times faster (In terms of settling time) than other formulations reported in the literature.

INDEX TERMS Doubly-fed induction generator, particle swarm optimization, predictive control, wind energy.

I. INTRODUCTION

During the last decades, there has been a significant effort to reduce the emission of greenhouse gases [1]. As a consequence, wind power penetration has been significantly increased over the last years, and now it represents a major renewable source of energy [2]. Nowadays, Doubly Fed Induction Generators (DFIGs) is one of the most commonly used generators in new turbines [1], and Vector Control (VC) is one of the most popular control approaches for DFIG Rotor Side Converter (RSC) [3]. In summary, this scheme controls active and reactive stator power separately, by decoupling the rotor current d - q synchronous frame [1], [4]. There were proposed several VC schemes for DFIG-based wind turbines, which includes the classical PI [5], Deadbeat

Control [6], [7], Neuro-Fuzzy Control [8], [9], Sliding Mode Control (SMC) [10] and Predictive Controllers [11], [12] with good performance.

The predictive control theory describes a set of controllers that uses the future behavior of the system to decide the best action to reach an objective by using the minimization of a cost function. The estimation of future behavior is determined by a mathematical model combined with actual and past measurements of the system. Predictive control for power converters and motor drives is divided into two categories: finite or continuous control set. Finite control set directly uses as a control input one of the eight possible space vector signals, considering a two level voltage source converter, under optimization process [3], [13], [14], and the continuous control set that uses a modulated control signal [15]. To reach this objective, the DFIG system can be modeled as a space state equation [16], [17], by using a transfer function of the

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system [11], [18], [19], non-linear mathematical models [20] or continuous time models [21].

The non-linear models are useful in many applications [22]–[24], because it implies a better representation of the real system. However, it increases the computational cost, which is critical in real-time systems with a low-cost DSP, especially when it uses high-frequency rates, which is DFIG's case. Therefore, there are some works with non-linear models but limited to simulations [20], [25], [26]. An interesting alternative is the adaptive robust control [27], due to its capability to deal with structured and unstructured uncertainties, especially to deal with fault operation conditions produced by voltage sags.

Even though MBPC minimizes the cost function, there is not an absolute rule that guarantees that the minimal predicted is in fact, a high-quality solution for a real plant, this is owing to the fact that the models are always an approximation of a physical system. Increasing the model complexity could reduce this approximation gap; however, the complex models have a relatively high on-line computational burden, which is critical for drive applications [28]–[30]. A possible solution to overcome the trade-off between the MBPC accuracy and its associated computational cost is to use a simple prediction model in conjunction with additional off-line optimization of a high-fidelity model. In the case of the MBPC, off-line optimization is related to the cost function parameters [31].

The Particle Swarm Optimization (PSO) is considered efficient for complex problems optimization because it lies in a simple programming concept and not require an objective function and/or a model to be differentiable or continuous [32]. Regarding DFIG, PSO was used to design of different controllers as classical PI/PID controllers [33], [34], discrete-time inverse optimal controller [35] and Sliding Mode Controller [36], as an alternative for the classical design.

In spite of the unnumerous advantages of PSO to optimize complex problems, in its original form, this algorithm lacks an inner mechanism for dealing with constraints. In [37], it was proposed the Dynamic-Objective Constraint-Handling Method (DOCHM) PSO, which transforms the original constrained objective function into two unconstrained problems. This method is bi-objective and deals with both unconstrained functions. However, this method uses two search region and it can degrade the convergence speed because the optimization needs to be done twice. In order to increase the convergence speed of DOCHM and overcome the problem of premature convergence, we include the adaptive inertia [38] in DOCHM. So in the beginning, the algorithm can find a high-quality solution fast, then in the later iterations the algorithm increases the search capacity, avoiding premature convergences. The new method is renamed as Weighted-Dynamic-Objective Constraint-Handling Method (WDOCHM).

Using the philosophy of low-cost MPBC algorithm in conjunction with PSO off-line optimization, a novel application of an MBPC with an incremental state-space model and an infinite

control set for DFIG, is proposed in this paper. In other words, there is an online optimization that is a characteristic of predictive control, and another offline optimization that uses a more complex optimization problem. This later optimization can be reduced through the low-cost optimization problem, commonly used in MBPC theory. The prediction model has the advantage to eliminate the flux component under predictions because this component is modeled as a constant disturbance. Here, a novel method is proposed to design weighting matrices using the novel WDOCHM-PSO. Finally, the results obtained in an experimental setup endorse this proposal.

II. DFIG MODEL-BASED PREDICTIVE CONTROL

A. ROTOR SIDE INCREMENTAL MODEL

As explained in [1], [4], [16], DFIG stator flux oriented vector control is a strategy used on RSC to control the stator active and reactive power separately. This strategy uses the information from the stator flux, $\vec{\lambda}_s$, to synchronize the plant space vectors in a stator flux reference frame and it controls rotor current direct and quadrature components. Since $\vec{\lambda}_s^s$ is oriented in its own reference frame, $\lambda_{sd}^s = |\vec{\lambda}_s^s|$ and $\lambda_{sq}^s = 0$ [4], [16]. Here, the subscripts d and q denote direct and quadrature components.

Assuming that stator voltage imposes the flux, $\vec{v}_s^s \approx -j\omega_s \vec{\lambda}_s^s$, making possible to derive the following state-space matrix DFIG model, as is described in [4], [16]:

$$\underbrace{\begin{bmatrix} \frac{di_{rd}^s}{dt} \\ \frac{di_{rq}^s}{dt} \end{bmatrix}}_{\frac{dx}{dt}} = \underbrace{\begin{bmatrix} -\frac{R_r}{\sigma L_r} & \omega_{sl} \\ -\omega_{sl} & -\frac{R_r}{\sigma L_r} \end{bmatrix}}_A \underbrace{\begin{bmatrix} i_{rd}^s \\ i_{rq}^s \end{bmatrix}}_x + \underbrace{\begin{bmatrix} \frac{1}{\sigma L_r} & 0 \\ 0 & \frac{1}{\sigma L_r} \end{bmatrix}}_B \times \underbrace{\begin{bmatrix} v_{rd}^s \\ v_{rq}^s \end{bmatrix}}_u + \underbrace{\begin{bmatrix} 0 \\ -\frac{\omega_{sl} L_m}{\sigma L_s L_r} |\vec{\lambda}_s^s| \end{bmatrix}}_w \quad (1)$$

$$\underbrace{\begin{bmatrix} y_d \\ y_q \end{bmatrix}}_y = \underbrace{\begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}}_C x \quad (2)$$

and the relationship between the components of the rotor current vector and the reactive Q_s and active P_s power is done by:

$$i_{rq}^s = -\frac{2L_s}{3L_m |\vec{v}_s^s|} P_s \quad \text{and} \quad i_{rd}^s = \frac{|\vec{\lambda}_s^s|}{L_m} - \frac{2L_s}{3L_m |\vec{v}_s^s|} Q_s \quad (3)$$

where x , u , y are plant states, inputs and outputs, respectively. Additionally, w represents a disturbance intrinsic to the system, due to the stator flux and the slip speed, s superscript denotes that the variable is oriented in stator flux referential, \vec{i}_r , \vec{v}_r , \vec{v}_s and $\vec{\lambda}_s$ are rotor current, rotor voltage, stator voltage and stator flux space-vectors, respectively. L_r , L_s , L_m represent rotor, stator and magnetizing inductances, and R_r represents rotor resistance. Also, $\omega_{sl} = \omega_s - p\omega_m$ denotes the slip speed, where p is the number of pair of poles, ω_m is

the mechanical shaft speed and ω_s is the grid frequency. And finally, $\sigma = 1 - \frac{L_m^2}{L_s L_r}$ is the total leakage factor.

Because $|\bar{\lambda}_s|$ is approximately constant, \mathbf{w} is considered constant. Moreover, Eq. (4) gives the discrete state-space model using a Zero-Order-Hold (ZOH) with no delay and T sample period [4], [16], [39].

$$\begin{aligned} \mathbf{x}(k+1) &= A_d(\omega_{sl}(k))\mathbf{x}(k) + B_d \mathbf{u}(k) + \mathbf{w}_d(k) \\ \mathbf{y}(k) &= C_d \mathbf{x}(k) \end{aligned} \quad (4)$$

where:

$$\begin{aligned} A_d(\omega_{sl}) &= e^{AT} \approx I + AT = \begin{bmatrix} 1 - \frac{R_r T}{\sigma L_r} & \omega_{sl}(k)T \\ -\omega_{sl}(k)T & 1 - \frac{R_r T}{\sigma L_r} \end{bmatrix} \\ B_d &= \int_0^T e^{A\tau} B d\tau \approx BT = \begin{bmatrix} T & 0 \\ \sigma L_r & T \end{bmatrix} \\ C_d &= C \end{aligned} \quad (5)$$

and $\mathbf{w}_d(k)$ is the discretized version of \mathbf{w} . As the mechanical time constant is much higher than the electrical time constant, ω_{sl} is approximately constant compared to the electrical dynamics. Thus, the DFIG state-space model can be linearized for each sample period in the matrix $A_d(\omega_{sl})$, using the measured slip speed in instant k , that is $\omega_{sl}(k)$ [4], [40].

According to [41], an incremental state-space model is expressed in terms of the state variables changes $\Delta \mathbf{x}(k) = \mathbf{x}(k) - \mathbf{x}(k-1)$ and inputs, $\Delta \mathbf{u}(k) = \mathbf{u}(k) - \mathbf{u}(k-1)$, which is represented by the following augmented state-space model [41], [42]:

$$\begin{aligned} \underbrace{\begin{bmatrix} \Delta \mathbf{x}(k+1) \\ \mathbf{y}(k+1) \end{bmatrix}}_{\xi(k+1)} &= \underbrace{\begin{bmatrix} A_d(\omega_{sl}) & 0 \\ C_d A_d(\omega_{sl}) & I \end{bmatrix}}_{A_a(\omega_{sl})} \underbrace{\begin{bmatrix} \Delta \mathbf{x}(k) \\ \mathbf{y}(k) \end{bmatrix}}_{\xi(k)} \\ &+ \underbrace{\begin{bmatrix} B_d \\ C_d B_d \end{bmatrix}}_{B_a} \Delta \mathbf{u}(k) + \Delta \mathbf{w}_d(k) \\ \mathbf{y}(k) &= \underbrace{\begin{bmatrix} 0 & I \end{bmatrix}}_{C_a} \underbrace{\begin{bmatrix} \Delta \mathbf{x}(k) \\ \mathbf{y}(k) \end{bmatrix}}_{\xi(k)} \end{aligned} \quad (6)$$

where $\xi(k+1)$ is the augmented state variables, and A_a ,¹ B_a , C_a are augmented matrices related to the incremental state-space model. Since the changes in flux disturbance are null, the terms of $\Delta \mathbf{w}_d(k) = \mathbf{w}_d(k) - \mathbf{w}_d(k-1)$ were suppressed in Eq. (6). This is possible due to the flux perturbation is constant, and $\Delta \mathbf{w}_d(k) = \mathbf{w}_d(k) - \mathbf{w}_d(k-1) \approx 0$. The model simplification by removing the flux influence, is one of the advantages of the incremental state-space model proposed here.

¹To simplify the notation we are going to use A_a instead of $A_a(\omega_{sl})$.

B. DFIG INCREMENTAL PREDICTIVE CONTROL

Prediction of the future outputs is possible by advancing and iterating Equation (6). This is shown in Equation (7), and is explained with more detail in [31]:

$$\begin{aligned} \mathbf{y}(k+1) &= C_a A_a \xi(k) + C_a B_a \Delta \mathbf{x}(k) \\ \mathbf{y}(k+2) &= C_a A_a^2 \xi(k) + C_a A_a B_a \Delta \mathbf{x}(k) \\ &\quad + C_a B_a \Delta \mathbf{x}(k+1) \\ &\vdots \end{aligned} \quad (7)$$

Finally, repeating and rewriting this process many times, all the predicted outputs are obtained in a compact matrix format, that is:

$$\mathcal{Y} = \mathcal{A} \xi(k) + \mathcal{B} \Delta \mathcal{U} \quad (8)$$

where:

$$\begin{aligned} \mathcal{Y} &= [\mathbf{y}(k+1) \quad \mathbf{y}(k+2) \quad \dots \quad \mathbf{y}(k+N_y)]^T \\ \Delta \mathcal{U} &= [\Delta \mathbf{u}(k) \quad \Delta \mathbf{u}(k+1) \quad \dots \quad \Delta \mathbf{u}(k+N_u-1)]^T \\ \mathcal{A} &= [C_a A_a \quad C_a A_a^2 \quad \dots \quad C_a A_a^{N_y}]^T \\ \mathcal{B} &= \begin{bmatrix} C_a B_a & 0 & \dots & 0 \\ C_a A_a B_a & C_a B_a & \dots & 0 \\ \vdots & \vdots & \ddots & \dots \\ C_a A_a^{N_y-1} B_a & C_a A_a^{N_y-2} B_a & \dots & C_a A_a^{N_y-N_u} B_a \end{bmatrix} \end{aligned} \quad (9)$$

In Eq. (8) and Eq. (9), $\Delta \mathcal{U}$ and \mathcal{Y} represent all predicted input increments and outputs, respectively. \mathcal{A} and \mathcal{B} are matrices that in turn contain matrices A_a , B_a and C_a . Furthermore, N_y is considered as the prediction whereas N_u is the control horizon, respectively.

Initially, the incremental prediction model is linearized considering the current ω_{sl} . Then, this model is used to calculate a quadratic cost function, given by Eq. (10). This is described in [31], [43] and [44]. Afterward, the permanence index is minimized and the corresponding control law can be obtained, as follows:

$$\begin{aligned} J &= \sum_{i=1}^{N_y} \mathbf{E}(k+i)^T W_y \mathbf{E}(k+i) \\ &\quad + \sum_{j=0}^{N_u-1} \Delta \mathbf{u}(k+j)^T W_u \Delta \mathbf{u}(k+j) \end{aligned} \quad (10)$$

Here, $\mathbf{E}(k+i) = \mathbf{y}(k+i) - \mathbf{y}_{\text{ref}}(k+i)$ represents the predicted errors, $W_y = \begin{bmatrix} W_{y,11} & W_{y,12} \\ W_{y,21} & W_{y,22} \end{bmatrix}$ and $W_u = \begin{bmatrix} W_{u,11} & W_{u,12} \\ W_{u,21} & W_{u,22} \end{bmatrix}$ are weighting matrices related to the predicted errors and predicted inputs, $\mathbf{y}_{\text{ref}}(k+i)$ are the future references, which are considered constant, during all the prediction period. The input applied by the controller is the minimal solution of Eq. (10) or, in an analytical point of view, when $\nabla_{\Delta \mathcal{U}} J = 0$. The result of this minimization is the following control

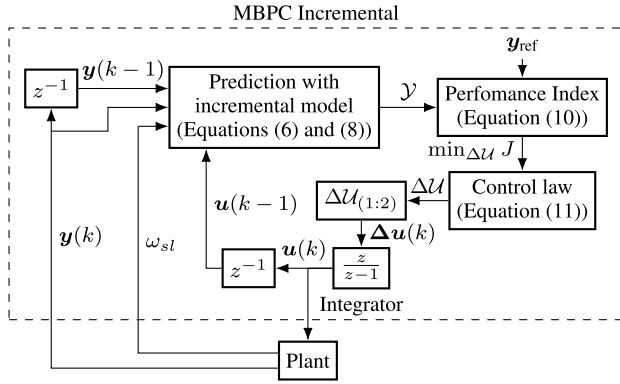


FIGURE 1. MBPC Incremental diagram.

law $\Delta \mathcal{U}$, as is depicted in [43]:

$$\Delta \mathcal{U} = (\mathcal{B}^T \mathcal{W}_y \mathcal{B} + \mathcal{W}_u)^{-1} \mathcal{B}^T \mathcal{W}_y (\mathcal{Y}_{\text{ref}} - \mathcal{A} \xi(k)) \quad (11)$$

where \mathcal{W}_y , \mathcal{W}_u are main diagonal repetition of matrices W_y and W_u , and

$$\begin{aligned} \mathcal{Y}_{\text{ref}} &= [\mathbf{y}_{\text{ref}}(k+1) \quad \dots \quad \mathbf{y}_{\text{ref}}(k+N_y)]^T \\ &= \mathbf{y}_{\text{ref}}(k+1) [1 \quad 1 \quad \dots \quad 1]^T \end{aligned} \quad (12)$$

are the outputs references, respectively.

The MBPC Incremental block diagram presented in Fig. 1 summarize the proposed incremental method.

In Fig. 1 just the control action related to the next sample time is applied to the plant. It means that only the first two elements of $\Delta \mathcal{U}$, or the $\Delta \mathcal{U}_{(1:2)}$, in junction with $\Delta \mathbf{u}(k) = [\Delta u_1 \quad \Delta u_2]^T$, are currently used. After this, the output increment is integrated before it goes to the plant, giving as result $u(k)$.

Concluding, the block diagram of the complete control strategy is shown in Fig. 2. In this figure, the abc superscript represent three-phases variables, $P_{s,\text{ref}}$ and $Q_{s,\text{ref}}$ are the active and reactive power references. In addition, a Space-Vector Pulsed-width Modulator (SVPWM) is applied to modulated voltage in the RSC, as is depicted in [4].

C. MBPC INCREMENTAL CLOSED-LOOP ANALYSIS

This sub-section deals with the controller-plant closed-loop transfer matrix equivalent and thus analyze the controller behavior under lower frequencies, constant errors, and/or disturbances.

Firstly, Eq. (11) is rewritten in terms of the gain $\mathcal{K} = (\mathcal{B}^T \mathcal{W}_y \mathcal{B} + \mathcal{W}_u)^{-1} \mathcal{B}^T \mathcal{W}_y$:

$$\Delta \mathcal{U} = \mathcal{K} (\mathcal{Y}_{\text{ref}} - \mathcal{A} \xi(k)) \quad (13)$$

Sequentially, the control law is defined as follows:

$$\Delta \mathbf{u} = \tilde{\mathcal{K}} (\mathcal{Y}_{\text{ref}} - \mathcal{A} \xi(k)) \quad (14)$$

where

$$\tilde{\mathcal{K}} = [\mathcal{K}_{(1)} \quad \dots \quad \mathcal{K}_{(n_y)}] \quad (15)$$

is the first two lines of \mathcal{K} . Also, in Equation (15), $\mathcal{K}_{(i)} \in \mathbb{R}^{2 \times 2}$ are square matrices.

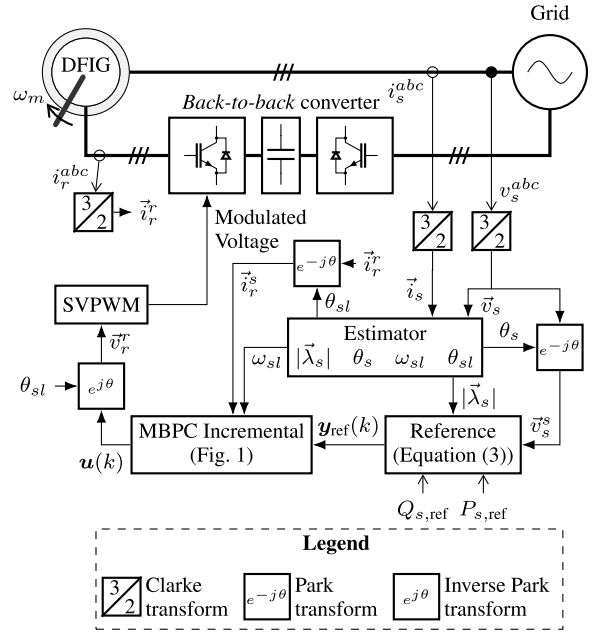


FIGURE 2. DFIG rotor side converter control diagram.

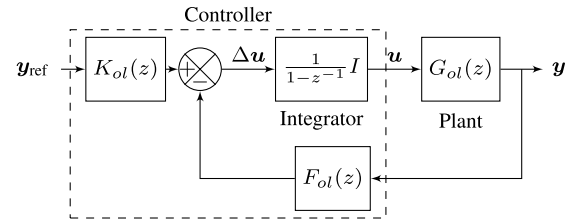


FIGURE 3. Block diagram of (16) and (18).

Alternatively:

$$\Delta \mathbf{u} = K_{ol}(z) \mathbf{y}_{\text{ref}} - F_{ol}(z) \mathbf{y} \quad (16)$$

where:

$$\begin{aligned} F_{ol}(z) &= \sum_{i=1}^{n_y} \mathcal{K}_{(i)} \left(I + \sum_{j=0}^{i-1} A_d^{i-j} - z^{-1} \sum_{j=0}^{i-1} A_d^{i-j} \right) \\ K_{ol}(z) &= \sum_{i=1}^{n_y} \mathcal{K}_{(i)} \end{aligned} \quad (17)$$

and the plant transfer matrix is calculated using:

$$G_{ol} = (zI - A_d) B_d^{-1} + G_d \quad (18)$$

where:

$$G_d = \begin{bmatrix} 0 & 0 \\ -\frac{\omega_{sl} L_m}{\sigma L_s L_r} T |\bar{\lambda}_s| & 0 \end{bmatrix} \quad (19)$$

Also, in Fig. 3 the equivalent diagram of Eq. (16) and Eq. (18) are shown.

The closed loop transfer matrix of Fig. 3 is presented in Eq. (20).

$$G_{cl} = K_{ol} I_{ol} G_{ol} (I + F_{ol} I_{ol} G_{ol})^{-1} \quad (20)$$

where $I_{ol}(z) = \frac{1}{1-z^{-1}} I$ is an integrator.

The presence of the integrator in the inner loop of Fig. 3 and in Eq. (20) indicates that the controller has a pole in the unitary circle, which means that the controller has a very high gain for the error $\Delta \mathbf{u}$, avoiding steady-state errors caused by low frequency or constant disturbances, as the flux component [45].

III. DESIGN OF WEIGHTING MATRICES USING PSO

PSO is inspired by the social and cooperative behavior presented in several species, as birds and fishes. This algorithm tries to minimize a fitness function, and each particle represents a potential solution to the optimization problem. During each PSO iteration, particles move in the direction of the optimal solution [46].

A. OPTIMIZATION PROBLEM AND FITNESS FUNCTION

The system works inside the linear region. In this situation, the operation point is proportional to the slip speed variation and the stator flux. Furthermore, the controller linearizes at each iteration, by inputting current slip speed. So, to simplify this idea, it can be noticed that the controller uses the constant shaft speed and the step response, as input references.

The Integral Absolute Magnitude of Error (ITAE) of both direct, $ITAE_d$, and quadrature components, $ITAE_q$, was used to bench-marking the current responses. The performance index is commonly used as a guideline to tuning the controller parameters [33], [47]. This criterion has some interesting advantages, it can be mentioned, among others, that is more selective, produce less overshoot and oscillation than Integral Square Error (ISE) and Integral of Absolute Error (IAE) and in accord with criteria described in [48]–[50], it can be defined as follows:

$$ITAE_d = \int_{t_{step_d}=1}^{t_{step_d}+0.49} (t - t_{step_d}) |E_d(t)| dt + \int_{t_{step_q}=1.5}^{t_{step_q}+0.49} (t - t_{step_q}) |E_d(t)| dt \quad (21)$$

$$ITAE_q = \int_{t_{step_q}=1.5}^{t_{step_q}+0.49} (t - t_{step_q}) |E_q(t)| dt + \int_{t_{step_d}=1}^{t_{step_d}+0.49} (t - t_{step_d}) |E_q(t)| dt \quad (22)$$

where $t_{step_d} = 1$ s and $t_{step_q} = 1.5$ s are the instants of time where the step function is applied, under direct and quadrature components. $E_d(t) = i_{rd,ref}^s(t) - i_{rd}^s(t)$, $E_q(t) = i_{rq,ref}^s(t) - i_{rq}^s(t)$ are the errors of the direct and quadrature components.

As constraints of the optimization problem, the direct, $M_{p,d} < 35\%$, the quadrature, $M_{p,q} < 35\%$, the current

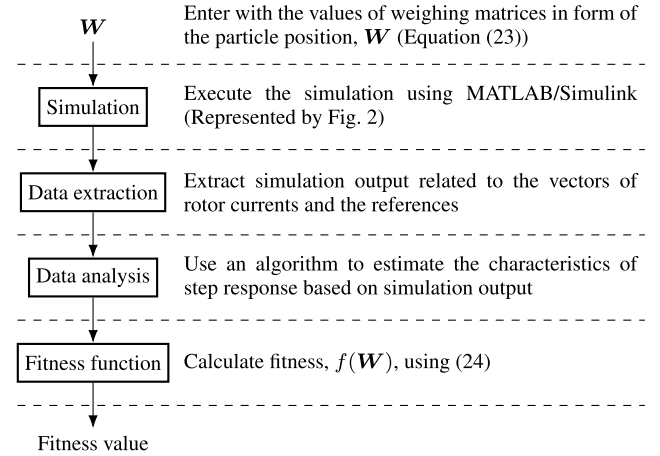


FIGURE 4. Step by step of fitness function.

overshoots, the direct, $t_{ss,d} < 3$ ms, the quadrature, $t_{ss,q} < 3$ ms and the setting time, were used.

Each particle position, or the variable to be optimized, W , is a vector that includes the elements of MBPC weighting matrices:

$$W = [W_{y,11}, W_{y,12}, \dots, W_{u,21}, W_{u,22}]^T = [W_1, W_2, \dots, W_8]^T \quad (23)$$

Hence, the optimization problem, or the guidelines for the controller tuning, is defined as:

$$\begin{aligned} & \text{minimize } f(W) = \max(\{ITAE_d, ITAE_q\}) \\ & \text{subject to } M_{p,d} \text{ and } M_{p,q} < 35\% \\ & \quad t_{ss,d} \text{ and } t_{ss,q} < 3 \text{ ms} \end{aligned} \quad (24)$$

where \mathbb{S} is the search space, or, in other words, the objective function domain.

In short, the algorithm of fitness function runs the model described in Fig. 2, using a simulation of electrical system, and it analyzes the step response for references of direct, $i_{rd,ref}^s$ and quadrature, $i_{rq,ref}^s$, axes of rotor current. The algorithm of the fitness function is represented in Fig. 4.

Firstly, the fitness algorithm receives as parameter the weighting matrices elements, or particle position, W (please see Eq. 23). Secondly, the algorithm executes the simulation using the W values. After the simulation finishes, the vectors for currents references and rotor currents are used as simulation outputs. Analyzing these data, the characteristics of the step response, including ITAE, overshoot, and settling time, must be calculated. Finally, with step response parameters, the fitness value is calculated.

B. PARTICLES MOTION

On each iteration, particles moves using a combination of the best personal experience, denoted by subscript 'pbest', the best global experience, denoted by the subscript, 'gbest', and its actual velocity, $W_{vel}(k)$, as is explained in [46], [51]–[53]. Thus, next velocity component, $W_{vel,i}$, of

current particle, j , is defined as follows:

$$\begin{aligned}
 W_{\text{vel},i}^j(k+1) &= \underbrace{w_{\text{PSO}}(k) \cdot W_{\text{vel},i}^j(k)}_{\text{Actual velocity component}} \\
 &+ \underbrace{c_p \cdot \text{rand}([1, 0]) \left(W_{\text{pbest},i}^j(k) - W_i^j(k) \right)}_{\text{Best personal experience component}} \\
 &+ \underbrace{c_g \cdot \text{rand}([1, 0]) \left(W_{\text{gbest},i}^j(k) - W_i^j(k) \right)}_{\text{Best global experience component}} \quad (25)
 \end{aligned}$$

and next position component is given by:

$$W_i^j(k+1) = W_i^j(k) + W_{\text{vel},i}^j(k+1) \quad (26)$$

In Equation (25), $\text{rand}([1, 0])$ represents a random uniform number between 0 and 1, $c_g = 1.5$ and $c_p = 1.5$ are the social and cognitive acceleration factors, and $w_{\text{PSO}}(k)$ is the actual adaptive inertia proposed by [38].

The time-varying inertia $w_{\text{PSO}}(k)$ is focused on the exploration when each particle's best personal position is near to another particle, otherwise, its inertia decreases, reversing the trend to the minimization. Thus, the actual inertia is given by:

$$w_{\text{PSO}}(k) = 0.9 - 0.4 \frac{d(k)}{\max(\{d(k)\}_{1 \leq k \leq k_{\max}})} \quad (27)$$

Here k_{\max} is the maximum number of iterations and $d(k)$ is the maximum standard deviation between W_i^j , where j represents each particle of the population.

$$d(k) = \max \left(\left\{ \text{std} \left(\left\{ W_{\text{pbest},i}^j \right\}_{1 \leq j \leq P_{\text{size}}} \right) \right\}_{1 \leq i \leq 8} \right) \quad (28)$$

where P_{size} is the population size and i refers to each W_{pbest}^j component.

Finally, to clamp down all particle within \mathbb{S} search space a simple method presented in [37] is used. \mathbb{S} is limited to the lower bond $[150, -300, -300, 150, 10^{-9}, -10^{-1}, -10^{-1}, 10^{-9}]^T$ and to the upper bound $[10^4, 300, 300, 10^4, 1, 10^{-1}, 10^{-1}, 1]^T$. Basically, if a certain particle, with position W^j , goes outside of the function domain, its violated position component W_i^j is recalculated once again using Equation (29).

$$W_i^j = \frac{W_{\text{pbest},i}^j + W_{\text{gbest},i}^j + W_{\text{pbest},i}^{r1} + W_{\text{pbest},i}^{r2}}{4} \quad (29)$$

superscripts $r1$ and $r2$ represent two randomly selected particles within swarm population.

C. DYNAMIC-OBJECTIVE CONSTRAINT-HANDLING METHOD

The essence of DOCHM is to divide the constrained optimization problem into two unconstrained objectives. The primary objective function, $\Phi(W)$, is related constraints:

$$\begin{aligned}
 \phi(W) &= \max(\{0, M_{p,d} - 35\}) + \max(\{0, M_{p,q} - 35\}) \\
 &+ \max(\{0, t_{ss,d} - 3\}) + \max(\{0, t_{ss,q} - 3\}) \quad (30)
 \end{aligned}$$

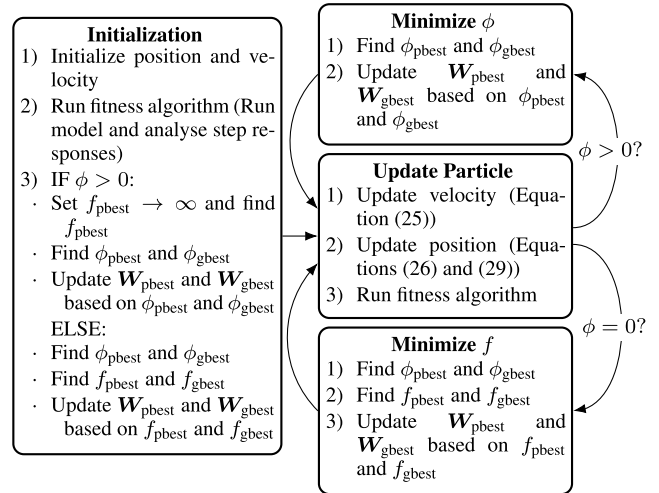


FIGURE 5. PSO with DOCHM pseudo-code state-machine diagram.

$\phi(W)$ represents how far the particle is from the feasible region. Thus, if $\phi(W) > 0$ at least one constraint is active and $\phi(W)$ must be minimized. In contrast, if $\phi(W) = 0$, then the particle is inside of the feasible region and the minimization process changes to secondary objective function, $f(W)$, that is related to the optimization problem without constraints.

$$f(W) = \max(\{ITAE_d, ITAE_q\}) \quad (31)$$

Eventually, during the optimization process, a particle can go outside of the feasible region and the minimization process returns to the primary objective function, but, as $\phi(W)$ comes back to zero again, the minimization returns to $f(W)$, and this loop repeats every time.

The PSO based DOCHM algorithm is present in Fig. 5.

IV. RESULTS OF WDOCHM-PSO ALGORITHM

The WDOCHM-PSO, the fitness algorithms, and the model used for its calculations were built using MATLAB/Simulink. For this model, a constant switching frequency equal to 10^{-4} seconds was used. The mechanical speed ω_m was equal to 1690 rpm, the control and prediction horizons were equal to $N_y = 20$ and $N_u = 10$. The computational time is near to $16 \mu\text{s}$, or 6.25 times the switching frequency [54], besides the DFIG parameters are also presented in Table 1.

According to [46], the suggested number of population is 2 to 5 times the dimension of the problem, and some works use 300 as the maximum number of iterations [55], [56]. Therefore, since each particle is a high-fidelity model, which means a high-cost particle, the PSO population $P_{\text{size}} = 16$ particles, and the maximum number of iterations was $k_{\max} = 300$, which takes approximately two days in an Intel Core i7 and 8 GB of RAM computer.

Fig. 6 describes the progress of $\phi(W)$ and $f(W)$. Initially, all particles are outside of the feasible region, so $f_{\text{gbest}} \rightarrow \infty$. After the 21st iteration, one particle reached to $\phi(W) = 0$, (or $\log(\phi(W)) = 0 \rightarrow -\infty$) and f_{gbest} decreases from infinity to a 1565.7. Therefore, after the optimal value of

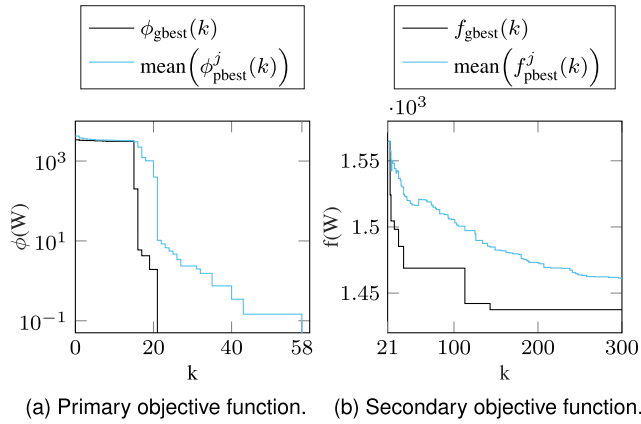


FIGURE 6. The progress of $\phi(W)$ and $f(W)$ as the number of iteration increases.

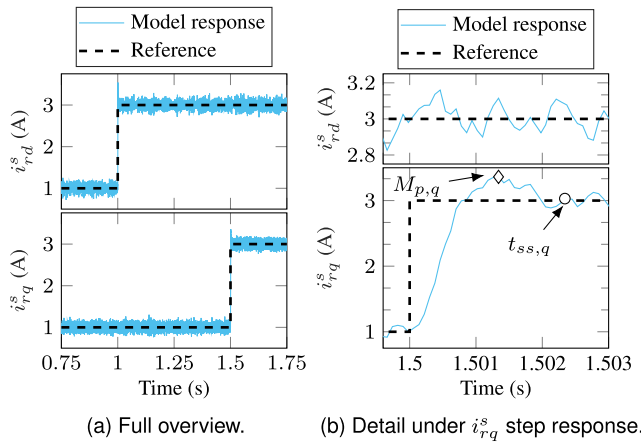


FIGURE 7. Simulated result for WDOCHM-PSO solution of W .

$\phi(W) = 0$ was found, there is a tendency to more particles go to this region and the mean of all ϕ_{pbest}^j decreases. After the 57th iteration all $\phi_{pbest}^j = 0$.

Also, the solution found by WDOCHM-PSO Method, was presented by:

$$W_{opt} = \begin{bmatrix} 365 \\ -299 \\ 212 \\ 742 \\ 9 \times 10^{-3} \\ -3 \times 10^{-2} \\ -2 \times 10^{-2} \\ 9.5 \times 10^{-3} \end{bmatrix} \quad (32)$$

and the corresponding response is given in Fig. 7, where Fig.7a indicates the full overview of both direct and quadrature responses, and Fig.7b shows the detail under i_{rq}^s step response. In Fig. 7b, the estimated settling time for quadrature current was $t_{ss,q} = 2.34$ ms and the overshoot for quadrature, $M_{p,q} = 20.47\%$.

Finally, Fig. 8 presents the model working in other situations of active and reactive power generation, as $P_s = 0$

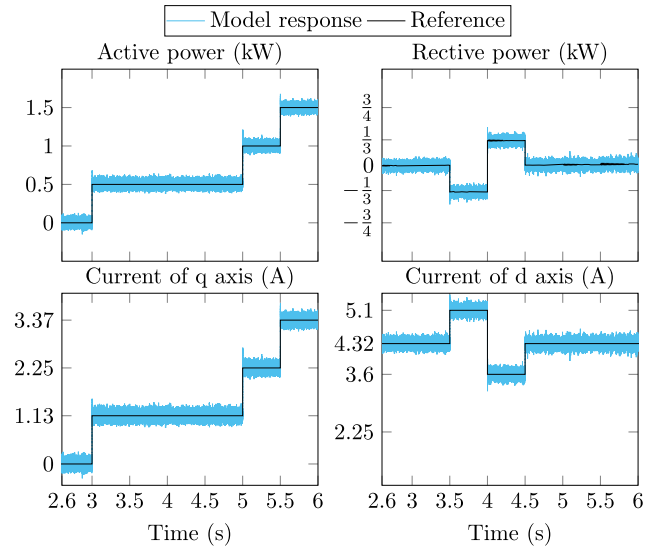


FIGURE 8. Model working for some power generation and the corresponding current.

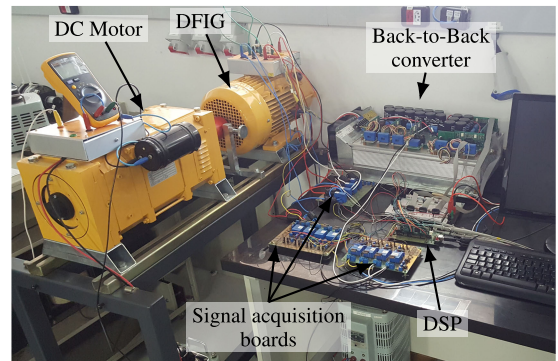


FIGURE 9. Workbench used to get experimental results.

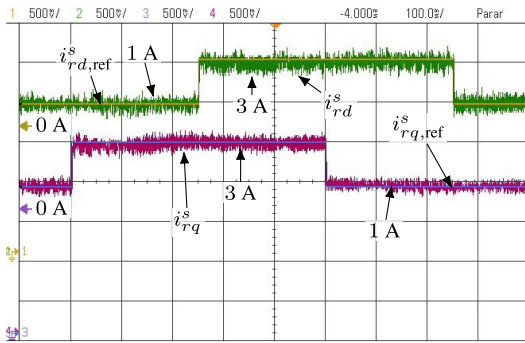
and $Q_s = 0$. Also, Fig. 8 shows the corresponding current of direct and quadrature components for the generated power.

V. EXPERIMENTAL RESULTS

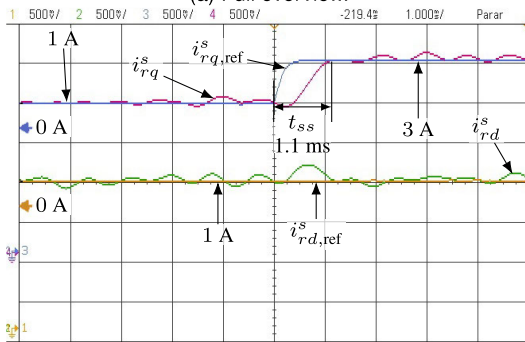
An experimental workbench was used to validate the design of weighting matrices and the theory presented in previous sections (Fig. 9). This workbench includes a Digital Signal Processor (DSP) TMS320F28335, a data acquisition board, a DC motor used to emulate the wind speed, a back-to-back converter and a 3 kW DFIG. The sampling time is the same value of the space vector modulation frequency (10 kHz). The speed is measured by using a 3600 PPR encoder. Moreover, DFIG parameters are presented in Table 1.

A. CONSTANT MECHANICAL SPEED OPERATION

Firstly, the same steps signals used as PSO Model references (Fig. 7) were used here, in order to test the performance of the proposal. In this way, the $i_{rd,ref}$ and $i_{rq,ref}$ changed from 1 A to 3 A and the speed is 1690 rpm as depicted in Figure 10. It can be noted that the proposed MBPC, using the

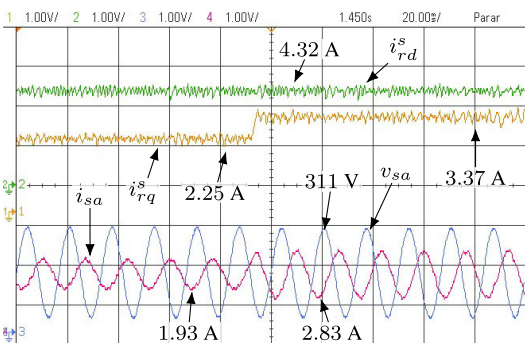


(a) Full overview.

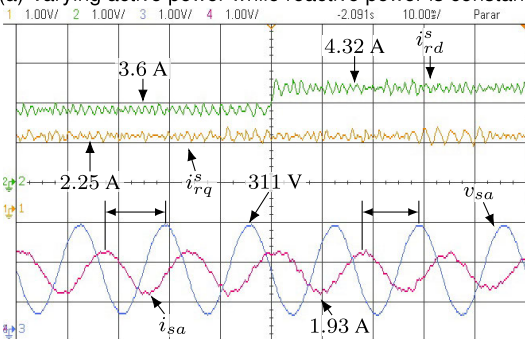


(b) Detail under i_{rq}^s step response.

FIGURE 10. Step responses of the proposal.



(a) Varying active power while reactive power is constant.



(b) Varying reactive power while active power is constant.

FIGURE 11. Stator voltage, v_{sa} , and current, i_{sa} , and rotor synchronous components, i_{rd}^s and i_{rq}^s behavior during a step test.

incremental space-state model and the PSO application to the weighting matrices, allows controlling the rotor current components. The quadrature settling time, $t_{ss,q} = 1.1$ ms,

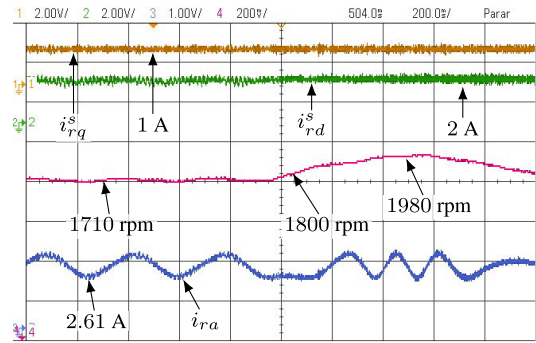


FIGURE 12. Test for several mechanical speed operation.

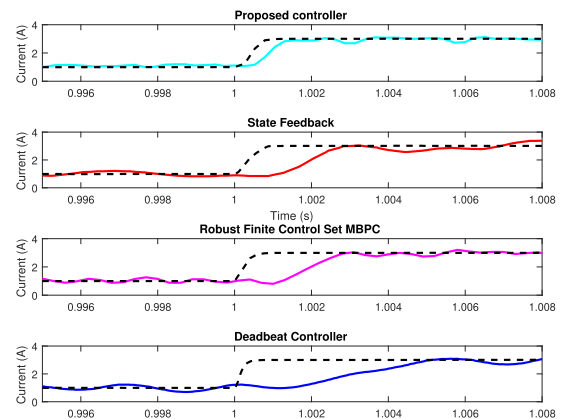


FIGURE 13. Comparison with the proposed controller and other approaches.

is better than the result shown in the simulated case as is depicted in Figure 10.

B. COMPARING ROTOR CURRENT COMPONENTS AND GENERATED POWER

In the second test, the behavior of the Phase A stator voltage v_{sa} and current, i_{sa} , is presented during a change in the rotor current references i_{rd}^s and i_{rq}^s . Fig. 11 presents the case where i_{rq}^s changes from 2.25 A to 3.37 A, while i_{rd}^s A keeps constant at 4.32 A. When the stator current amplitude rises from 1.93 A to 2.83 A, then, a consequent increment in the active power from -864 W to -1295 W can be noticed.

Moreover, the reactive power keeps constant at $Q_s = 266$ var. In the same way, Fig. 11b represents another scenario where i_{rd}^s was varied from 3.6 A to 4.32 A, while i_{rq}^s remains constant at 2.25 A. Therefore, a decrease in the reactive power from 543 var to 266 var is produced, while the active power was kept constant at $P_s = -864$ W. Again, in both cases, the proposed MBPC, using the PSO to design the weighting matrices, controls the rotor current in a right way.

C. SEVERAL MECHANICAL SPEED OPERATION

In the third experiment, both rotor currents keep constant at $i_{rd}^s = 2$ A and $i_{rq}^s = 1$ A, while the rotor speed ω_m , varying from 1710 rpm to 1980 rpm (please see Fig. 12), since

TABLE 1. DFIG parameters.

| Parameter | Value |
|---|-----------------------|
| Stator resistance per phase (R_s) | 1 Ω |
| Stator inductance per phase (L_s) | 0.2010 H |
| Rotor resistance per phase (R_r) | 3.122 Ω |
| Rotor inductance per phase (L_r) | 0.2010 H |
| Mutual inductance (L_m) | 0.1917 H |
| Synchronous stator speed (ω_s) | 120 π rad/s |
| Pole pairs (p) | 2 |
| Nominal Active Power (P_s) | 3 kW |
| Nominal Stator Voltage (V_s) | 220/380 Δ -Y V |
| Nominal Stator Flux (λ_s) | 0.8249 Wb |
| DC-link volage (V_{cc}) | 127 V |
| Nominal Rotor Voltage (V_r) | 440 Y V |

the rotor speed is related with the rotor current frequency i_{ra} . In fact, when $\omega_m = 1800$ rpm, or $\omega_{sl} = 0$ the rotor current waveform is a DC signal. Also, it can be seen that this proposal well-works when the rotor speed changes owing to the fact the MBPC controller reaches its references.

D. COMPARISON WITH OTHER CONTROLLERS

In this section, a comparison between the proposed controller with other alternatives presents in the literature is done. Figure 13 shows that comparison using the same step response presented in Figure 10 and the State Feedback controller depicted in [57], the Robust Finite Control Set proposed in [58] and the deadbeat controller described in [59]. It can be noticed that the proposed controller is nearly two times faster and exhibits less settling time than the other ones.

VI. CONCLUSION

In this paper, a methodology for designing the weighting matrices of an MBPC's cost function using WDOCHM-PSO was proposed. As many artificial intelligence methods, the WDOCHM-PSO is very flexible and it can be adapted to project many other controllers easily, whether linear or nonlinear, model-based or not, since the proposed fitness uses non-linear simulations as the main component. Good quality parameters, considering the designer requirements, were achieved, because the settling time and the overshoot of the step response were lower than the maximum specification. For this WDOCHM-PSO, none noise restrictions were directly considered, because, in its inner response, the solution is obtained with the ITAE minor value. And according to [60], the solution of this controller should indirectly minimize the noise.

The advantage of using an off-line algorithm to optimize the MBPC cost function is that there is no limited time to process the algorithm, thus, more complex models can be used during the optimization process. On the other hand, due to the DSP time limitations, it was proposed a simple incremental predictive model without the flux component. As a result, the analyzed response for the experimental case was under the defined constraints and faster than the optimal of the simulated model. Therefore, the off-line optimization method using PSO attended the expectations and it can be used to design MPBC weighting matrices.

Finally, it is possible to use this WDOCHM-PSO tuning approach in other situations. For doing that, it is necessary to change the fitness function, including the model and/or simulation, the fitness calculation processing, and constraints.

APPENDIX

See Table 1.

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