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# **WE RESEARCH ARTICLE**

# ADP-Based  $H_{\infty}$  Optimal Control of Robot Manipulators With Asymmetric Input Constraints and Disturbances

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**ABSTRACT** Trajectory tracking control for robot manipulators is an attractive topic in the research community. This is a challenging problem because robot manipulators are complex nonlinear systems. Furthermore, the tracking control performance for robot manipulators is greatly affected by input constraints and external disturbances. This paper proposes a novel  $H_{\infty}$  optimal controller for robot manipulators with asymmetric input constraints and external disturbances based on adaptive dynamic programming (ADP). Firstly, a strict feedback nonlinear system is used to represent the robot manipulator dynamics, and then a feedforward controller is designed to construct the tracking error dynamics. Secondly, a value function is introduced, and the Hamilton-Jacobi-Isaacs equation is made and approximated online by the principle of adaptive dynamic programming. Thirdly, the optimal control law and disturbance compensation law are determined. The stability and convergence of the proposed algorithm are analyzed by the Lyapunov technique. Finally, the controller performance is verified through simulation and experimental results with STM32F407 of STMicroelectronics.

**INDEX TERMS** Robot manipulators, asymmetric input constraints, adaptive dynamic programming,  $H_{\infty}$ optimal control.

#### **I. INTRODUCTION**

Tracking control for robot manipulators is a complex problem because robot manipulators are nonlinear systems with unknown and uncertain parameters [\[1\],](#page-9-0) [\[2\]. F](#page-9-1)urthermore, the manipulators are often affected by external disturbances and constrained inputs. For many decades, such a problem has attracted much attention from researchers. In [\[3\],](#page-9-2) a neural adaptive PID control scheme was applied to robot manipulators with uncertain parameters without measuring velocity and acceleration. A self-tuning fuzzy controller was introduced in  $[4]$  for robots with uncertain parameters. A robust mode predictive controller was proposed in [\[5\]](#page-9-4) for robots with disturbances. In [\[6\], an](#page-9-5) adaptive fuzzy sliding controller and disturbance observer were

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<span id="page-0-8"></span><span id="page-0-7"></span><span id="page-0-6"></span><span id="page-0-1"></span><span id="page-0-0"></span>presented. Additionally, neural adaptive sliding controllers were proposed in [7] [and](#page-9-6) [\[8\], in](#page-9-7) which radial basis function neural networks (RBFNNs) are applied to estimate unknown nonlinear functions. Overall, using neural networks [\[7\],](#page-9-6) [\[9\]](#page-9-8) or fuzzy models [\[6\], th](#page-9-5)e adaptive sliding controller has effectively handled the tracking control problem for robots with disturbances, but they cannot handle input saturation, leading to poor or even unstable control performance [\[10\],](#page-9-9) [\[11\].](#page-9-10)

<span id="page-0-16"></span><span id="page-0-15"></span><span id="page-0-14"></span><span id="page-0-13"></span><span id="page-0-12"></span><span id="page-0-11"></span><span id="page-0-10"></span><span id="page-0-9"></span><span id="page-0-5"></span><span id="page-0-2"></span>He et al. [\[12\]](#page-10-0) proposed an adaptive controller based on RBFNN to eliminate the effect of the input saturation. In [\[13\],](#page-10-1) an adaptive fuzzy dynamic surface controller was introduced for robot manipulators, in which the input saturation was considered using a smooth function along with the mean value theorem. On the other hand, the backend systems were designed in [\[14\],](#page-10-2) [\[15\], a](#page-10-3)nd [\[16\]](#page-10-4) to solve the input saturation. More generally, in [\[17\], a](#page-10-5)n asymmetric saturation model

<span id="page-1-4"></span><span id="page-1-3"></span>based on a Gaussian error function was applied to handle the influence of asymmetric input saturation. An adaptive neural controller and a hyperbolic tangent function were also proposed in [\[18\]](#page-10-6) to deal with the phenomenon. Additionally, in [\[19\], a](#page-10-7) controller for asymmetrically constrained inputs was designed by converting a system with asymmetrically constrained inputs into a system with symmetrically constrained inputs. In [\[20\], a](#page-10-8) constraint controller was built based on the asymmetric barrier Lyapunov function. However, the proposed algorithms mentioned above do not minimize any cost function, which means they are not optimal. Therefore, the goal of robust optimal control is important and equally challenging.

<span id="page-1-7"></span><span id="page-1-6"></span><span id="page-1-5"></span>The robust optimal control problem in an  $H_{\infty}$  control manner is solved if the optimal control law and the external disturbance compensation law are determined. Generally, the laws depend on the solution of the Hamilton-Jacobi-Issac (HJI) equation [\[21\]. U](#page-10-9)nfortunately, solving the HJI equation is a challenge since it is a higher-order nonlinear differential equation and has no analytical solution. Recently, ADP, a reinforcement learning (RL) version, has become a useful method for approximating online solutions of HJI equations [\[21\],](#page-10-9) [\[22\]. R](#page-10-10)obust optimal control algorithms were designed for robots in  $[23]$  and  $[24]$ , but they used function approximations with double NNs. On the contrary, in [\[25\],](#page-10-13) the  $H_{\infty}$  optimal controller was designed based on ADP combined with the zero-sum game theory and implemented with only one NN, but the input constraint problem was not mentioned. In [\[26\], t](#page-10-14)he saturated optimal tracking controller using RL was offered for a robot manipulator, but the external disturbance rejection was not mentioned. The robust optimal control algorithm was introduced in [\[27\], w](#page-10-15)here rejecting disturbances was offered but dealing with saturated inputs was not considered. In our previous work  $[28]$ , we designed an  $H_{\infty}$  optimal controller for single-wheel robots with input constraints but did not address the asymmetrically saturated inputs. In addition, an  $H_{\infty}$  optimal control algorithm with asymmetrically saturated inputs was proposed in our previous work [\[29\]; h](#page-10-17)owever, the algorithm only considers nonholonomic mobile agents. In [\[30\]](#page-10-18) and [\[31\],](#page-10-19) optimal control algorithms with asymmetrically saturated inputs were designed, but the algorithms did not consider external disturbances and were only applied to affine nonlinear systems. We designed a constrained optimal controller for robot manipulators with asymmetrically saturated inputs based on ADP in  $[32]$ ; however, the algorithm did not consider external disturbances and was only performed on simulations.

<span id="page-1-17"></span><span id="page-1-14"></span>To the best of our knowledge, the problem of  $H_{\infty}$ optimal control for robot manipulators with asymmetric input saturation and external disturbance has not been solved. Although, the  $H_{\infty}$  performance indexes of the manipulators can be easily obtained by combining the zero-sum game theory and ADP, the indexes cannot be robust stable under asymmetrically saturated inputs. Therefore, this paper is the first work using the ADP method to design a constrained  $H_{\infty}$  optimal controller (CHOC) for robot manipulators with asymmetric input constraints and external disturbances. The main contributions of this paper are shown as follows:

- 1) Different from the works in [\[24\],](#page-10-12) [\[25\],](#page-10-13) [\[26\], a](#page-10-14)nd [\[27\],](#page-10-15) we design a new feedforward controller to convert the robust optimal tracking control problem with asymmetric input saturation for robot manipulators into a robust optimal control problem for affine nonlinear systems.
- 2) Unlike the work in  $[24]$ , we design a robust optimal control method based on ADP and game theory to determine the optimal control and disturbance compensation laws. Furthermore, the RL algorithm uses only one NN instead of two for an online estimate of the solution of the HJI equation to reduce the computational complexity.
- 3) Instead of using only the persistent excitation (PE) condition, we apply the concurrent learning (CL) technique to avoid affecting the actuator and ensure fast convergence.

<span id="page-1-9"></span><span id="page-1-8"></span>The remainder of the article is organized as follows: Section [II](#page-1-0) presents the robot manipulator's dynamics and the control objective. Section [III](#page-2-0) provides the design of the feedforward controller and the CHOC. Section  $IV$  shows the simulation and experimental results. Finally, Section [V](#page-9-11) gives the conclusion of the paper.

# <span id="page-1-10"></span><span id="page-1-0"></span>**II. PROBLEM FORMULATION**

# <span id="page-1-11"></span>A. KINEMATIC AND DYNAMIC MODEL OF ROBOT MANIPULATOR

<span id="page-1-13"></span><span id="page-1-12"></span>Consider a model of an n-link robot manipulator with asymmetric input constraints and disturbances described by the following kinematic and dynamic equations [\[33\],](#page-10-21) [\[34\]:](#page-10-22)

<span id="page-1-19"></span><span id="page-1-18"></span>
$$
P(x_P, y_P, z_P) = F(\delta),\tag{1}
$$

$$
\delta = I(x_P, y_P, z_P),\tag{2}
$$

<span id="page-1-2"></span>
$$
\mathcal{H}(\delta)\ddot{\delta} + \mathcal{C}(\delta, \dot{\delta})\dot{\delta} + \mathcal{G}(\delta) + \mathcal{F}(\dot{\delta}) = \tau - \tau_d, \qquad (3)
$$

<span id="page-1-16"></span><span id="page-1-15"></span>where  $x_P$ ,  $y_P$ , and  $z_P$  are the position values of *X*, *Y* , and *Z*, respectively, in the *OXYZ* coordinate system,  $P = [x_P, y_P, z_P]^T$  represents the endpoint position in the workspace,  $\delta \in \mathbb{R}^{n \times 1}$  is joint variable,  $F(\delta)$ :  $\mathbb{R}^n \to \mathbb{R}^3$  is a function of forward kinematics,  $I(x_P, y_P, z_P) : \mathbb{R}^3 \to \mathbb{R}^n$  is a function of inverse kinematics,  $\mathcal{H}(\delta) \in \mathbb{R}^{n \times n}$  is a symmetric and positive definite inertial matrix,  $C(\delta, \dot{\delta}) \in \mathbb{R}^{n \times n}$  is the Coriolis-centripetal matrix,  $\mathcal{G}(\delta) \in \mathbb{R}^{n \times 1}$  is gravity vector,  $\mathcal{F}(\dot{\delta}) \in \mathbb{R}^{n \times 1}$  is the static friction vector,  $\tau_d \in \mathbb{R}^{n \times 1}$  is the disturbances vector, and  $\tau = [\tau_1, \tau_2, \dots, \tau_n]^T \in \mathbb{R}^{n \times 1}$  is the control inputs vector.  $\tau$  is bounded by

<span id="page-1-1"></span>
$$
\chi_1 \leq \tau_i \leq \chi_2, i = 1, 2, \cdots, n,
$$
 (4)

with  $|\chi_1| \leq |\chi_2|$ .

*Property 1:* The matrices  $\mathcal{H}(\delta)$ ,  $\mathcal{C}(\delta, \delta)$ ,  $\mathcal{G}(\delta)$ ,  $\tau_d$  are bounded by  $m_{\text{min}} \leq \|\mathcal{H}(\delta)\| \leq m_{\text{max}}$ ,  $\|\mathcal{C}(\delta, \delta)\| \leq c_{\text{max}}$ ,  $||\mathcal{G}(\delta)|| \leq g_{\text{max}}$ ,  $||\tau_d|| \leq d_{\text{max}}$ , where  $m_{\text{min}}$ ,  $m_{\text{max}}$ ,  $c_{\text{max}}$ , *g*max, *d*max are positive constants.

<span id="page-2-1"></span>

**FIGURE 1.** Control structure diagram.

*Remark [1](#page-1-1):* Property 1 is practical by  $\delta$  and  $\dot{\delta}$  are bounded. Due to the physical limitations of the electrical drive circuit and motors, the control inputs are constrained. The case of asymmetric input constraints occurs when the actuator is partially reduced in efficiency, changes in the mechanical structure, and the faulty power drive circuit.

In preparation for the CHOC design, the system  $(3)$  is represented as a strict-feedback nonlinear system, i.e.,

$$
\begin{cases} \n\dot{\delta} = \mathcal{F}_{\delta}(\delta) + \mathcal{G}_{\delta}(\delta)\psi \\
\dot{\psi} = \mathcal{F}_{\psi}(\delta, \psi) + \mathcal{G}_{\psi}(\delta, \psi)\tau + \mathcal{K}_{\psi}(\delta, \psi)\tau_d\n\end{cases}
$$
\n(5)

where  $\mathcal{F}_{\delta}(\delta) = 0_{n \times 1}$ ,  $\mathcal{G}_{\delta}(\delta) = I_n$ ,  $\psi = \delta$ ,  $\mathcal{F}_{\psi}(\delta, \psi) =$  $-\mathcal{H}^{-1}(\delta)(\mathcal{C}(\delta,\dot{\delta})\psi + \mathcal{G}(\delta) + \mathcal{F}(\dot{\delta})) \in \mathbb{R}^{n\times 1}, \mathcal{G}_{\psi}(\delta,\psi) =$  $\mathcal{H}^{-1}(\delta) \in \mathbb{R}^{n \times n}, \mathcal{K}_{\psi}(\delta, \psi) = -\mathcal{H}^{-1}(\delta) \in \mathbb{R}^{n \times n}.$ 

*Remark 2:*  $\mathcal{F}_{\psi}(\delta, \psi)$ ,  $\mathcal{G}_{\psi}(\delta, \psi)$ , and  $\mathcal{K}_{\psi}(\delta, \psi)$  are bounded by  $||\mathcal{F}_{\psi}(\delta, \psi)|| \leq m_{\min}^{-1}$  (*c*<sub>max</sub> + *g*<sub>max</sub>) $||\psi||$ ,  $||\mathcal{G}_{\psi}(\delta, \psi)|| \leq m_{\min}^{-1}, ||\mathcal{K}_{\psi}(\delta, \psi)|| \leq m_{\min}^{-1}$ . The upper bounds of  $\mathcal{F}_{\psi}(\delta, \psi)$ ,  $\mathcal{G}_{\psi}(\delta, \psi)$ , and  $\mathcal{K}_{\psi}(\delta, \psi)$  are only intended to demonstrate stability for closed-loop system and are not used in the control law. Therefore, determining the values of the upper bounds of  $\mathcal{F}_{\psi}(\delta, \psi)$ ,  $\mathcal{G}_{\psi}(\delta, \psi)$ , and  $\mathcal{K}_{\psi}(\delta, \psi)$  is not necessary.

#### B. CONTROL OBJECTIVE

The controller's objective is to control the tracking of the desired trajectory in the workspace, ensuring that  $\lim_{t \to \infty} ||e_P|| = \lim_{t \to \infty} ||P - P_d|| = 0$  when the external disturbances are zeros, with  $P_d = [x_{Pd}, y_{Pd}, z_{Pd}]^T$  being the desired position in the workspace. In the first step, we use inverse kinematic equations to convert the desired trajectory in the workspace to joint space. The CHOC is implemented in the second step to ensure tracking the desired trajectory for robot manipulators with asymmetric input constraints in the joint space, i.e., providing  $\lim_{t\to\infty} \|\delta(t) - \delta_{ref}(t)\| = 0$  when the external disturbances are zeros, where  $\delta_{ref}(t) \in \mathbb{R}^{n \times 1}$  is the reference trajectory in the joint space. However, the external disturbances are non-zeros; thus, the objective is to design the control law to make the tracking errors  $L_2$ -bounded [\[35\].](#page-10-23) We use forward kinematic equations to convert the trajectory from joint space to workspace. These steps are described in Fig. [1.](#page-2-1)

<span id="page-2-10"></span>*Assumption 1:*  $\delta_{ref}(t)$  is smooth and bounded.

# <span id="page-2-0"></span>**III. ASYMMETRICALLY CONSTRAINED** H<sup>∞</sup> **OPTIMAL CONTROL**

This section designs an CHOC for the robot manipulator. First, the feedforward control law is designed to convert the problem of  $H_{\infty}$  optimal tracking control for the robot manipulator into an issue of  $H_{\infty}$  optimal control for an affine nonlinear system. Then, the asymmetrically constrained  $H_{\infty}$ optimal control law is designed based on ADP method combined with the game theory.

# A. FEEDFORWARD CONTROL

In this section, we design feedforward control inputs, then transform the system  $(5)$  into an affine nonlinear system, known as an augmented system. The design steps are developed based on the backstepping technique [\[36\].](#page-10-24)

*Step 1:* Define the tracking error as  $e_{\delta} = \delta - \delta_{ref}$ ,  $e_{\psi} =$  $\psi - \psi_{ref}$ . Then, the dynamics of the position tracking errors are presented as

<span id="page-2-12"></span>
$$
\dot{e}_{\delta} = -\dot{\delta}_{ref} + \mathcal{G}_{\delta}(\delta)\psi_{ref}^{*} + \mathcal{G}_{\delta}(\delta)\psi_{ref}^{a} + \mathcal{G}_{\delta}(\delta)e_{\psi}, \quad (6)
$$

<span id="page-2-2"></span>where  $\psi_{ref} = \psi_{ref}^* + \psi_{ref}^a$  are the virtual control inputs,  $\psi_{ref}^*$ are the virtual optimal control inputs,  $\psi_{ref}^a$  are the feedforward virtual control inputs designed as

<span id="page-2-8"></span><span id="page-2-4"></span><span id="page-2-3"></span>
$$
\psi_{ref}^a = \mathcal{G}_{\delta}^{-1}(\delta)(\dot{\delta}_{ref} + K_1 e_{\psi} - K_2 e_{\delta}). \tag{7}
$$

By substituting  $(7)$  into  $(6)$ , the dynamics  $(6)$  is rewritten as

<span id="page-2-6"></span>
$$
\dot{e}_{\delta} = \mathcal{G}_{\delta}(\delta)\psi_{ref}^{*} + K_{1}e_{\psi} - K_{2}e_{\delta} + \mathcal{G}_{\delta}(\delta)e_{\psi}
$$
  
=  $\bar{\mathcal{F}}_{\delta}(e_{\delta}, e_{\psi}) + \psi_{ref}^{*} + e_{\psi},$  (8)

where  $\bar{\mathcal{F}}_{\delta}(e_{\delta}, e_{\psi}) = K_1 e_{\psi} - K_2 e_{\delta}, K_1$  and  $K_2$  are the positive definite matrices.

*Step 2:* The dynamics of the angular velocity tracking errors are calculated as follows:

$$
\dot{e}_{\psi} = -\dot{\psi}_{ref} + \mathcal{F}_{\psi}(\delta, \psi) + \mathcal{G}_{\psi}(\delta, \psi)\tau^* + \mathcal{G}_{\psi}(\delta, \psi)\tau^a
$$
  
+  $\mathcal{K}_{\psi}(\delta, \psi)\tau_d$ , (9)

where  $\tau = \tau^* + \tau^a$ ,  $\tau^*$  are the actual optimal control inputs,  $\tau^a$  are the feedforward actual control inputs designed as

<span id="page-2-5"></span>
$$
\tau^{a} = \mathcal{G}_{\psi}^{-1}(\delta, \psi)(\dot{\psi}_{ref} + \mathcal{F}_{\psi}(e_{\delta}, e_{\psi}) - \mathcal{F}_{\psi}(\delta, \psi) - e_{\delta} - K_{3}e_{\delta} - K_{4}\dot{e}_{\delta}). \tag{10}
$$

By substituting  $(10)$  into  $(9)$ , the dynamics  $(9)$  is rewritten as

<span id="page-2-9"></span>
$$
\dot{e}_{\psi} = \bar{\mathcal{F}}_{\psi}(e_{\delta}, e_{\psi}) + \mathcal{G}_{\psi}(\delta, \psi)\tau^* - e_{\delta} + \mathcal{K}_{\psi}(\delta, \psi)\tau_d, (11)
$$

where  $\bar{\mathcal{F}}_{\psi}(e_{\delta}, e_{\psi}) = \mathcal{F}_{\psi}(e_{\delta}, e_{\psi}) - K_3 e_{\delta} - K_4 \dot{e}_{\delta}$ ,  $K_3$  and  $K_4$ are the positive definite matrices.

*Lemma 1:* Consider the following augmented system:

<span id="page-2-7"></span>
$$
\dot{\mathcal{Z}} = \bar{\mathcal{F}}_{\delta\psi}(e_{\delta}, e_{\psi}) + \mathcal{G}_{\delta\psi}(\delta, \psi)\mu^* + \mathcal{K}_{\delta\psi}(\delta, \psi)d, \quad (12)
$$

<span id="page-2-11"></span>where

 $\mathcal{Z} = [e_{\delta}^T, e_{\psi}^T]^T, \bar{\mathcal{F}}_{\delta\psi}(e_{\delta}, e_{\psi}) = [\bar{\mathcal{F}}_{\delta}^T(e_{\delta}, e_{\psi}), \bar{\mathcal{F}}_{\psi}^T(e_{\delta}, e_{\psi})]^T,$  $\mathcal{G}_{\delta\psi}(\delta, \psi) = \text{diag} \left[ \mathcal{G}_{\delta}(\delta), \mathcal{G}_{\psi}(\delta, \psi) \right], \quad \mathcal{K}_{\delta\psi}(\delta, \psi) = \text{diag} \left[ \mathcal{G}_{\delta}(\delta), \mathcal{G}_{\psi}(\delta, \psi) \right], \quad \mathcal{K}_{\delta\psi}(\delta, \psi) = \text{diag} \left[ \mathcal{G}_{\delta}(\delta), \mathcal{G}_{\psi}(\delta, \psi) \right],$  $diag[0_{n\times n}, \mathcal{K}_{\psi}(\delta, \psi)], \mu^* = [\psi_{ref}^{*T}, \tau^{*T}]^T, \mu^a =$  $[\psi_{ref}^{aT}, \tau^{aT}]^{T}, \quad \mu = [\psi_{ref}^{T}, \tau^{T}]^{T} = \mu^{*} + \mu^{a},$ 

 $d = [0_{1 \times n}, \tau_d^T]^T$ . Suppose the feedforward virtual control inputs are presented as [\(7\),](#page-2-3) and the feedforward actual control inputs are presented as [\(10\).](#page-2-5) Then, the  $H_{\infty}$ optimal control issue for the dynamics  $(5)$  and dynamics  $(12)$ are equivalent.

*Proof:* For dynamics [\(5\),](#page-2-2) we select the Lyapunov function form

$$
\mathfrak{J}_1 = \frac{1}{2} e_\delta^T e_\delta + \frac{1}{2} e_\psi^T e_\psi.
$$
 (13)

Taking derivative of  $(13)$  and combining  $(8)$  and  $(11)$  yields

$$
\dot{\mathfrak{J}}_1 = e_\delta^T \bar{\mathcal{F}}_\delta(e_\delta, e_\psi) + e_\psi^T \bar{\mathcal{F}}_\psi(e_\delta, e_\psi) + e_\delta^T \psi_{ref}^*
$$
\n
$$
+ e_\psi^T \mathcal{G}_\psi(\delta, \psi) \tau^* + e_\psi^T \mathcal{K}_\psi(\delta, \psi) \tau_d + e_\delta^T e_\psi - e_\psi^T e_\delta. \tag{14}
$$

It can be seen that  $e_{\delta}^{T}e_{\psi} = e_{\psi}^{T}e_{\delta}$ . Thus, [\(14\)](#page-3-1) becomes

$$
\dot{\mathfrak{J}}_1 = \mathcal{Z}^T(\bar{\mathcal{F}}_{\delta\psi}(e_\delta, e_\psi) + \mathcal{G}_{\delta\psi}(\delta, \psi)\mu^* + \mathcal{K}_{\delta\psi}(\delta, \psi)d). \tag{15}
$$

For dynamics [\(12\),](#page-2-7) we select the Lyapunov function form

$$
\mathfrak{J}_2 = \frac{1}{2} \mathcal{Z}^T \mathcal{Z}.
$$
 (16)

Taking derivative  $(16)$  and combining  $(12)$ , one has

$$
\dot{\mathfrak{J}}_2 = \mathcal{Z}^T(\bar{\mathcal{F}}_{\delta\psi}(e_\delta, e_\psi) + \mathcal{G}_{\delta\psi}(\delta, \psi)\mu^* + \mathcal{K}_{\delta\psi}(\delta, \psi)d). \tag{17}
$$

It can be easily seen that, if  $\mu^*$  makes the dynamics [\(12\)](#page-2-7) stable, then  $\dot{J}_2$  < 0. Through comparison of [\(17\)](#page-3-3) and [\(15\),](#page-3-4)  $\dot{J}_2$  < 0 implies  $\dot{J}_1$  < 0. It can be concluded that the dynamics [\(5\)](#page-2-2) is also stable. Therefore, the  $H_{\infty}$  optimal control issue of the dynamics [\(5\)](#page-2-2) is converted to the  $H_{\infty}$  optimal control issue for dynamics [\(12\).](#page-2-7)

This completes the proof.

*Remark 3:* The feedforward virtual control inputs and the feedforward actual control inputs are designed as  $(7)$  and  $(10)$ , and constrained by

$$
\begin{cases}\n\chi_1 - \nu_1 \tanh(1) \leq \psi_{\text{refi}}^a \leq \chi_2 - \nu_2 \tanh(1) \\
\chi_1 - \nu_1 \tanh(1) \leq \tau_i^a \leq \chi_2 - \nu_2 \tanh(1)\n\end{cases}
$$
\n(18)

where  $i = 1, 2, \dots, n$ ,  $|v_1| < |\chi_1|$ ,  $|v_2| < |\chi_2|$ . We can see that if  $\psi^*_{\text{refi}}$  and  $\tau^*_{i}$  are designed to satisfy the constraint  $\nu_1 \leq$  $\psi_{refi}^* \leq \nu_2$ ,  $\nu_1 \leq \tau_i^* \leq \nu_2$ , then  $\nu_{refi}$  and  $\tau_i$  will be limited by  $\chi_1 \leq \psi_{\text{ref}} \leq \chi_2, \chi_1 \leq \tau_i \leq \chi_2.$ 

# B.  $H_{\infty}$  OPTIMAL CONTROL

This section designs the asymmetrically constrained  $H_{\infty}$ optimal control law for dynamics [\(12\)](#page-2-7) based on the ADP method combined with the game theory [\[22\]. T](#page-10-10)o derive the controller, we give the following definitions.

<span id="page-3-11"></span>*Definition 1* [\[37\]:](#page-10-25) Consider the  $\mathcal{Z} = \mathcal{F}(\mathcal{Z}, \mu), \mathcal{Z} \in$  $\mathbb{R}^n$  with  $\mathcal{Z}_0$  being the equilibrium point.  $\mathcal{Z}_0$  is said to be ultimately uniformly bounded (UUB) if there exists a compact set  $\mathcal{O} \in \mathbb{R}^n$  so that for all  $\mathcal{Z}_0 \in \mathcal{O}$ , there exists a

bound *C* and time  $T(C, \mathcal{Z}_0)$  such that  $\|\mathcal{Z} - \mathcal{Z}_0\| \leq C$  for all  $t > t_0 + T$ .

*Definition 2*  $[21]$ : The system  $(12)$  has  $L_2$ -gain less than or equal  $\rho \ \forall d \in L_2 [0, \infty], 0 \leq T \leq \infty$  if the following expression is satisfied:

$$
\int_{0}^{T} (\mathcal{Z}^{T} \Theta \mathcal{Z} + \mathcal{F}(\mu)) d\tau \leq \rho^{2} \int_{0}^{T} (d^{T} d) d\tau, \qquad (19)
$$

<span id="page-3-0"></span>where  $\Theta$  is the positive definite matrix,  $F(\mu)$  is the energy cost function and non-negative,  $\mu$  is an estimate of  $\mu^*$  at time *t*, and  $\rho > 0$  is the attenuation level of the disturbance.

<span id="page-3-1"></span>Based on the condition [\(19\),](#page-3-5) the cost function is presented as

<span id="page-3-12"></span><span id="page-3-9"></span><span id="page-3-6"></span><span id="page-3-5"></span>
$$
\mathcal{V}\left(\mathcal{Z}\right) = \int_{t}^{\infty} \kappa(\tau) d\tau,\tag{20}
$$

<span id="page-3-4"></span>where  $\kappa(t) = \mathcal{Z}^T \Theta \mathcal{Z} + \mathcal{F}(\mu) - \rho^2 d^T d$ .

<span id="page-3-2"></span>By the control inputs are asymmetrically saturated, the energy cost function is chosen as [\[30\]](#page-10-18) and [\[40\]](#page-10-26)

$$
F(\mu) = (\nu_2 - \nu_1) \int_{\frac{\nu_2 + \nu_1}{2}}^{\mu} \tanh^{-T} \left( \frac{2s - \nu_2 - \nu_1}{\nu_2 - \nu_1} \right) R ds,
$$
\n(21)

<span id="page-3-3"></span>where  $R = diag(R_1, ..., R_n) > 0$ . Defining  $\varrho = (v_2 + v_1)/2$ and  $\zeta = (\nu_2 - \nu_1)/2$ , [\(21\)](#page-3-6) is rewritten as

$$
F(\mu) = 2\varsigma \int_{\varrho}^{\mu} \tanh^{-T} \left( \frac{s - \varrho}{\varsigma} \right) R ds. \tag{22}
$$

We apply integral by parts and perform some conversion steps. We can derive that

$$
F(\mu) = 2\zeta \tanh^{-T}(\frac{\mu - \varrho}{\varsigma})R(\mu - \varsigma) + \varsigma^{2}\bar{R}\ln(\bar{1} - (\frac{\mu - \varrho}{\varsigma}^{2}),
$$
\n(23)

with  $\bar{R} = [R_1, \ldots, R_n], \bar{1} = [1, \ldots, 1]^T \in \mathbb{R}^{n \times 1}$ .

Consider the Hamilton function, which has the following form:

$$
\mathcal{H}(\mathcal{Z}, \mu, d, \mathcal{V}_{\mathcal{Z}}) = \mathcal{Z}^T \Theta \mathcal{Z} + F(\mu) - \rho^2 d^T d + \mathcal{V}_{\mathcal{Z}}^T (\bar{\mathcal{F}}_{\delta \psi} + \mathcal{G}_{\delta \psi} \mu + \mathcal{K}_{\delta \psi} d), \quad (24)
$$

with  $V_z = \partial V(z) / \partial Z$ . The optimal value function  $V^*(z)$  is determined by using zero-sum game theory:

<span id="page-3-8"></span><span id="page-3-7"></span>
$$
\mathcal{V}^*(\mathcal{Z}) = \min_{\mu} \max_{d} \int_0^\infty \kappa(\tau) d\tau.
$$
 (25)

ADP and the zero-sum game theory are used to find the optimal value  $\mathcal{V}^*(\mathcal{Z})$  that satisfies the Nash condition

<span id="page-3-10"></span>
$$
\mathcal{V}^*(\mathcal{Z}) = \min_{\mu} \max_{d} \int_0^\infty \kappa(\tau) d\tau = \max_{d} \min_{\mu} \int_0^\infty \kappa(\tau) d\tau.
$$
\n(26)

Then, the saddle point  $(\mu^*, d^*)$  exists, where  $\mu^*$  is the optimal control law,  $d^*$  is the disturbance compensation law. To ensure the Nash condition  $(26)$ , the following Isaacs condition must be satisfied:

$$
\min_{\mu} \max_{d} \mathcal{H}(\mathcal{Z}, \mu, d, \mathcal{V}_{\mathcal{Z}}^{*}) = \max_{d} \min_{\mu} \mathcal{H}(\mathcal{Z}, \mu, d, \mathcal{V}_{\mathcal{Z}}^{*}).
$$
\n(27)

The saddle point  $(\mu^*, d^*)$  is determined through the stationary conditions  $\partial \mathcal{H}(\mathcal{Z}, \mu, d, V^*_{\mathcal{Z}})/\partial \mu = 0$ ,  $\partial \mathcal{H}(\mathcal{Z}, \mu, d, \mathcal{V}_{\mathcal{Z}}^*)/\partial d = 0$ . Hence, we obtain  $\mu^*$  and  $d^*$  as follows:

$$
\mu^* = -\varsigma \tanh(\frac{1}{2\varsigma}R^{-1}\mathcal{G}_{\delta\psi}^T \mathcal{V}_{Z}^*) + \varrho, \tag{28}
$$

$$
d^* = \frac{1}{2\rho} \mathcal{K}_{\delta\psi}^T \mathcal{V}_{\mathcal{Z}}^*,\tag{29}
$$

where  $V_Z^* = \partial V^*(\mathcal{Z}) / \partial \mathcal{Z}$ .

Substituting  $(28)$  and  $(29)$  into  $(24)$ , we obtain the HJI equation as follows:

$$
\mathcal{H}^* \mathcal{Z}, \mu^*, d^*, \mathcal{V}_{\mathcal{Z}}^* = \mathcal{Z}^T \Theta \mathcal{Z} + F(\mu^*) - \rho^2 d^{*T} d^* + \mathcal{V}_{\mathcal{Z}}^{*T} (\bar{\mathcal{F}}_{\delta\psi} + \mathcal{G}_{\delta\psi} \mu^* + \mathcal{K}_{\delta\psi} d^*) = 0.
$$
\n(30)

If  $V^*(\mathcal{Z})$  of the HJI equation [\(30\)](#page-4-2) is found, one can determine  $\mu^*$  and  $d^*$ . The minimal positive definite smooth solutions  $V^*(\mathcal{Z})$  always exist [\[21\]. H](#page-10-9)owever, it cannot be obtained by analytical methods. Therefore, we need to estimate the solution of equation [\(30\).](#page-4-2) With the requirement of a simple online approximation algorithm, a simple NN is applied and presented as follows:

$$
\mathcal{V}^*(\mathcal{Z}) = \varpi^T \zeta(\mathcal{Z}) + \varepsilon(\mathcal{Z}),\tag{31}
$$

$$
\mathcal{V}_{\mathcal{Z}}^* = \varpi^T \frac{\partial \zeta(\mathcal{Z})}{\partial \mathcal{Z}} + \frac{\partial \varepsilon(\mathcal{Z})}{\partial \mathcal{Z}} = \varpi^T \zeta_{\mathcal{Z}} + \varepsilon_{\mathcal{Z}},\qquad(32)
$$

where NN has a structure with one hidden layer and one output,  $\varpi \in \mathbb{R}^{hc}$  is the weight vector,  $\zeta(\mathcal{Z}) : \mathbb{R}^n \to \mathbb{R}^{hc}$  is a smooth function vector and a completely independent basis set that satisfies Assumption [2,](#page-4-3) *hc* is a number of hidden cells,  $\varepsilon(\mathcal{Z})$  is the approximation error.

<span id="page-4-3"></span>*Assumption 2 [\[39\]:](#page-10-27)*  $\|\zeta(\mathcal{Z})\| \leq s_{\zeta}, \|\zeta_{\mathcal{Z}}\|$  $||\partial \zeta (Z)/\partial Z||$  ≤  $s_{\nabla \zeta}$ ,  $||\varepsilon (Z)||$  ≤  $s_{\varepsilon}$ ,  $||\varepsilon_Z||$  $\|\partial \varepsilon(\mathcal{Z}) / \partial \mathcal{Z}\|$  ≤  $s_{\nabla \varepsilon}$ , where  $s_{\zeta}$ ,  $s_{\nabla \zeta}$ ,  $s_{\varepsilon}$ ,  $s_{\nabla \varepsilon}$  are positive constants.

The Hamilton function [\(24\)](#page-3-8) becomes

$$
\mathcal{H}(\mathcal{Z}, \mu^*, d^*, \varpi^T \zeta_{\mathcal{Z}}) = \mathcal{Z}^T \Theta \mathcal{Z} + F(\mu^*) - \rho^2 d^{*T} d^* \n+ \varpi^T \zeta_{\mathcal{Z}} (\bar{\mathcal{F}}_{\delta \psi} + \mathcal{G}_{\delta \psi} \mu^* + \mathcal{K}_{\delta \psi} d^*) \n= \varepsilon_H,
$$
\n(33)

where  $\varepsilon_H = -\varepsilon_{\mathcal{Z}}^T(\bar{\mathcal{F}}_{\delta\psi} + \mathcal{G}_{\delta\psi}\mu^* + \mathcal{K}_{\delta\psi}d^*).$ 

As  $\varpi$  are unknown, the value function is given as follows:

$$
\hat{\mathcal{V}}(\mathcal{Z}) = \hat{\boldsymbol{\varpi}}^T \zeta(\mathcal{Z}).\tag{34}
$$

Using  $\hat{V}(Z)$  for [\(28\)](#page-4-0) and [\(29\),](#page-4-1) the approximate control laws of [\(28\)](#page-4-0) and [\(29\)](#page-4-1) are expressed as

$$
\hat{\mu} = -\varsigma \tanh(\frac{1}{2\varrho}R^{-1}\mathcal{G}_{\delta\psi}^T\zeta_{\mathcal{Z}}^T\hat{\varpi}) + \varrho. \tag{35}
$$

<span id="page-4-11"></span>
$$
. . . . . . . . . .
$$

<span id="page-4-8"></span><span id="page-4-5"></span>
$$
\hat{d} = \frac{1}{2\rho} \mathcal{K}_{\delta\psi}^T \zeta_{\mathcal{Z}}^T \hat{\varpi}.
$$
\n(36)

Similarly, the Hamiltonian function becomes

$$
\mathcal{H}(\mathcal{Z}, \hat{\mu}, \hat{d}, \hat{\omega}^T \zeta \mathcal{Z}) = \mathcal{Z}^T \Theta \mathcal{Z} + F(\hat{\mu}) - \rho^2 \hat{d}^T \hat{d} \n+ \hat{\omega}^T \zeta \mathcal{Z} (\bar{\mathcal{F}}_{\delta \psi} + \mathcal{G}_{\delta \psi} \hat{\mu} + \mathcal{K}_{\delta \psi} \hat{d}) \n= \varepsilon_e,
$$
\n(37)

where  $\varepsilon_e$  is the Hamilton function approximation error. From  $(33)$  and  $(37)$ , we can obtain

$$
\varepsilon_e = -\tilde{\omega}^T \zeta_{\mathcal{Z}} (\bar{\mathcal{F}}_{\delta \psi} + \mathcal{G}_{\delta \psi} \hat{\mu} + \mathcal{K}_{\delta \psi} \hat{d}) + \varepsilon_H, \qquad (38)
$$

<span id="page-4-1"></span><span id="page-4-0"></span>where  $\tilde{\varpi} = \varpi - \hat{\varpi}$ . If we propose a parameter tuning law to minimize  $E = \frac{1}{2} \varepsilon_e^T \varepsilon_e$ , then  $\hat{\omega} \to \overline{\omega}$ . Applying the algorithm gradient descent combined with the CL technique, the parameter tuning law designed as

<span id="page-4-6"></span>
$$
\dot{\hat{\omega}} = -\alpha \frac{\Psi}{\left(\Psi^T \Psi + 1\right)^2} (\Psi^T \hat{\omega} + \kappa(t))
$$

$$
-\alpha \sum_{i=1}^P \frac{\Psi(t_i)}{\left(\Psi(t_i)^T \Psi(t_i) + 1\right)^2} (\Psi(t_i)^T \hat{\omega} + \kappa(t_i)),
$$
(39)

<span id="page-4-2"></span>where  $\kappa(t) = \mathcal{Z}^T \Theta \mathcal{Z} + F(\hat{\mu}) - \rho^2 \hat{d}^T \hat{d}, \Psi = \zeta \mathcal{Z} (\bar{\mathcal{F}}_{\delta \psi} + \bar{\mathcal{F}}_{\delta \psi})$  $\mathcal{G}_{\delta\psi}\hat{\mu} + \mathcal{K}_{\delta\psi}\hat{d}$ ,  $\alpha > 0$  is the learning rate. We use the CL technique in [\(39\)](#page-4-6) to relax the PE condition and ensure convergence fast.  $\Psi(t_i)$ ,  $\kappa(t_i)$  are stored in  ${\Psi(t_i), \kappa(t_i)}_{i=1}^P$ , with  ${\Psi(t_i)}_{i=1}^P$  must be linearly independent, i.e. *rank*( ${\Psi(t_1)}$ ),  $\Psi(t_2), \ldots, \Psi(t_p) = P.$ 

The approximate error dynamics of the weights can be determined as follows:

<span id="page-4-10"></span><span id="page-4-9"></span>
$$
\dot{\tilde{\varpi}} = -\alpha \bar{\Psi} (\bar{\Psi}^T \tilde{\varpi} - \bar{\varepsilon}_e) - \alpha \sum_{i=1}^P \bar{\Psi}(t_i) (\bar{\Psi}^T(t_i) \tilde{\varpi} - \bar{\varepsilon}_e(t_i)),
$$
\n(40)

with  $\bar{\Psi} = \Psi / (\Psi^T \Psi + 1), \|\bar{\varepsilon}_e(s)\| = \|\varepsilon_H / (1 + \Psi^T \Psi)\| \leq$  $s_{\bar{\varepsilon}}$ , where  $s_{\bar{\varepsilon}}$  is a positive constant.

#### C. PROOF OF STABILITY AND CONVERGENCE

The following theorem analyzes the stability and convergence of the CHOC.

*Theorem 1:* Consider the tracking error dynamics [\(12\)](#page-2-7) with asymmetrically constrained inputs and disturbances. Let Assumptions [1](#page-2-10)[-2](#page-4-3) be satisfied. Assuming the cost function is defined in  $(20)$ , the optimal control law is determined by  $(35)$ , and the disturbance compensation law is defined by  $(36)$ , where the on-line adjustment of the values is determined by [\(39\).](#page-4-6) Then, the tracking and approximation errors are UUB.

<span id="page-4-4"></span>*Proof:* Select a Lyapunov function as

$$
\mathfrak{J}_3 = \underbrace{\mathcal{V}^*(\mathcal{Z})}_{\mathfrak{J}_{31}} + \underbrace{\frac{1}{2}trace\left(\tilde{\varpi}^T\tilde{\varpi}\right)}_{\mathfrak{J}_{32}}.
$$
 (41)

<span id="page-4-7"></span>Taking derivative  $\mathfrak{J}_{31}$  along trajectories of  $\dot{\mathcal{Z}} = \bar{\mathcal{F}}_{\delta \psi} + \bar{\mathcal{F}}_{\delta \psi}$  $\mathcal{G}_{\delta\psi}\hat{\mu} + \mathcal{K}_{\delta\psi}\hat{d}$ , we can get

$$
\dot{\mathfrak{J}}_{31} = \mathcal{V}_{\mathcal{Z}}^{*T} \dot{\mathcal{Z}} = \mathcal{V}_{\mathcal{Z}}^{*T} (\bar{\mathcal{F}}_{\delta\psi} + \mathcal{G}_{\delta\psi}\hat{\mu} + \mathcal{K}_{\delta\psi}\hat{d})
$$

$$
= \mathcal{V}_{\mathcal{Z}}^{*T}(\bar{\mathcal{F}}_{\delta\psi} + \mathcal{G}_{\delta\psi}\mu^* + \mathcal{K}_{\delta\psi}d^*) + \mathcal{V}_{\mathcal{Z}}^{*T}\mathcal{G}_{\delta\psi}(\hat{\mu} - \mu^*) + \mathcal{V}_{\mathcal{Z}}^{*T}\mathcal{K}_{\delta\psi}(\hat{d} - d^*).
$$
 (42)

Observing  $(30)$  and  $(32)$ ,  $(42)$  becomes

$$
\dot{3}_{31} = -Z^T \Theta Z - F(\mu^*) + \rho^2 d^{*T} d^*
$$
  
+  $\mathcal{G}_{\delta\psi} \left( \zeta_Z^T \omega + \varepsilon_Z \right) (\hat{\mu} - \mu^*)$   
+  $\mathcal{K}_{\delta\psi} \left( \zeta_Z^T \omega + \varepsilon_Z \right) (\hat{d} - d^*).$  (43)

Using Young's inequality to  $(43)$ , we can derive that

$$
\dot{\mathfrak{J}}_{31} \leq -\mathcal{Z}^T \Theta \mathcal{Z} - \mathcal{F} \left( \mu^* \right) + \rho^2 d^{*T} d^*
$$
\n
$$
+ \frac{1}{4} \left\| \mathcal{G}_{\delta \psi} \left( \zeta_Z^T \varpi + \varepsilon_Z \right) \right\|^2 + \left\| \hat{\mu} - \mu^* \right\|^2
$$
\n
$$
+ \frac{1}{4} \left\| \mathcal{K}_{\delta \psi} \left( \zeta_Z^T \varpi + \varepsilon_Z \right) \right\|^2 + \left\| \hat{d} - d^* \right\|^2. \tag{44}
$$

On the other hand, we can gain  $\mathcal{Z}^T \Theta \mathcal{Z} \geq \lambda_{\min}(\Theta) ||\mathcal{Z}||^2$ , with  $\lambda$  (.) is an eigenvalue of the matrix. Then, [\(44\)](#page-5-3) is rewritten as follows:

$$
\dot{\mathfrak{J}}_{31} \leq -\lambda_{\min}(\Theta) \|Z\|^2 + \rho^2 \left\|\hat{a}\right\|^2
$$
\n
$$
+ \frac{1}{4} \left\| \mathcal{G}_{\delta\psi} \left( \zeta_z^T \varpi + \varepsilon_z \right) \right\|^2 + \left\|\hat{\mu} - \mu^*\right\|^2
$$
\n
$$
+ \frac{1}{4} \left\| \mathcal{K}_{\delta\psi} \left( \zeta_z^T \varpi + \varepsilon_z \right) \right\|^2 + \left\|\hat{a} - d^*\right\|^2 - \mathcal{F} \left( \mu^* \right). \tag{45}
$$

Applying inequality  $(x + y)^2 \le 2(x^2 + y^2)$ , we can derive that

$$
\begin{split} \dot{\mathfrak{J}}_{31} &\leq -\lambda_{\min} \left( \Theta \right) \| \mathcal{Z} \|^2 + \rho^2 \left\| \hat{d} \right\|^2 \\ &+ \frac{1}{2} (\left\| \mathcal{G}_{\delta \psi} \zeta_Z^T \varpi \right\|^2 + \left\| \mathcal{G}_{\delta \psi} \varepsilon_Z \right\|^2) + \left\| 2 \zeta \tanh(\hat{\xi}) \right\|^2 \\ &+ \left\| 2 \zeta \tanh(\xi^*) \right\|^2 + \frac{1}{2} \left\| \mathcal{K}_{\delta \psi} \right\|^2 \left\| \zeta_Z \right\|^2 \left\| \varpi \right\|^2 \\ &+ \frac{1}{2} \left\| \mathcal{G}_{\delta \psi} \varepsilon_Z \right\|^2 + \frac{1}{2\rho^2} \left\| \mathcal{K}_{\delta \psi} \right\|^2 \left\| \zeta_Z \right\|^2 \left\| \tilde{\varpi} \right\|^2 \\ &+ \frac{1}{2\rho^2} \left\| \mathcal{K}_{\delta \psi} \right\|^2 \left\| \varepsilon_Z \right\|^2 - \mathcal{F} \left( \mu^* \right), \end{split} \tag{46}
$$

where  $\hat{\xi} = \frac{1}{2\varsigma} R^{-1} \mathcal{G}_{\delta\psi}^T \zeta \frac{T}{2} \hat{\omega}, \xi^* = \frac{1}{2\varsigma} R^{-1} \mathcal{G}_{\delta\psi}^T (\zeta \frac{T}{2} \omega + \varepsilon_{\mathcal{Z}}).$ We can easily deduce that  $F(\mu^*) \geq 0^5$ , [\(46\)](#page-5-4) becomes

$$
\dot{\mathfrak{J}}_{31} \le -\lambda_{\min}(\Theta) \left\| \mathcal{Z} \right\|^2 + \eta_2 \|\tilde{\varpi}\|^2 + \eta_1,\qquad(47)
$$

where

 $\eta_1 = \frac{1}{2} \mathcal{G}_{\delta\psi}^2 \arctan(\frac{s_{\zeta\zeta}^2}{\sigma} + \frac{s_{\varepsilon}^2}{\sigma} + \frac{s_{\zeta}^2}{\sigma^2}) + 8\varsigma^2 + \frac{1}{2} \mathcal{K}_{\delta\psi}^2 \arctan(\frac{3}{2} s_{\zeta\zeta}^2 s_{\varpi}^2 + \frac{1}{2} \mathcal{K}_{\delta\psi}^2 s_{\zeta\zeta}^2 s_{\zeta\zeta}^2)$  $s_{\varepsilon}^2 + \frac{1}{2\rho}$  $\frac{1}{2ρ^2}$ s<sup>2</sup><sub>ε</sub>), η<sub>2</sub> =  $\frac{1}{2ρ}$  $\frac{1}{2\rho^2} \mathcal{K}_{\delta\psi\max}^2 s_{\nabla\varpi}^2$ ,  $\|\varpi\| \leq s_{\varpi}$ ,  $s_{\varpi} > 0$ ,  $\mathcal{G}_{\delta\psi \text{ max}} = \max\{1, m_{\text{min}}^{-1}\}, \mathcal{K}_{\delta\psi \text{ max}} = m_{\text{min}}^{-1}.$ 

Taking derivative  $\mathfrak{J}_{32}$  along [\(40\)](#page-4-10) we have

$$
\dot{\mathfrak{J}}_{32} = -\alpha \tilde{\omega}^T \Xi \tilde{\omega} + \alpha \tilde{\omega}^T (\bar{\Psi} \bar{\varepsilon}_e + \sum_{i=1}^P \bar{\Psi}(t_i) \bar{\varepsilon}_e(t_i)), \tag{48}
$$

<span id="page-5-8"></span><span id="page-5-1"></span>

<span id="page-5-2"></span>**FIGURE 2.** Actual Scorbot-ER.

<span id="page-5-3"></span>where  $\Xi = \bar{\Psi}\bar{\Psi}^T + \sum_{i=1}^P \bar{\Psi}(t_i)\bar{\Psi}^T(t_i) > 0$ . Applying Young's inequality to [\(48\),](#page-5-5) we can deduce that

$$
\dot{\mathfrak{J}}_{32} \leq -(\alpha - 1)\lambda_{\min}(\Xi) \|\tilde{\varpi}\|^2 + \frac{\alpha^2}{4}(P+1)s_{\tilde{\varepsilon}}^2. \tag{49}
$$

Observing [\(47\)](#page-5-6) and [\(49\),](#page-5-7)  $\dot{\mathfrak{J}}_3$  becomes

<span id="page-5-7"></span>
$$
\dot{\mathfrak{J}}_3 \leq -\lambda_{\min}(\Theta) \| \mathcal{Z} \|^2 - \eta_3 \| \tilde{\varpi} \|^2 + \eta_4,\tag{50}
$$

where  $\eta_3 = (\alpha - 1)\lambda_{min}(\Xi) - \eta_2$ ,  $\eta_4 = \eta_1 + \frac{\alpha^2}{4}$  $\frac{\gamma^2}{4}(P+1)s_{\bar{\varepsilon}}^2,$  $\alpha > \frac{\eta_2}{\lambda_{min}(\Xi)} + 1.$ 

 $\dot{J}_3 < 0$ , if and only if

$$
\begin{cases}\n\|\mathcal{Z}\| > \sqrt{\frac{\eta_4}{\lambda_{\min}(\Theta)}} = s_{\mathcal{Z}} \\
\|\tilde{\varpi}\| > \sqrt{\frac{\eta_4}{\eta_3}} = s_{\tilde{\varpi}}\n\end{cases} \tag{51}
$$

It can be concluded that if the tracking errors ∥Z∥ or the approximation errors  $\|\tilde{\varpi}\|$  exceed stable regions, then  $\dot{\mathfrak{J}}_3$  < 0, i.e., the tracking errors or the approximation errors are pulled inside the stable regions. In other words, the tracking and approximation errors are UUB [\[37\]](#page-10-25) (see Definition [1\)](#page-3-10).

<span id="page-5-4"></span>This completes the proof.

# <span id="page-5-0"></span>**IV. SIMULATION AND EXPERIMENT**

<span id="page-5-6"></span>This section verifies the performance of the CHOC through simulation and experiment. The results are compared with the robust adaptive controller (RAC) [\[8\].](#page-9-7)

Consider the Scorbot-ER robotic manipulator [\[38\], i](#page-10-28)llus-trated in Figs. [2,](#page-5-8) [3,](#page-6-0) where  $l_1 = 0.35$ m,  $l_e = 0.025$ m,  $l_2 = 0.222$ m,  $l_3 = 0.222$ m. The position of *P* is determined by

<span id="page-5-9"></span><span id="page-5-5"></span>
$$
P = \begin{bmatrix} x_P \\ y_P \\ z_P \end{bmatrix} = \begin{bmatrix} (l_3 \cos \delta_3 + l_2 \cos \delta_2 + l_e) \cos \delta_1 \\ (l_3 \cos \delta_3 + l_2 \cos \delta_2 + l_e) \sin \delta_1 \\ l_3 \sin \delta_3 + l_2 \sin \delta_2 + l_1 \end{bmatrix}.
$$
 (52)

<span id="page-6-0"></span>

<span id="page-6-1"></span>**FIGURE 3.** The scheme of Scorbot-ER.



**FIGURE 4.** Evolution of NN weights.

The matrices of the components in the dynamics equation have the following form:

$$
\mathcal{H}(\delta) = \begin{bmatrix} \mathcal{H}_{11} & 0 & 0 \\ 0 & \sigma_6 & l_2 \sigma_2 \cos(\delta_3 - \delta_2) \\ 0 & l_2 \sigma_2 \cos(\delta_3 - \delta_2) & \sigma_7 \end{bmatrix},\tag{53}
$$

$$
\mathcal{C}(\delta, \dot{\delta}) = \begin{bmatrix} \Delta_1 \dot{\delta}_2 + \Delta_2 \dot{\delta}_3 & \Delta_1 \dot{\delta}_1 & \Delta_2 \dot{\delta}_1 \\ -\Delta_1 \dot{\delta}_1 & 0 & -\Delta_3 \dot{\delta}_3 \\ -\Delta_2 \dot{\delta}_1 & \Delta_3 \dot{\delta}_2 & 0 \end{bmatrix},\tag{54}
$$

$$
\mathcal{F}(\dot{\delta}) = \begin{bmatrix} \sigma_8 \dot{\delta}_1 + \sigma_{11} \operatorname{sgn}(\dot{\delta}_1) \\ \sigma_9 \dot{\delta}_2 + \sigma_{12} \operatorname{sgn}(\dot{\delta}_2) \\ \sigma_{10} \dot{\delta}_3 + \sigma_{13} \operatorname{sgn}(\dot{\delta}_3) \end{bmatrix},
$$
\n(55)

$$
\mathcal{G}(\delta) = \begin{bmatrix} 0 & \sigma_1 g \cos \delta_2 & \sigma_2 g \cos \delta_3 \end{bmatrix}^T, \tag{56}
$$

where  $\mathcal{H}_{11} = \mathcal{H}_{111} + \mathcal{H}_{112} + \mathcal{H}_{113} + \mathcal{H}_{114}$ ,  $\mathcal{H}_{111} =$  $2\sigma_1 l_e \cos \delta_2$ ,  $\mathcal{H}_{112} = 2\sigma_2 (l_e + l_2 \cos \delta_2) \cos \delta_3$ ,  $\mathcal{H}_{113} =$ 0.5 $\sigma_3 \cos(2\delta_2)$ ,  $\mathcal{H}_{114}$  = 0.5 $\sigma_4 \cos(2\delta_3) + \sigma_5$ ,  $\Delta_1$  =  $-(\sigma_1 l_e \sin \delta_2 + \sigma_2 l_2 \sin \delta_2 \cos \delta_3 + 0.5 \sigma_3 \sin(2\delta_2))$ ,  $\Delta_2$  =  $-(\sigma_2(l_e + l_2 \cos \delta_2) \sin \delta_3 + 0.5\sigma_4 \sin(2\delta_3))$ ,  $\Delta_3$  =  $l_2\sigma_2 \sin(\delta_3 - \delta_2)$ ,  $\sigma_1 = \sigma_4 = \sigma_7 = 0.006$ ,  $\sigma_2 = 0.002$ ,  $\sigma_3 = \sigma_5 = \sigma_6 = 0.011, \sigma_8 = \sigma_9 = \sigma_{10} = 0.52,$  $\sigma_{11} = 0.019, \sigma_{12} = \sigma_{13} = 0.018, \delta(0) = [0.01, 0.01, 0.01]^T,$  $\dot{\delta}(0) = [0, 0, 0]^T$ ,  $\chi_1 = -0.8$ ,  $\chi_2 = 0.9$ . The desired

<span id="page-6-2"></span>

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**FIGURE 5.** Trajectory tracking of CHOC and RAC in joint space: (a) Position  $\delta_1$ ; (b) Position  $\delta_2$ ; (c) Position  $\delta_3$ .

<span id="page-6-3"></span>

**FIGURE 6.** Angular velocity of joints of CHOC and RAC: (a)  $\psi_1$ ; (b)  $\psi_2$ ; (c)  $\psi_3$ .

trajectory of  $P_d$  is described as follows:

$$
P_d = \begin{bmatrix} x_{P_d} \\ y_{P_d} \\ z_{P_d} \end{bmatrix} = \begin{bmatrix} 0.36 + 0.05 \sin(0.08\pi t) \\ 0.05 \cos(0.08\pi t) \\ 0.4 \end{bmatrix}.
$$
 (57)

# A. SIMULATION

The parameters of the CHOC are selected as  $K_1 = K_2$  $K_3 = K_4 = diag[10, 10, 10]$ , the initial weights are zeros,  $\Theta = I \in \mathbb{R}^{6 \times 6}, R = I \in \mathbb{R}^{6 \times 6}, \alpha = 100, \nu_2 = 0.15,$  $\nu_1 = -0.1, \rho = 0.5, \chi_2 = 0.9, \chi_1 = -0.8$ , the activation function is

$$
\zeta(e) = [e_{\delta 1}^2, e_{\delta 1}e_{\delta 2}, e_{\delta 1}e_{\delta 3}, e_{\delta 1}e_{\psi 1}, e_{\delta 1}e_{\psi 2}, e_{\delta 1}e_{\psi 3}, e_{\delta 2}^2,
$$

<span id="page-7-0"></span>

**FIGURE 7.** Position tracking errors of CHOC and RAC in joint space: (a) CHOC; (b) RAC.

<span id="page-7-1"></span>

**FIGURE 8.** Angular velocity tracking errors of CHOC and RAC in joint space: (a) CHOC; (b) RAC.

$$
e_{\delta 2}e_{\delta 3}, e_{\delta 2}e_{\psi 1}, e_{\delta 2}e_{\psi 2}, e_{\delta 2}e_{\psi 3}, e_{\delta 3}^2, e_{\delta 3}e_{\psi 1}, e_{\delta 3}e_{\psi 2}, e_{\delta 3}e_{\psi 3}, e_{\psi 1}^2, e_{\psi 1}e_{\psi 2}, e_{\psi 1}e_{\psi 3}, e_{\psi 2}^2, e_{\psi 2}e_{\psi 3}, e_{\psi 3}^2]^T,
$$

the sampling period  $T = 0.01s$ . The control law of RAC is given as  $\tau = \hat{f}(z) + K_6 s - v$ , where  $\hat{f}(z)$  is RBFNN and is an approximation of  $f(z)$ ,  $f(z) = H(\ddot{\delta}_{ref} + K_5\dot{e}_\delta)$  +  $\mathcal{C}(\dot{\delta}_{ref} + K_5 e_{\delta}) + \mathcal{G} + \mathcal{F}, K_5 = K_6 = diag[10, 10, 10],$  $s = \dot{e}_\delta + K_5 e_\delta$ ,  $v = -K_7$  sgn(*s*),  $K_7 = 0.2$ . We change the model parameters at the 60th second for both control laws, i.e., the values  $\sigma_1, \sigma_2, \ldots, \sigma_{12}$ , and  $\sigma_{13}$  are changed to double

<span id="page-7-2"></span>

**FIGURE 9.** Trajectory tracking of CHOC and RAC in the worksapce: (a) Position  $x_P$ ; (b) Position  $y_P$ .

<span id="page-7-3"></span>

**FIGURE 10.** Tracking errors of CHOC and RAC in the workspace: (a) CHOC; (b) RAC.

the initial values. Fig. [4](#page-6-1) presents the convolution of the NN weights, where the weights converge after 10s. At the time of load change (60th second), the weights change to adapt to the load change and then continue to converge. The results of comparing the trajectories of the positions of CHOC and RAC in joint space are illustrated in Fig. [5.](#page-6-2) Fig. [6](#page-6-3) shows CHOC and RAC's angular velocity tracking trajectory. The position and angular velocity tracking errors of the controllers in joint space are illustrated in Figs. [7,](#page-7-0) [8.](#page-7-1) The comparison

<span id="page-8-0"></span>

FIGURE 11. Position of point P of CHOC and RAC: (a) During learning; (b) After convergence.

<span id="page-8-1"></span>

results show that CHOC's ability to track trajectories in the joint space is better than RAC's. CHOC's position tracking error does not exceed 12.10<sup>-4</sup> rad. When the load changes suddenly, the position tracking error increases but does not exceed 4.10<sup>-3</sup> rad and then returns to the value 12.10<sup>-4</sup> rad (see Fig. [7\)](#page-7-0). In addition, Fig. [8](#page-7-1) shows that CHOC's angular velocity tracking error has a smaller value than RAC.

The position trajectories in the controllers' workspace are shown in Fig. [9.](#page-7-2) The tracking errors in the controllers' workspace are shown in Fig. [10,](#page-7-3) which indicates that the tracking errors of CHOC are smaller than RAC. The tracking error of CHOC in the workspace is less than 0.6 mm for the *X*-position, 0.45 mm for the *Y* -position, and 0.6 mm for the *Z*-position (see Fig. [10\)](#page-7-3). The position of point *P* of CHOC and RAC are illustrated in Fig. [11,](#page-8-0) where Fig. [11a](#page-8-0) shows the results during the learning process, and Fig. [11b](#page-8-0) shows the results after convergence. The control inputs of CHOC are presented in Figs. [12](#page-8-1) and [13a,](#page-8-2) where Fig. [12](#page-8-1) shows the optimal feedback control inputs and Fig. [13a](#page-8-2) illustrates the actual control torque inputs. The control torque results of RAC are presented in Fig. [13b.](#page-8-2) Fig. [12](#page-8-1) shows that the optimal control inputs are within the allowable asymmetry limit and

<span id="page-8-2"></span>

<span id="page-8-3"></span>**FIGURE 13.** Control torques of CHOC and RAC: (a) CHOC; (b) RAC.



**FIGURE 14.** Experimental results of the trajectory tracking of CHOC and RAC in the workspace: (a) Position  $x_p$ ; (b) Position  $y_p$ .

change value when the minimum and maximum thresholds are reached, which leads to the actual control torques not violating the asymmetry constraint (Fig. [13a\)](#page-8-2). Since RAC does not handle the influence of input constraints, the control torques fluctuate enormously in the early stages (Fig. [13b\)](#page-8-2). After the algorithms converge, the actual control torques of CHOC reach the optimal value while the torques of RAC do not (Fig. [13\)](#page-8-2). The simulation results show that CHOC has better control performance than RAC.

#### B. EXPERIMENT

In this section, we present experimental results on Scorbot-ER. Scorbot's hardware is set up as follows (see Fig. [2\)](#page-5-8): The drive motor of the robot joints is a DC motor with a built-in gearbox with a gear ratio of 127.7:1. In addition,

<span id="page-9-12"></span>

<span id="page-9-13"></span>**FIGURE 15.** Experimental results of the tracking error of CHOC and RAC in the workspace: (a) CHOC; (b) RAC.



**FIGURE 16.** Experimental results: Position of point P of CHOC and RAC after convergence.

the motors are equipped with optical encoders to measure the angular velocity and angular position of the joints. The control algorithms are implemented on the STM32F407 microcontroller. To interface the STM32F407 with the motors, we use the L298 driver circuit.

For the CHOC and RAC algorithms, we use convergence weights from simulation to make real-time learning faster. The position tracking trajectories in the workspace of CHOC and RAC are presented in Fig. [14.](#page-8-3) The position tracking errors of CHOC in the workspace are shown in Fig. [15a,](#page-9-12) indicating that the tracking errors are significant in the early stages and small as the algorithm converges. Fig. [15b](#page-9-12) shows tracking errors of RAC in the workspace. Comparing the tracking errors of CHOC and RAC, we quickly see that the tracking errors of CHOC are smaller than RAC. Fig. [16](#page-9-13) presents the position of point *P* of CHOC and RAC after convergence, showing that CHOC's tracking performance is better than RAC's. Through experimental results, the control effectiveness of CHOC has been verified.

A video illustrating experimental results can be found at the following URL: https://www.youtube.com/watch?v=A-S4Chi5798.

# <span id="page-9-11"></span>**V. CONCLUSION**

The  $H_{\infty}$  optimal control algorithm has been proposed for robot manipulators with asymmetric input constraints and disturbances. The feedforward control is proposed to transform the  $H_{\infty}$  optimal control problem for robot manipulators into the  $H_{\infty}$  optimal control problem for the affine systems. The constrained optimal control and disturbance compensation laws have been built based on ADP and the game theory. The proposed control algorithm has been analyzed to show that the closed-loop dynamics is UUB stable and the control parameters converge to the near-optimal values. The performance of the proposed algorithm has been verified through comparative simulation and experimental results. CHOC can be developed to be applied to multi-agent systems, such as decentralized control of robot manipulators or some multiple robot manipulator operations. Future work will focus on decentralized optimal control of robot manipulators and distributed optimal control for multi-robot systems.

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# **CONFLICT OF INTEREST**

The authors declare no potential conflict of interests.

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