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Adaptive Sliding Mode Control Design for Nonlinear Unmanned Surface Vessel Using RBFNN and Disturbance-Observer

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ABSTRACT Unmanned surface vessel(USV) has been applied in the maritime security inspection and resources exploration to execute complex works with its advantages in automation and intelligence. While the nonlinear USV working in the complex ocean environment, the good trajectory tracking performance is an important capacity. However, the nonlinearity, modeling uncertainties (e.g., modeling error and parameter variations) and external disturbance (wind, wave, current, etc) are the main difficulties, which deteriorates the control performance. To solve this issue, most existing algorithms for USV's tracking have been developed based on the linearization of the USV's nonlinear dynamic model at specific equilibrium point. However, the integrated effect of nonlinearities, modeling uncertainties and external disturbance has not been well considered, which can degrade the USV's tracking performance. Therefore, to achieve the good tracking performance for USV, a nonlinear dynamic model is strictly derived in this paper with the integrate consideration of abovementioned issues, and an adaptive sliding mode control design using RBFNN(Radial Basis Function Neural Network) and disturbance-observer is subsequently developed, where a RBFNN approximator is designed to approximate and compensate modeling uncertainties, and a disturbance-observer is designed to estimate and compensate the effect of the external disturbance. Furthermore, the global stability of the overall closed-loop system of USV are strictly guaranteed. The comparative simulation is carried out to validate the fast response, better transient performance and robustness of our proposed control design via comparing with the existing methods.

INDEX TERMS Adaptive sliding mode control, neural network, unmanned surface vessel(USV), Lyapunov stability theorem, disturbance observer.

I. INTRODUCTION

With the development of advanced mechatronics and automation [1]–[4], unmanned surface vessel(USV) can undertake the oceanic tasks with the advantages of intelligence and safety, and has been widely applied in military and commercial fields in a long time [5]–[9], such as oceanic exploration and collection, maritime rescue, environmental inspection, etc. While the USV working in the complex ocean environ-

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ment, the nonlinearity, modeling uncertainties which consist of(e.g., modeling error and parameter variations) and external disturbance (wind, wave, current, etc) are the main difficulties, which deteriorates the control performance [10], [11], thus the accurate trajectory tracking is still a challenge for the USV's control.

To improve the USV's trajectory tracking performance, several control algorithms have been developed based on the linearization of the USV's nonlinear dynamic model at specific equilibrium point, which has good performance for trajectory tracking around the specific equilibrium point.

For instance, to solve the uncertainties and time-varying disturbances, an adaptive course control algorithm was proposed by Qiu *et al*. [12] based on trajectory linearization control technology, where a finite-time disturbances observer was designed by Wang *et al*. [13] to estimate the external disturbances and sideslip angle which can effect the USV's tracking performance. To estimate the unmodeled dynamics and disturbances, a fuzzy observer is developed in [14]. Moreover, to solve the matched and unmatched uncertainties, an adaptive control algorithm for the USV's tracking is proposed by Shin *et al*. [15] based on linear dynamic model using dynamic surface and virtual control. Though these control algorithms have been applied in the USV's trajectory tracking, they are not suitable for any desired complex trajectories and only suitable for specific desired trajectories at the equilibrium point. Besides, the modeling uncertainties and external disturbance are not well considered in the aforementioned linear controller, thus a more suitable trajectory tracking controller based on the nonlinear USV's dynamics with integrate consideration of the nonlinearity, modeling uncertainties and external disturbances is still a challenge to be developed.

For the USV's trajectory tracking with complex external disturbances, some trajectory control algorithms have been proposed based on nonlinear dynamic model [16]–[23], such as adaptive back-stepping control, sliding mode control, neural network-based control, model predictive control, etc. For instance, to achieve the good tracking performance, a disturbance-observer-based sliding mode control design is proposed in [16], where a disturbance observer is designed to estimate and compensate the modeling uncertainties and external disturbance. A nonlinear modeling scheme based on active Kalman filter was proposed by Han *et al*. [24] to estimate and compensate the unstructured model. An adaptive trajectory tracking control algorithm was proposed by Chen *et al*. [25]. It is worth mentioning that neural networks is designed in this paper to estimate the USV's hydrodynamics and external disturbances, and an observer is used to observe the unmeasurable velocities of output feedback controller. To deal with the saturation and constraints of USV's actuator and yaw, Guo and Zhang [26] proposed a tracking control algorithm using back-stepping control method, where an observer is derived to observe the USV's actuator dead-zones and disturbances. Moreover, a model-based event-triggered control method is proposed by Deng *et al*. [27], where a neural networks approximator is designed to estimate uncertainties from both internal system and external environment. To process the external environment disturbances, a control scheme based on model predictive method is proposed by Tan *et al*. [28] with the consideration of USV's physical constraints, which can be calculated and implemented online. To improve the tracking performance and robustness, a control strategy with the integrate consideration of portcontrolled hamiltonian method and back-stepping method is proposed by Lv *et al*. [29], which provides the experience to develop the comprehensive application of multiple control methods.

Though several control algorithms have been tried to achieve better performance, few algorithms can take full account of the integrated challenges consist of the USV's nonlinearities, modeling uncertainties and external disturbance. Therefore, good tracking performance is still challenging to be developed for USV's control. In this paper, an adaptive sliding mode control algorithm using RBFNN and disturbance-observer is proposed based on nonlinear USV's dynamic model. The main works and improvements over the existing methods can be listed as follows:

(a) A nonlinear dynamic model for USV is established, where the modeling uncertainties and external disturbances are fully taken into account.

(b) The modeling uncertainties are related with the motion state and dynamic parameters of the USV. To approximate and compensate modeling uncertainties, a RBFNN approximator is designed with its characteristic of universal approximation.

(c) Considering that the time-varying external disturbance is caused by the wind, wave and current, a disturbance observer is designed in this paper to observe and then compensate it.

(d) Integrating the RBFNN approximation and disturbanceobserver, an adaptive sliding mode algorithm is developed for USV's nonlinear dynamics, which is more suitable for any desired trajectories.

Therefore, the integrate effect of system nonlinearity, modeling uncertainties and external disturbance of the USV can be well handled using RBFNN and disturbanceobserver. Furthermore, to verify the effectiveness and priority of our proposed control design, the comparative simulations with different controllers and different desired trajectories are implemented, which shows the faster response, better transient and robustness of proposed control design.

The structure of this paper is roughly described as follows. Section II mainly describes the framework of the RBFNN and the definition of related physical parameter. With full consideration of modeling uncertainties and external disturbances, the modeling of the USV is derived in Section III. Subsequently, Section IV mainly describes the design of the adaptive slide-mode controller using RBFNN and disturbance observer. To prove the stability of the overall closed-loop system, the related derivation is also described in Section IV. Moreover, Section V presents the simulation of USV's tracking with different control algorithms and different desired trajectories, which verifies the significant tracking priority of proposed control design. Finally, the contributions and future research are summarized in Section VI.

II. THE DEFINITION OF RBFNN

Due to the advantages of universal approximation [30], the RBFNN algorithm has been widely applied for the modeluncertain systems. Moreover, the framework of the RBFNN and its definition can be described as follows in detail:

Definition 1 : As a kind of forward neural network, RBFNN consists of input layer, hidden layer and output layer,

 \mathbf{y}

FIGURE 1. The framework of RBFNN.

the detailed framework is shown in Fig.1, where *u* and *n* represent the number of node in input layer and hidden layer respectively.

The input function of RBFNN can be listed as:

$$
p = [p_1, \dots, p_i, \dots, p_u]^T
$$
 (1)

Then the Gaussian function can be described as:

$$
h_j(p) = \exp\left(-\frac{\left\|p - c_j\right\|^2}{b_j^2}\right) \tag{2}
$$

$$
h(p) = [h_1(p), \ldots, h_j(p), \ldots, h_n(p)]^T
$$
 (3)

where $\|\bullet\|$ represents the Euclidean norm of \bullet , $j = 1, ..., n$, $c_j = [c_{j1}, \ldots, c_{ji}, \ldots, c_{ju}]^T$. The output function of RBFNN is defined as follows:

$$
y = \sum_{j=1}^{n} W_j h_j(p) + \varsigma = W^T h(p) + \varsigma
$$
 (4)

where $W = [W_1, \ldots, W_j, \ldots, W_n]^T$ represents the output weight of each hidden node, ς is the estimation error.

III. THE MODELING OF USV's OVERALL SYSTEM

A. NONLINEAR MODELING

As a 3-DOF object in the plane, the model of USV is described in Fig[.2,](#page-2-0) and its kinematic model is derived as:

$$
\dot{\eta} = R(\psi) V \tag{5}
$$

As shown in Fig[.2,](#page-2-0) $\eta = \left[x \, y \, \psi\right]^T$ represents the position vector of USV, where x , y represent the global coordination and ψ represents the heading angle of the USV. $V = [u \ v \ r]^T$ represents the corresponding velocity state vector, where \overline{u} , \overline{v} and *r* represent the surge, sway velocities and yaw rotational velocity respectively. $R(\psi)$ represents the rotation matrix from body frame {*b*} to ground frame {*i*}, which is defined as follows,

$$
R(\psi) = \begin{bmatrix} \cos \psi & -\sin \psi & 0\\ \sin \psi & \cos \psi & 0\\ 0 & 0 & 1 \end{bmatrix}
$$
 (6)

FIGURE 2. The model of USV.

With the integrate consideration of external disturbance and modeling uncertainties, the USV's nonlinear dynamic model is be derived as:

$$
M'\dot{V} + C'V + D'V = \tau + R^T d_s \tag{7}
$$

where $\tau\left(\left[\tau_x \tau_y \tau_n\right]^T\right)$ is the USV's control vector. *M'* is the USV's inertial matrix, which consists of known part M'_0 and unknown part $\Delta M'$. The coriolis and centripetal term C' is a skew-symmetric matrix, which consists of known part C'_0 and unknown part $\Delta C'$. *D'* is the damping matrix, which consists of known part D'_0 and unknown part $\Delta D'$. d_s represents the USV's external disturbance caused by wind, wave and current, which can be estimated and compensated by disturbance observer in Section IV. The matrices M'_0 , C'_0 and D'_0 can be described as follows:

$$
M'_{0} = \begin{bmatrix} m - X_{\dot{u}} & 0 & 0 \\ 0 & m - Y_{\dot{v}} & mx_{g} - Y_{\dot{r}} \\ 0 & mx_{G} - N_{\dot{v}} & I_{zz} - N_{\dot{r}} \end{bmatrix}
$$
(8)

$$
C'_{0} = \begin{bmatrix} 0 & 0 & -m_{22}v - m_{23}r \\ 0 & 0 & m_{11}u \\ m_{22}v + m_{23}r & -m_{11}u & 0 \end{bmatrix}
$$
(9)

and

$$
D'_{0} = -\begin{bmatrix} X_{u} & 0 & 0\\ 0 & Y_{v} & Y_{r} \\ 0 & N_{v} & N_{r} \end{bmatrix}
$$
 (10)

where *m* represents the mass of USV. *Izz* represents the USV's moment of inertia. x_g represents the gravity center's position. *X*∗, *Y*∗, *Z*[∗] represent hydrodynamic coefficients. *mij* represents the element in the row *i* and column *j* of M'_0 .

As we all know, the driving forces and torque of the USV can be produced by the water-jet propulsion as:

$$
\tau = \begin{bmatrix} \tau_x \\ \tau_y \\ \tau_n \end{bmatrix} = \begin{bmatrix} T\cos\delta_r \\ -T\sin\delta_r \\ x_{\delta_r} T\sin\delta_r \end{bmatrix}
$$
(11)

where *T* represents driving force produced by the water-jet propulsion, δ_r represents the rudder angle of the water-jet propulsion, the x_{δ_r} represents the install position of the waterjet nozzle.

By substituting [\(5\)](#page-2-1) to [\(7\)](#page-2-2), the nonlinear USV system can be derived in the Lagrange form [17]:

$$
M_0 \ddot{\eta} + C_0 \dot{\eta} + D_0 \dot{\eta} = R\tau + d_m + d_s \tag{12}
$$

where $M_0 = RM_0'R^T$, $C_0 = R(C_0 - M_0'R^T \dot{R}) R^T$, $D_0 =$ RD'_0R^T , d_m represents various modeling uncertainties (e.g., modeling error and parameter variations), which can be approximated by RBFNN. Some properties and assumptions of [\(12\)](#page-3-0) can be listed as follows:

Property 1: $\dot{M}_0 - 2C_0$ is skew-symmetric, $\dot{M}_0 - 2C_0 = [\dot{M}_0 - 2C_0]^T$.

Assumption 1: The modeling uncertainties d_m is with the upper bound as: $\Vert d_m \Vert \leq d_m^*$

Assumption 2: While the USV working, it is usually considered that the external disturbances d_s is superimposed by the low frequency period signals [13], and the derivative of d_s is assumed to have an upper bound, which means $\|\dot{d}_s\| \leq d'_s$.

B. LINEAR MODELING

To enhance the trajectory tracking performance of USV, several proposed control algorithms have been proposed based on the linearization of the USV's nonlinear dynamic model at specific equilibrium point. To better analyze the physical meaning of the specific equilibrium point, the derivation process of the linear model is described as follows in detail.

According to [15], the equilibrium point of USV is defined as:

$$
u = u_0, v = 0, r = 0, \delta_r = 0, T = T_0 \tag{13}
$$

With the abovementioned nonlinear dynamic model [\(7\)](#page-2-2) and equilibrium point [\(13\)](#page-3-1), the USV's surge dynamics is derived as:

$$
\Delta \dot{u} = a_u \Delta u + b_u \Delta T \tag{14}
$$

where $\Delta u = u - u_0$, $\Delta T = T - T_0$, $a_u = \frac{X_u}{m - X_u}$, $b_u = \frac{1}{m - X_u}$.

With the Nomoto's steering model $[15]$, the USV's yaw dynamics can be calculated with the following first-order dynamics:

$$
\tau_r \Delta \dot{r} + \Delta r = b_r \Delta \delta_r \tag{15}
$$

where $b_r = \frac{T_0 x_{\delta_r}}{I_{zz} - N_f}$.

Moreover, assume that *u* is almost unchanged and the yaw perturbation is small, the USV's sway dynamics can be derived as:

$$
\tau_{\beta} \Delta \dot{\beta} + \Delta \beta = -K_{\beta} \Delta r \tag{16}
$$

where β represents the sideslip angle which can be calculated by $\Delta \beta = \alpha \tan 2 \left(\frac{\Delta v}{\Delta u} \right)$.

With the abovementioned surge dynamics [\(14\)](#page-3-2), yaw dynamics [\(23\)](#page-4-0), and sway dynamics [\(16\)](#page-3-3), the linearized dynamic model of USV is summarized as:

$$
\dot{x} = Ax + B\delta + \Delta \tag{17}
$$

FIGURE 3. Adaptive sliding mode control architecture for nonlinear USV system using RBFNN and disturbance-observer.

where $x = [u \beta \psi r]^T$ represents the motion state of USV, $\delta = \left[T \delta_r\right]^{T}$ represents the control input,

$$
A = \begin{bmatrix} a_u & 0 & 0 & 0 \\ 0 & -\frac{1}{\tau_\beta} & 0 & -\frac{K_\beta}{\tau_\beta} \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & -\frac{1}{\tau_r} \end{bmatrix}
$$
 (18)

$$
B = \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 0 & \frac{b_r}{\tau_r} \end{bmatrix}
$$
 (19)

With the desired trajectory η_d , $\dot{\eta}_d$ and $\ddot{\eta}_d$ designed by trajectory planner, the control objective for the USV is to track the desired trajectory with faster response, better transient and robustness, where $\eta_d = \left[x_d y_d \psi_d\right]^T$, $\dot{\eta}_d = \left[x_d y_d \psi_d\right]^T$, $\ddot{\eta}_d = \left[\ddot{x}_d \ddot{y}_d \ddot{\psi}_d \right]^T$,

IV. ADAPTIVE SLIDING MODE CONTROL DESIGN USING RBFNN AND DISTURBANCE-OBSERVER

As shown in Fig[.3,](#page-3-4) an adaptive sliding mode controller is proposed using RBFNN and disturbance-observer to track the desired trajectory η_d , $\dot{\eta}_d$, $\ddot{\eta}_d$. A RBFNN approximator is designed to approximate and compensate modeling uncertainties, and a disturbance-observer is designed to estimate and compensate the effect of the external disturbance. Furthermore, the overall stability of the control system is proved.

A. CONTROLLER DESIGN

According to [31]–[34], the sliding-mode surface *s* can be defined with the tracking errors:

$$
s = Ke + \dot{e} \tag{20}
$$

where $e = \eta - \eta_d$, $K = diag\{k_1, ..., k_i, ..., k_w\}$, $i =$ $1, 2, \ldots, w$.

Substituting [\(12\)](#page-3-0) into [\(20\)](#page-3-5), thus,

$$
M_0 \dot{s} = M_0 (\ddot{\eta} - \ddot{\eta}_d + k\dot{e})
$$

= $M_0 (k\dot{e} - \ddot{\eta}_d) + M_0 \ddot{\eta}$
= $M_0 (k\dot{e} - \ddot{\eta}_d) - C_0 \dot{\eta} - D_0 \dot{\eta} + R\tau + d_s + d_m$
= $M_0 (k\dot{e} - \ddot{\eta}_d) - C_0 (s + \dot{\eta}_d - k\dot{e}) - D_0 \dot{\eta} + R\tau + d_s + d_m$
= $M_0 (k\dot{e} - \ddot{\eta}_d) - C_0 (\dot{\eta}_d - k\dot{e}) - C_0 s - D_0 \dot{\eta}$
+ $R\tau + d_s + d_m$ (21)

Then, the adaptive control input τ for the USV can be designed as [35]–[38]:

$$
\tau = -R^T \left(\rho + K_{\nu} s + \hat{d}_m + \xi s a t \left(s \right) + \hat{d}_s \right) \tag{22}
$$

where $\rho = -M_0(\ddot{\eta}_d - k\dot{e}) - C_0(\dot{\eta}_d - ke) - D_0\dot{\eta}$. \hat{d}_s represent the estimate value of the disturbance observer. $sat(s)$ is a saturation function to avoid the chattering phenomenon, which is designed as:

$$
sat(s) = \begin{cases} 1 & s > \Delta \\ \frac{1}{\Delta}s & ||s|| \le \Delta \\ -1 & s < -\Delta \end{cases}
$$
 (23)

, Δ is the positive boundary layer, which can be selected as: $\Delta = \frac{4\varepsilon}{\xi}.$

 \hat{d}_m represent the output value of the RBFNN to approximate the unknown nonlinear modeling uncertainties d_m , which can be designed as follows:

$$
\hat{d}_m = \hat{W}h(p) \tag{24}
$$

where

$$
p = [\dot{\eta}_d, \dot{e}, \eta_d, e] \tag{25}
$$

$$
h_j(p) = \exp\left(-\frac{||p - c_j||^2}{h^2}\right) \tag{26}
$$

$$
h(p) = [h_j(p)]^T
$$
 (27)

$$
d_m = Wh(p) + \varsigma \tag{28}
$$

where ζ is the estimate error of RBFNN with the upper bound:

$$
\|\varsigma\| \le \varsigma^* \tag{29}
$$

The adaptive law of RBFNN can be designed as:

$$
\dot{\hat{W}} = \frac{1}{\delta} sh(p) \tag{30}
$$

where δ is the positive parameter of the adaptive law.

B. DISTURBANCE-OBSERVER DESIGN

To estimate and compensate the USV's external disturbance caused by wind, wave and current, the disturbance observer \hat{d}_s is designed with the following equation:

$$
\dot{Z} = L(C_0\dot{\eta} + D_0\dot{\eta} - R\tau) - L\hat{d}_s
$$

$$
\hat{d}_s = Z + P
$$
 (31)

where,

$$
L = H^{-1} M_0^{-1} \tag{32}
$$

$$
P = H^{-1}\dot{\eta} \tag{33}
$$

H is a invertible constant matrix, and can be selected according to the transient and steady-state response of disturbance observer. Moreover, *M*⁰ and *L* are a constant. Therefore, the derivative of P is derived with [\(32\)](#page-4-1) and [\(33\)](#page-4-1):

$$
\dot{P} = LM_0 \ddot{\eta} \tag{34}
$$

As $\tilde{d}_s = d_s - \hat{d}_s$ represent the estimate error, the derivative of \tilde{d}_s can be derived as the following equation:

$$
\dot{\tilde{d}}_s = \dot{d}_s - \dot{\tilde{d}}_s \n= \dot{d}_s - \dot{Z} - \dot{P} \n= \dot{d}_s - L (C_0 \dot{\eta} + D_0 \dot{\eta} - R\tau) + L \dot{d}_s - LM_0 \ddot{\eta} \n= \dot{d}_s + L \hat{d}_s - L (M_0 \ddot{\eta} + C_0 \dot{\eta} + D_0 \dot{\eta} - R\tau) \n= \dot{d}_s + L \hat{d}_s - L (d_s + d_m) \n= \dot{d}_s - L \tilde{d}_s - L d_m \n= -L \tilde{d}_s + \mu
$$
\n(35)

where $\mu = \dot{d}_s - Ld_m$, which is bounded since \dot{d}_s and d_m are bounded with assumption 1 and 2. It follows that:

$$
\tilde{d}_s \le e^{-Lt} \tilde{d}_s(0) + \frac{\mu}{L} \left(1 - e^{-Lt} \right) \tag{36}
$$

Thus, the observer error \tilde{d}_s is bounded as $t \to \infty$, which reflects stability of disturbance observer [\(31\)](#page-4-2) with boundedinput bounded-output stability.

C. OVERALL STABILITY ANALYSIS

Considering the closed-loop USV control system, one theorem can be listed in the following to verify the overall stability.

Theorem: With the controller [\(22\)](#page-4-3), RBFNN approximator [\(24\)](#page-4-4) and disturbance observer [\(31\)](#page-4-2), the modeling uncertainties and external disturbances can be well estimated and compensated and the overall control system is stable, all the signals are bounded.

Proof : Considering the overall closed-loop system, define the Lyapunov Function *V* as:

$$
V = \frac{1}{2} sM_0 s + \frac{1}{2} \delta \tilde{W}^T \tilde{W}
$$
 (37)

where $\tilde{W} = W - \hat{W}$. Then, the derivative of *V* is derived as the following equation.

$$
\dot{V} = s^T M_0 \dot{s} + \frac{1}{2} s \dot{M}_0 s + \delta \tilde{W}^T \dot{\tilde{W}} \tag{38}
$$

For

$$
M_0 \dot{s} = d_m - K_v s - \hat{d}_m - \xi s \, \text{at} \, (s) - \hat{d}_s - C_0 s + d_s \quad (39)
$$

Substituting [\(39\)](#page-4-5) and [\(22\)](#page-4-3) into [\(38\)](#page-4-6), then,

$$
\dot{V} = s^T \left(d_m - K_v s - \hat{d}_m - \xi s \alpha t \left(s \right) - \hat{d}_s - C_0 s + d_s \right)
$$
\n
$$
+ \frac{1}{2} s \dot{M}_0 s + \delta \tilde{W}^T \dot{\tilde{W}}
$$
\n
$$
= s^T \left(\tilde{d}_m - K_v s - \xi s \alpha t \left(s \right) + \tilde{d}_s \right)
$$
\n
$$
+ \frac{1}{2} s^T \left(\dot{M}_0 - 2C_0 \right) s - \delta \tilde{W}^T \dot{\tilde{W}}
$$
\n
$$
= -s^T K_v s + s^T \left(\xi + \tilde{d}_s - \xi s \alpha t \left(s \right) \right)
$$
\n
$$
+ \tilde{W}^T \left(s h \left(p \right) - \delta \dot{\tilde{W}} \right)
$$
\n
$$
= -s^T K_v s + s^T \left(\xi + \tilde{d}_s - \xi s \alpha t \left(s \right) \right) \tag{40}
$$

TABLE 1. The USV's principle physical parameters [39], [40].

Parameter	Value	Parameter	Value
$_{m}$	23.8	N_r	-1.9
I_{zz}	1.76	X_{ii}	-2.0
X_G	0.046	$Y_{\dot{v}}$	-10.0
X_{u}	-0.72253	$Y_{\dot{\tau}}$	-0.1
Y_{v}	-0.88965	N_{i}	
Y_r	-7.25	$N_{\dot{r}}$	-0.1
N.,	0.0313		

where $\tilde{d}_m = d_m - \hat{d}_m = \tilde{W}h(p) + \varsigma$, ξ is selected as: $\xi \ge$ $\zeta^* + \left\| \tilde{d}_s(0) \right\| + \left\| \frac{\mu}{L} \right\|$ $\frac{\mu}{L}$.

If $\|s\| \leq \Delta$, then the derivative of *V* can be calculated as:

$$
\dot{V} = -s^T K_v s + s^T \left(\varsigma + \tilde{d}_s \right) - \frac{\xi^2}{4\varepsilon} s^2
$$
\n
$$
= -s^T K_v s + s^T \left(\varsigma + \tilde{d}_s \right) - \xi \|s\| + \varepsilon
$$
\n
$$
- \left(\frac{1}{2\sqrt{\varepsilon}} \xi \|s\| - \sqrt{\varepsilon} \right)^2
$$
\n
$$
\leq -s^T K_v s + \|s\| \left(\varsigma^* + \left\| \tilde{d}_s(0) \right\| + \left\| \frac{\mu}{L} \right\| - \xi \right) + \varepsilon
$$
\n
$$
\leq -s^T K_v s + \varepsilon
$$
\n(41)

If $||s|| > \Delta$, then the derivative of *V* can be calculated as:

$$
\dot{V} = -s^T K_v s - \xi \|s\| + s^T \left(\xi + \tilde{d}_s\right)
$$
\n
$$
\leq -s^T K_v s + \left(\|\xi\| + \left\|\tilde{d}_s\right\| - \xi\right) \|s\|
$$
\n
$$
\leq -s^T K_v s + \left(\xi^* + \left\|\tilde{d}_s(0)\right\| + \left\|\frac{\mu}{L}\right\| - \xi\right) \|s\|
$$
\n
$$
\leq -s^T K_v s
$$
\n
$$
\leq -s^T K_v s + \varepsilon
$$
\n(42)

According to the Lyapunov stability theorem, *s*, \tilde{d}_m , *e* and \dot{e} are all bounded, the stability of the overall closed-loop USV control system is verified and all the signals are bounded. The proof of *Theorem* is complete.

Remark: With the proposed control design [\(22\)](#page-4-3), the external disturbances d_s can be well estimated and compensated, and the modeling uncertainties d_m can be well approximated and compensated, thus the good trajectory tracking performance can be achieved.

V. SIMULATION

A. SIMULATION SETUP

To describe and verify the tracking performance of our proposed control algorithm, the benchmark prototype CyberShip II [39], [40] is selected as the platform of our comparative simulation, and the principal physical parameters of this platform are listed in Table [1.](#page-5-0) All quantities in the simulation are expressed in the international system of units (SI).

To make the trajectory tracking superiority of our control design more convincing, the linear control, PID control, sliding mode control are selected as comparative object of our proposed control design, which can be shown in detail as below:

C1: Linear controller (proposed in [15]). The primary parameters are specified as $k_v = 100$, $k_{c1} = 1$, $k_{c2} = 1$, $d_w = 0.1$, $\tau = 0.1$, $\gamma_u = \gamma_\beta = \gamma_r = 0.01$.

C2: Traditional PID controller. The control law is selected as the following equation:

$$
\tau = R(\psi)^{T} \left(K_{p} \left(\eta_{d} - \eta \right) + K_{d} \left(\dot{\eta}_{d} - \dot{\eta} \right) + K_{i} \int (\eta_{d} - \eta) \, dt \right)
$$
\n(43)

where $K_p = diag\{100, 100, 100\}$, $K_d = diag\{100, 100, 100\}$, $K_i = diag\{20, 20, 20\}.$

C3: Sliding mode controller proposed in [16]. The primary parameters are specified as $K_v = diag\{100, 100, 100\},\$ ξ = *diag*{100, 100, 100}, K = *diag*{10, 10, 10}, H = $diag\{0.1309, 0.1309, 0.1309\}, \delta = 0.1, \Delta = 0.1.$

C4: Adaptive sliding mode controller using RBFNN and disturbance-observer. The control law, rbfnn approximator and disturbance-observer for the USV are selected as [\(22\)](#page-4-3), [\(24\)](#page-4-4) and [\(31\)](#page-4-2), where $K_v = diag\{100, 100, 100\}$, ξ = *diag*{100, 100, 100}, K = *diag*{10, 10, 10}, H = $diag\{0.1309, 0.1309, 0.1309\}, \delta = 0.1, \Delta = 0.1.$

To verify the robustness of these comparative control algorithms, the complex modeling uncertainties d_m can be simulated as: $d_m = \left[d_{mx} d_{my} d_{my} \right]^T$, where

$$
d_{mx} = 2\cos(\dot{x}_d) + \sin(x_d) + 2\sin(x - x_d) + 2\cos(\dot{x} - \dot{x}_d)
$$
\n(44)

$$
d_{my} = 3\cos(\dot{y}_d) + \sin(y_d) + 4\sin(y - y_d) + 2.5\cos(\dot{y} - \dot{y}_d)
$$
\n(45)

$$
d_{m\psi} = 0.4 \cos (\dot{\psi}_d) + 0.3 \sin (\psi_d) + 4 \sin (\psi - \psi_d) + 2 \cos (\dot{\psi} - \dot{\psi}_d)
$$
(46)

Moreover, the external disturbances d_s can be simulated as:

$$
d_s = \begin{bmatrix} 15 + 0.2\sin\left(0.1\pi t - \frac{\pi}{3}\right) + 0.3\cos\left(0.3\pi t + \frac{\pi}{4}\right) \\ 17 + 0.3\sin\left(0.2\pi t + \frac{\pi}{4}\right) + 0.4\cos\left(0.1\pi t - \frac{\pi}{6}\right) \\ 7 + 0.5\sin\left(0.3\pi t + \frac{\pi}{6}\right) + 0.15\cos\left(0.2\pi t - \frac{\pi}{3}\right) \end{bmatrix}
$$
(47)

In addition, MIAC index [41] is introduced to evaluate and compare the tracking performance of these comparative control algorithms with the integrate consideration of the transient and steady-state tracking, which is designed as:

$$
M I A C = \frac{1}{t_f - t_0} \int_{t_0}^{t_f} |j_e| dt
$$
 (48)

where j_e represents the tracking errors x_e , y_e , ψ_e , t_0 and *t^f* represent the start simulate time and end simulate time respectively.

As is known to all, the linear controller has good trajectory tracking performance for some specific trajectories at the equilibrium point [\(13\)](#page-3-1) and not track well for the trajectory which away from the equilibrium point [15]. To verify tracking performance of these comparative control algorithms with different desired trajectories, two simulations are designed as follows, which reflects somewhat practical tracking performance.

FIGURE 4. The tracking performance of these comparative controllers in SET1.

SET1: Straight-Line Trajectory Tracking SET2: Circular Trajectory Tracking

B. SIMULATION OF SET1

To compare with the tracking performance of these control algorithms with the physical effect of the specific equilibrium point, the desired trajectory for simulation SET1 is designed as $(51)-(52)$ $(51)-(52)$ $(51)-(52)$ and Fig[.4\(](#page-6-1)c). Simultaneously, the equilibrium point of this desired trajectory is listed as follows:

$$
u = 3, v = 0, r = 0, \delta_r = 0, T = 10 \tag{49}
$$

$$
x_d = \frac{3\sqrt{2}}{2}t0 \le t \le 40
$$
 (50)

$$
y_d = \frac{3\sqrt{2}}{2}t0 \le t \le 40
$$
 (51)

$$
\psi_d = \frac{\pi}{4} \cdot 0 \le t \le 40 \tag{52}
$$

The USV's initial state in SET1 is selected as: $\eta = [0 \ 0 \ \frac{\pi}{4}],$ $\dot{\eta} = \left[3\cos\left(\frac{\pi}{4}\right)3\sin\left(\frac{\pi}{4}\right)0\right].$

With the comparative simulation in SET1, the simulate results are shown in Fig[.4-](#page-6-1)Fig[.6.](#page-6-2) The control input of proposed control is shown in Fig[.5.](#page-6-3) The trajectory tracking performance of USV with these comparative control algorithms is shown in Fig[.4\(](#page-6-1)a), which reflects that the desired trajectory

FIGURE 5. The control input of proposed controller in SET1.

FIGURE 6. The estimate performance of disturbance observer [\(31\)](#page-4-2) in SET1.

are tracked by the USV accurately and smoothly. Meanwhile, Fig[.4\(](#page-6-1)b) shows the corresponding tracking errors, which can be closer converge to a reasonable range gradually. Subsequently, the stability of these comparative control algorithms is verified. Furthermore, to show the control superiority of our control design in detail, the MIAC of different comparative control algorithms in SET1 are listed in table [2.](#page-7-0) It can be seen that *MIACx^e* of C4 is 9.47% of C1, 7.32% of C2 and 81.82% of C3, *MIACy^e* of C4 is 6.19% of C1, 5.09% of C2 and 77.78% of C3, *MIAC*ψ*^e* of C4 is 7.24% of C1, 3.02% of C2 and 73.33% of C3. Therefore, for the desired trajectory at the equilibrium point, it can be judged that the proposed control design has better performance in terms of transient response and robustness compared with other controllers.

Furthermore, Fig[.6\(](#page-6-2)a) shows the estimate performance of observer [\(31\)](#page-4-2), where the blue line and the red line represent the estimate value of observer [\(31\)](#page-4-2) and the practical external disturbances respectively. Simultaneously, to show the estimated capacity of the observer [\(31\)](#page-4-2), the corresponding

MIAC	Value	
	$MIAC_{x_e} = 0.0190$	
$C1$: controller proposed in [15]	$MIAC_{ue} = 0.0226$	
	$MIAC_{\psi_e} = 0.0152$	
	$\overline{MIAC_{x_e}} = 0.0246$	
$C2:PID$ controller	$MIAC_{y_e} = 0.0275$	
	$MIAC_{\psi_e} = 0.0364$	
C3: controller proposed in [16]	$\overline{MIAC_{x_e}} = 0.0022$ $MIAC_{ue} = 0.0018$	
	$MIAC_{\psi_e} = 0.0015$	
	$M I A C_{x_e} = 0.0018$	
C4: proposed control design	$MIAC_{y_e} = 0.0014$	
	$MIAC_{\psi_e}=0.0011$	

TABLE 2. MIAC of these comparatives controllers in SET1.

estimate error is shown in Fig[.6\(](#page-6-2)b), which can achieve ultimately bounded within a reasonable range gradually. Therefore, the stability and estimate capacity of disturbance observer are verified. Subsequently, the approximate performance of rbfnn approximator (24) is shown in Fig[.7\(](#page-7-1)a). The blue line and the red line in Fig[.7\(](#page-7-1)a) represent the approximate value of rbfnn [\(24\)](#page-4-4) and the practical modeling uncertainties respectively. Fig[.7\(](#page-7-1)b) describes the corresponding approximate error, which can converge to a reasonable range gradually. Therefore, the stability and approximate capacity of rbfnn approximator [\(24\)](#page-4-4) are verified.

C. SIMULATION OF SET2

To evaluate and analyze the tracking performance of these comparative control algorithms with the trajectory away from the equilibrium point, the desired trajectory of SET2 is designed as $(53)-(55)$ $(53)-(55)$ $(53)-(55)$ and Fig[.8\(](#page-7-3)c).

$$
x_d = \begin{cases}\n15 & 0 \le t \le T_1 \\
15 \cos (0.2 (t - T_1)) & T_1 < t \le T_2 \\
-15 & T_2 < t \le T_3 \\
-15 & T_3 < t \le T_4\n\end{cases} \tag{53}
$$
\n
$$
-15 \cos (0.2 (t - T_4)) & T_4 < t \le T_5 \\
15 & T_5 < t \le T_6\n\end{cases} \tag{53}
$$
\n
$$
y_d = \begin{cases}\n5t - 0.1t^2 & 0 \le t \le T_1 \\
40 - 3 (t - T_2) - 0.1 (t - T_2)^2 & T_2 < t \le T_3 \\
-5t + 0.1 (t - T_3)^2 & T_3 < t \le T_4 \\
-15 \sin (0.2 (t - T_4)) - 40 & T_4 < t \le T_5 \\
3t + 0.1 (t - T_5)^2 - 40 & T_5 < t \le T_6\n\end{cases} \tag{54}
$$
\n
$$
y_{t} = \begin{cases}\n\frac{\pi}{2} & 0 \le t \le T_1 \\
0.2 (t - T_1) + \frac{\pi}{2} & T_1 < t \le T_2 \\
\frac{3\pi}{2} & T_2 < t \le T_3\n\end{cases} \tag{55}
$$

$$
\psi_d = \begin{cases}\n\frac{2}{2} & t_2 < t \leq T_3 \\
\frac{3\pi}{2} & T_3 < t \leq T_4 \\
0.2(t - T_4) + \frac{3\pi}{2} & T_4 < t \leq T_5 \\
\frac{5\pi}{2} & T_5 < t \leq T_6\n\end{cases}
$$
(55)

where $T_1 = 10$, $T_2 = 10 + 5\pi$, $T_3 = 20 + 5\pi$, $T_4 = 30 + 5\pi$, $T_5 = 30 + 10\pi$, $T_6 = 40 + 10\pi$. The USV's initial state in SET2 is selected as: $\eta = [15 \ 0 \ \frac{\pi}{2}], \dot{\eta} = [0 \ 5 \ 0].$

With the comparative simulation in SET2, the corresponding results are shown in Fig[.8-](#page-7-3)Fig[.10.](#page-8-0) The control input of

FIGURE 7. The approximate performance of rbfnn [\(24\)](#page-4-4) in SET1.

FIGURE 8. The tracking performance of these comparative controllers in SET₂.

proposed control is shown in Fig[.9.](#page-8-1) The trajectory tracking performance is shown in Fig[.8\(](#page-7-3)a), and corresponding tracking error *e* is shown in Fig[.8\(](#page-7-3)b). It can be derived from above results that the corresponding tracking error *e* of C1 and C2, approximator error \tilde{d}_m and estimate error \tilde{d}_s can converge to a small reasonable range gradually, which verifies the stability

FIGURE 9. The control input of proposed controller in SET2.

FIGURE 10. The estimate performance of disturbance observer [\(31\)](#page-4-2) in SET2.

and effectiveness of PID controller and proposed control design. It is worth noting that the tracking error *e* of C3 is larger than C1 and C2 evidently, which reflects that the linear controller is not suitable for the USV's tracking with the desired trajectory away from the equilibrium point. Table [3](#page-8-2) shows the MIAC in SET2, which can be calculated that *MIACx^e* of C4 is 0.013% of C1, 0.082% of C2 and 10% of C3, *MIACy^e* of C4 is 0.022% of C1, 0.318% of C2 and 41.3% of C3, *MIAC* $_{\psi_e}$ of C4 is 0.48% of C1, 0.49% of C2 and 15.1% of C3. Therefore, according to the abovementioned comparison, with the desired trajectory away from the equilibrium point, the proposed control design has better performance in terms of transient response and robustness than other comparative controllers.

According to the above analysis of simulate results in SET1 and SET2, compared with linear controller C1, PID controller C2 and sliding mode controller C3, the tracking priority and effectiveness of proposed control design are verified.

TABLE 3. MIAC of these comparative controllers in SET2.

FIGURE 11. The approximate performance of rbfnn [\(24\)](#page-4-4) in SET2.

VI. CONCLUSION

An adaptive sliding mode control design using RBFNN and disturbance-observer is proposed in this paper for nonlinear USV system with modeling uncertainties (e.g., modeling error and parameter variations) and external disturbances (wind, wave, current, etc). Firstly, a nonlinear dynamic model is derived in detail with full account of modeling uncertainties and external disturbance. Subsequently, a RBFNN approximator is designed to approximate and compensate modeling uncertainties, and a disturbance-observer is developed to estimate and compensate the external disturbance. Based on the RBFNN approximator and disturbance-observer, a slidemode controller is developed to track any kinds of desired trajectories. Thus, the integrated effect of nonlinearities, modeling uncertainties and external disturbances for the USV's trajectory tracking can be well handled. Moreover, the overall stability of the closed-loop control system is strictly guaranteed. A set of comparative simulations are implemented and MIAC is calculated simultaneously to analyze the tracking performance. The simulate results verify the significant priority of our proposed control design with faster response, better transient and robustness. In the future, we will consider the impact of control input saturation, violent external

disturbances and modeling uncertainties on system robust stability and tracking performance. Furthermore, the proposed control design will be applied for the practical experiments, and corresponding experimental results will be analyzed.

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