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A Robust Method for Block Adjustment of UAV SAR Images

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ABSTRACT To reduce the number of ground control points required for block adjustment of unmanned aerial vehicle (UAV) synthetic aperture radar (SAR) images, a robust block adjustment method is pro-posed. The proposed method aims to solve the problem of unstable adjustment solutions caused by the jittering of UAV SAR platforms. This method is based on the range Doppler model. By analyzing the correlation between the antenna phase center, velocity, and Doppler centroid of the seven orientation parameters, this method performs a block adjustment of UAV SAR images with three orientation parameters: the antenna phase center initial imaging moment, Doppler centroid, and proximity delay. UAV SAR images obtained from an area of 2×3 km in Dengfeng were used to conduct adjustment experiments. The results obtained using different orientation parameter set-tings verify the robustness and effectiveness of the proposed method. The adjustment results under various control-point layout plans indicate that a uniform layout plan can achieve an adjustment accuracy better than 1 m with a small number of control points.

INDEX TERMS Unmanned aerial vehicle (UAV), SAR images, range Doppler model, three-parameter, block adjustment.

I. INTRODUCTION

Unmanned aerial vehicle (UAV) synthetic aperture radar (SAR) is a new surveying and mapping system, which includes a combination of SAR sensors on a UAV platform. The system exploits the features of targeted data acquisition, short cycle time, low cost, safety and reliability, mobility, and flexibility enabled by the UAV platform, as well as the active all-day microwave remote sensing and all-weather acquisition of high-resolution images offered by the SAR sensor [1], [2]. UAV SAR has been widely used in topographic surveying [3], emergency surveying [4], landslide detection [5], forest monitoring [6], and other fields. To conduct topographic surveys using a UAV SAR system and generate high-resolution images, the orientation parameters

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of each SAR image must be accurately acquired using block adjustment with the support of ground control points [7]. However, due to the low flight altitude and small imaging width of the UAV SAR, multiple images of multiple strips are required to cover the entire survey area. Thus, a large number of control points are required for block adjustment to obtain the orientation parameters. In addition, the vibration of the UAV SAR leads to a lack of robustness in the block-adjustment solution. To reduce the number of control points required for adjust-ment, enhance the robustness of the adjustment solution, and to improve the effectiveness and performance of surveying and producing UAV SAR images, obtaining a robust solution for block adjustment is an urgent requirement.

The first step in achieving block adjustment of the SAR images is to select a suitable imaging model. Common imaging models include range-Doppler (RD) [8], range

co-planarity [9], equivalent collinearity [10], and rational function models (RFM) [11]; specifically, the RD model is also known as the Leberl model under the zero Doppler effect [12]. Considerable research has been conducted on block-adjustment methods based on the aforementioned imaging models. Xiong et al. [7] solved seven orientation parameters of the RD model in a UAV SAR image area using a route-constraint model. Yue et al. [13] explored a block adjustment algorithm for UAV SAR images based on the Leberl and RD models under sparse or no control conditions. Zhang et al. [14] derived an unbiased estimation of the rational polynomial coefficient (RPC) for SAR images by analyzing the geometric effect of SAR images, and demonstrated that the RPC model is applicable to satellite SAR images using a large number of strict sensor-based experiments. Jiao et al. [15] proposed an error source-based method to improve the geometric positioning accu-racy of Gaofen-3 (GF-3) SAR satellites. Zhang et al. [16] used the RD model to construct a position and orientation system (POS) to assist the aerotriangulation model for UAV SAR images. In addition, Jiao et al. [17] proposed a new slant-range error update model and an error-source-based weighting strategy that can effectively improve the three-dimensional (3D) geolocation accuracy of the multi-observation dataset of GF-3 satellites. Yang et al. [18] matched SAR images with data from imaging models and information from an ex-ternal digital elevation model (DEM) to perform block adjustment. Toutin and Omari [19] created a new hybrid model by combining RFM and physical models to process RADAR-SAT-2 data without ground control points. Huang and He [20] combined model construction, parameter solutions, and RFM model refinement for block adjustment. Zhang et al. [21] generated an RPC model by compensating GF-3 satellite images with geometric calibration and then used a digital elevation model (DEM) to assist block adjustment. Jiao et al. [22] used the OS-SIFT algorithm [23] to match the optical and SAR images and performed joint adjustment with the RFM model.

Current studies on block adjustment mainly focus on satellite-and large air-craft-based SAR images. Except for the study conducted by Xiong et al. [7], in which UAV SAR images were incorporated into the experiments, there have been few studies on UAV SAR images. Compared to satellite-and large aircraft-based SAR systems, the UAV SAR system has a smaller imaging width and poorer platform stability. In response to these features, this study presents the design of a block adjustment method for UAV SAR images that solves three orientation parameters: the initial imaging moment of the antenna phase center, Doppler centroid, and proximity delay, which effectively improves the robustness and accuracy of the adjustment solution.

The contributions of the work are as follows.

- 1. A robust block adjustment algorithm based on RD model for UAV SAR image is proposed.
- 2. The influence of the number and distribution of different control points on the adjustment results is analyzed.

The remainder of this paper is organized as follows. The basic RD model and our proposed selection of orientation parameters are introduced in Sec. II. In Sec. III, the experimental results are presented to verify the robustness and accuracy of our proposed block adjustment method using SAR images covering the Dengfeng area. Finally, the conclusions and discussion are presented in Sec. IV.

II. ROBUST APPROACH TO BLOCK ADJUSTMENT OF UAV SAR IMAGES

A. RANGE DOPPLER MODEL

The RD model is a direct description of the SAR imaging principle [24] and has clear geometric and physical meanings [25]. The model uses the distance and Doppler condition equations to describe the correspondence between the ground point and corresponding image point at a given moment as follows:

$$R_{S}^{2} = (R_{0} + M_{s} \cdot x)^{2} = (X - X_{S})^{2} + (Y - Y_{S})^{2} + (Z - Z_{S})^{2}$$
$$V_{X} (X - X_{S}) + V_{Y} (Y - Y_{S}) + V_{Z} (Z - Z_{S}) = -\frac{\lambda R_{S}}{2} f_{dc}$$
(1)

where (X, Y, Z) represent the coordinates of the ground point in the object space coordinate system; (x, y) represent the pixel coordinates of the image point; x represents the range; y represents the azimuth; $[X_S Y_S Z_S]^T$ is the instantaneous position vector of the antenna phase center; M_s is the slant range sampling interval; R_0 is the proximity delay; f_{dc} is the Doppler shift parameter; $[V_X V_Y V_Z]^T$ is the instantaneous velocity vector of the time-of-flight; and λ is the radar wavelength. The antenna phase center and flight velocity were calculated as follows:

$$\begin{cases} t = t' \cdot y + t_0 \\ X_S = X_{S_0} + V_{X_0} \cdot t + a_{X_0} \cdot t^2 + \cdots \\ Y_S = Y_{S_0} + V_{Y_0} \cdot t + a_{Y_0} \cdot t^2 + \cdots \\ Z_S = Z_{S_0} + V_{Z_0} \cdot t + a_{Z_0} \cdot t^2 + \cdots \\ V_X = V_{X_0} + 2a_{X_0} \cdot t + \cdots \\ V_Y = V_{Y_0} + 2a_{Y_0} \cdot t + \cdots \\ V_Z = V_{Z_0} + 2a_{Z_0} \cdot t + \cdots \end{cases}$$
(2)

where t' is the temporal interval between adjacent azimuthal rows; typically, the ratio of the azimuthal distance and sampling interval M_a ; t_0 is the imaging moment of the firstrow images, $\begin{bmatrix} a_{X_0} & a_{Y_0} & a_{Z_0} \end{bmatrix}^T$ is the antenna phase center acceleration vector at the initial moment, $\begin{bmatrix} V_{X_0} & V_{Y_0} & V_{Z_0} \end{bmatrix}^T$ is the time-of-flight instantaneous velocity vector at the initial moment, and $\begin{bmatrix} X_{S_0} & Y_{S_0} & Z_{S_0} \end{bmatrix}^T$ is the antenna phase center instantaneous position vector at the initial moment.

B. SELECTION OF ORIENTATION PARAMETERS

Seven parameters, namely, the initial position of the antenna phase center $\begin{bmatrix} X_{S_0} & Y_{S_0} & Z_{S_0} \end{bmatrix}^T$, initial velocity $\begin{bmatrix} V_{X_0} & V_{Y_0} & V_{Z_0} \end{bmatrix}^T$, and Doppler centroid f_{dc} , are generally selected as the orientation parameters for block adjustment of

SAR images using the RD model [7]. The seven-parameter method can efficiently restore the motion state of the SAR system and is suitable for the block-adjustment solution of both satellite-based and airborne SAR images. However, UAV SAR images have a smaller width, and a large number of SAR images are required to cover the survey area. The need for additional parameters for the adjustment solution will require a large number of ground control points, resulting in slow convergence or even non-convergence of the adjustment solution. Given that the initial velocity and Doppler centroid have a certain degree of correlation and that the velocity data from the POS data are generally relatively accurate, a total of four parameters, namely, the initial velocity of the antenna phase center $\begin{bmatrix} X_{S_0} & Y_{S_0} & Z_{S_0} \end{bmatrix}^T$ and the Doppler centroid f_{dc} , can be selected as the orientation parameters [26]. In comparison with the seven-parameter method, the four-parameter method reduces the correlation between the orientation parameters and improves robustness and efficiency. However, the UAV SAR platforms has poor flight stability. Thus, during the adjustment solution, the errors in the initial position are inclined to compensate for each other in the two directions (i.e., range and vertical direction) of the vertical flight trajectory, which is not conducive to the rapid convergence of the adjustment solution.

To address the problems related to the seven- and fourparameter methods, considering the features of the UAV SAR platform, this study selected the three orientation parameters for the block-adjustment solution: the initial imaging moment of the antenna phase center t_0 , Doppler shift parameter f_{dc} , and proximity delay R_0 . The three-parameter method restricts the initial position error to the along-the-track direction (azimuth) and the slant range direction, which avoids errors from the compensation of the range and vertical direction and reduces the correlation between the orientation parameters. This is conducive to the fast convergence of the adjustment solution and can effectively improve the robustness of the block-adjustment solution of UAV SAR images.

C. ORIENTATION PARAMETER SOLVING

After determining the orientation parameters, (2) can be simplified to

$$\begin{cases}
t = t' \cdot y + t_0 \\
X_S = X_{S_0} + V_{X_0} \cdot t \\
Y_S = Y_{S_0} + V_{Y_0} \cdot t \\
Z_S = Z_{S_0} + V_{Z_0} \cdot t \\
V_X = V_{X_0} \\
V_Y = V_{Y_0} \\
V_Z = V_{Z_0}
\end{cases}$$
(3)

The RD model is nonlinear and must be linearized to solve the orientation parameters. According to (1):

$$F_1 = (X - X_S)^2 + (Y - Y_S)^2 + (Z - Z_S)^2 - (R_0 + M_s \cdot x)^2$$

The error equation after linearization is

$$\boldsymbol{V} = \boldsymbol{A} \cdot \boldsymbol{\Delta}_{\boldsymbol{P}} + \boldsymbol{B} \cdot \boldsymbol{\Delta}_{\boldsymbol{G}} - \boldsymbol{L} \quad \boldsymbol{P} \tag{5}$$

where V is the error vector, A is the coefficient matrix of the orientation parameter corrections, Δ_P is the orientation parameter correction vector, B is the coefficient matrix of the ground point coordinate corrections, Δ_G is the ground point coordinate correction vector, L is a constant term, and P is the weight matrix.

The observations in the error equation include the range coordinate x and azimuthal coordinate y of the image point, which are considered independent of each other, and the root mean square errors (RMSE) are m_x and m_y , respectively. The unit weight RMSE is taken to be μ , and the weight matrix **P** of the error equation, according to the weight inverse matrix propagation formula, is

Р

$$= \begin{bmatrix} \frac{\mu^2}{\left(\frac{\partial F_1}{\partial x}\right)^2 m_x^2 + \left(\frac{\partial F_1}{\partial y}\right)^2 m_y^2} & 0\\ 0 & \frac{\mu^2}{\left(\frac{\partial F_2}{\partial x}\right)^2 m_x^2 + \left(\frac{\partial F_2}{\partial y}\right)^2 m_y^2} \end{bmatrix}$$
(6)

For a ground control point j, which is located at the overlap of the m images, the error equation can be formulated according to the RD model, and the unknown in the equation is the parameter correction in the m SAR images. The error equation is expressed as follows:

$$\boldsymbol{V}_{GCPj} = \boldsymbol{A}_{GCPj} \cdot \boldsymbol{\Delta}_{GCPj} - \boldsymbol{L}_{GCPj} \quad \boldsymbol{P}_{GCPj}$$
(7)

where $V_{GCPj} = \begin{bmatrix} V_1 & V_2 & \dots & V_m \end{bmatrix}_{2m \times 1}^T$ is the residual vector, $\mathbf{\Delta}_{GCPj} = \begin{bmatrix} \mathbf{\Delta}_{P1} & \mathbf{\Delta}_{P2} & \dots & \mathbf{\Delta}_{Pm} \end{bmatrix}_{3m \times 1}^T$ is the orientation parameter correction vector, $\mathbf{L}_{GCPj} = \begin{bmatrix} \mathbf{L}_{1j} & \mathbf{L}_{2j} & \dots & \mathbf{L}_{mj} \end{bmatrix}_{2m \times 1}^T$ is the constant term vector, \mathbf{A}_{GCPj} is the matrix of the orientation parameter correction coefficients of order $2m \times 3m$, and \mathbf{P}_{GCPj} is the weight matrix of order $2m \times 2m$.

For a particular connection point k, which is located at the overlap of n images, the unknowns in the error equation contain the ground coordinate corrections of the connection point Δ_{Gk} , in addition to the RD model parameter corrections in the n SAR images. The error equation is

$$\boldsymbol{V}_{TPk} = \boldsymbol{A}_{TPk} \cdot \boldsymbol{\Delta}_{TPk} + \boldsymbol{B}_{TPk} \cdot \boldsymbol{\Delta}_{Gk} - \boldsymbol{L}_{TPk} \quad \boldsymbol{P}_{TPk} \quad (8)$$

where $V_{TPk} = \begin{bmatrix} V_1 & V_2 & \dots & V_n \end{bmatrix}_{2n \times 1}^T$ is the residual vector, $\mathbf{\Delta}_{TPk} = \begin{bmatrix} \mathbf{\Delta}_{P1} & \mathbf{\Delta}_{P2} & \dots & \mathbf{\Delta}_{Pn} \end{bmatrix}_{3n \times 1}^T$ is the orientation parameter correction vector, $\mathbf{\Delta}_{Gk}$ is the ground coordinate correction of the connection point, $\mathbf{L}_{TPk} = \begin{bmatrix} \mathbf{L}_{1k} & \mathbf{L}_{2k} & \dots & \mathbf{L}_{nk} \end{bmatrix}_{2n \times 1}^T$ is the constant-term vector, \mathbf{A}_{TPk} is the matrix of the orientation parameter correction coefficients of order $2n \times 3n$, \mathbf{B}_{TPk} is



FIGURE 1. Process of the block adjustment.

the matrix of the connection point correction coefficients of order $2n \times 3$, and P_{TPk} is the weight matrix of order $2n \times 2n$.

The error equations for the control and connection points are formulated according to (7) and (8), respectively. Meanwhile, the error equation for the UAV SAR block adjustment is

$$\boldsymbol{V} = \boldsymbol{C} \cdot \boldsymbol{\Delta}_O + \boldsymbol{D} \cdot \boldsymbol{\Delta}_G - \boldsymbol{L} \quad \boldsymbol{P} \tag{9}$$

where V is the residual vector, C is the matrix of the orientation parameter correction coefficients, Δ_O is the orientation parameter correction vector, D is the matrix of the ground coordinate correction coefficients, Δ_G is the ground coordinate correction vector, L is a constant term vector, and P is the weight matrix. We have

$$C = \begin{bmatrix} A_{GCP} \\ A_{TP} \end{bmatrix} \quad D = \begin{bmatrix} 0 \\ B_{TP} \end{bmatrix}$$
$$L = \begin{bmatrix} L_{GCP} \\ L_{TP} \end{bmatrix} \quad P = \begin{bmatrix} P_{GCP} & 0 \\ 0 & P_{TP} \end{bmatrix} \quad (10)$$

The corresponding normal equation is

$$\begin{bmatrix} \boldsymbol{C}^T \boldsymbol{P} \boldsymbol{C} & \boldsymbol{C}^T \boldsymbol{P} \boldsymbol{D} \\ \boldsymbol{D}^T \boldsymbol{P} \boldsymbol{C} & \boldsymbol{D}^T \boldsymbol{P} \boldsymbol{D} \end{bmatrix} \begin{bmatrix} \boldsymbol{\Delta}_O \\ \boldsymbol{\Delta}_G \end{bmatrix} = \begin{bmatrix} \boldsymbol{C}^T \boldsymbol{P} \boldsymbol{L} \\ \boldsymbol{D}^T \boldsymbol{P} \boldsymbol{L} \end{bmatrix}$$
(11)

An iterative solution of the normal equations yields the RD model orientation parameters and the ground coordinates of the connection points for each SAR image.

The process for solving the SAR image block adjustment is presented in Fig. 1.

The following is the process for solving the SAR image block adjustment:

- 1. The ground and image coordinates of the control point, image coordinates of the connection point, and SAR system parameters are obtained. The SAR system parameters include the slope range sampling interval M_s , azimuth sampling interval M_a , radar wavelength λ , instantaneous position vector of the antenna phase center at the initial moment $\begin{bmatrix} X_{S_0} & Y_{S_0} & Z_{S_0} \end{bmatrix}^T$, and instantaneous velocity vector at the initial moment $\begin{bmatrix} V_{X_0} & V_{Y_0} & V_{Z_0} \end{bmatrix}^T$.
- 2. The ground coordinates of the connection point and initial values of the orientation parameters were determined. The orientation parameters included the initial imaging moment, t_0 , Doppler shift parameter f_{dc} , and proximity delay R_0 .
- 3. The error and normal equations are formulated to solve the corrections of the SAR image orientation parameter t_0, f_{dc}, R_0 , and the ground coordinates of the connection point (X, Y, Z). Subsequently, the initial values are corrected using the solved corrections.
- 4. Steps 2 and 3 are repeated until all corrections satisfy the given conditions for convergence or until the preset upper limit of the number of iterations is reached. Finally, the orientation parameters and ground coordinates of the connection points of all SAR images are obtained.

III. EXPERIMENTAL RESULTS AND ANALYSIS

A. EXPERIMENTAL DATA

To verify the effectiveness of the proposed method, we selected the UAV SAR images of the Dengfeng area in Henan Province acquired in March 2022 to conduct the block adjustment experiment. The airborne platform used for data acquisition was the Zongheng CW-30 medium-sized vertical takeoff and landing (VTOL) fixed-wing UAV loaded with MicroSAR sensors developed by the Institute of Electronics, Chinese Academy of Sciences. The sensor features a right-side averted vision as the imaging mode, and the operating frequency band is Ku-band. The experimental area is approximately 2 km from east to west and 3 km from north to south. It mainly consists of plains and hills, including residential areas, water systems, roads, vegetation, and other surface features. The highest elevation in the area is approximately 562 m, and it is located in the southwestern hilly region. The lowest elevation is approximately 424 m and is located southeast of the reservoir. The scope of the survey and topographic information are presented in Fig. 2.

The relative altitude of the flight was approximately 600 m, and the direction was from west to east, with a total of 13 strips. Specifically, 2–3 images were acquired for each strip, with an overlap of approximately 60% between the images of the adjacent strips (Fig. 3).

A total of 28 SAR images were acquired in the experimental area, and Table 1 lists the basic parameters.

Two example images of adjacent strips are shown in Fig. 4. Here, the horizontal to the right is the azimuthal direction, and the vertical down is the range direction.



FIGURE 2. Experimental area coverage and terrain information.



FIGURE 3. UAV SAR stereo imaging.

TABLE 1. Basic parameters of SAR images.

Parameters	Numerical value
Frequency band	Ku
Radar wavelength (mm)	20.534
Number of range direction pixels	3328
Number of azimuthal direction pixels	16384
Range direction sampling interval (m)	0.0539
Azimuthal direction sampling interval (m)	0.1249

Block adjustment of UAV SAR images requires a certain number of ground control points to be acquired, and a certain number of checkpoints are required to verify the adjustment accuracy. In this study, 9 control points were obtained by field measurements using corner reflectors, and 52 control points were manually collected using high-precision optical digital orthophoto maps (DOM) and digital surface models (DSM) of the area. The corner reflectors are mainly located in flat and open areas, and the manually selected points are mainly obvious markers (e.g., house corners, road intersections, and street light bases), as control points cannot be selected in dense vegetation and water areas. The coordinates of the corner



FIGURE 4. Examples of UAV SAR images.



FIGURE 5. Partial control points scenario. (a) Field-deployed corner reflectors. (b) Manually acquired control points in the optical images. (c) Corner reflectors in the SAR images. (d) Manually acquired control points in SAR images.

reflector were obtained using real-time kinematics (RTK) with a point accuracy of 2–3 cm. For the points selected using high-precision optical DOM and DSM, the point accuracy was approximately 4.7 cm. The accuracy of both types of control points was in accordance with relevant standards. Some of the control points are presented in Fig. 5.

The distribution of the control points in the experimental area is shown in Fig. 6. There were more control points in the north, central west, and southwest, which mainly comprises open land or artificial structures and were suitable to be



FIGURE 6. Distribution of control points.

selected as control points. The central east is the reservoir area, and the southeast is an area covered with vegetation, where it is difficult to obtain control points.

The connection points were matched and obtained using the SAR-SIFT algorithm [27]. After decimating, 4652 pairs of connection points were obtained from 28 SAR images. An average of 140 pairs of connection points were extracted for image pairs with 60% overlap, and 10 pairs of connection points were extracted for image pairs with 20% overlap.

B. BLOCK ADJUSTMENT RESULTS

A subset of the control points was selected for adjustment, and the remaining points were used as checkpoints to evaluate the adjustment results. In order to compare the results of the adjustment experiment better, first use the RD model parameters obtained by the UAV and the sensors on the SAR to calculate all the control points and count the residuals. It is calculated that without adjustment, the RMSE of all control points in the three directions are 2.868m, 1.045m and 3.318m respectively, totaling 4.509m. Four control-point layout plans were set up for the experiment (Fig. 7). Fig. 7 (a) shows the unilateral layout, where the selected control points are concentrated in the north of the experimental area and the checkpoints are in the south. This plan is consistent with the direction of the averted vision of the radar from north to south and can simulate the situation when the UAV cannot span the target area. Fig. 7 (b) shows the external layout, where the selected control points are within 200 m from the boundary of the experimental area and point C051 in the southeast corner. The check points are located inside the experimental area. This plan is used to simulate the situation



FIGURE 7. Control point layout plans. (a) Unilateral layout. (b) External layout. (c) Flat area layout. (d) Uniform layout.

where the inner area is inaccessible. Fig. 7 (c) shows the flat area layout, where the control points are located in the flatter area, and the check points are located in the hilly area with a larger slope. Generally, control points are easier to set up in areas with gentle slopes. This plan was used to simulate the case of inaccessible areas with an undulating terrain. Fig. 7 (d) shows the uniform layout, where the control points are located at the edges, corners, and center of the experimental area. The points were relatively evenly distributed and were few in number. This plan was used to simulate a situation where a small number of control points were used to conduct large-scale topographic surveys.

We conducted separate adjustment calculations using the proposed adjustment method according to the four layout plans. We compared the results with those obtained from the seven- and four-parameter methods. The evaluation indicators of the experiments included the RMSE of the control points and checkpoints, adjustment time, and number of iterations. The RMSE of the control and check points were counted in the adjustment experiment for accuracy assessment. Here, the control point RMSE mainly represents the self-consistency of the algorithm, and the check point RMSE mainly represents the accuracy of the algorithm. In this experiment, the RMSE was divided into the X, Y, and Z directions. Because the flight direction was from west to east, the X, Y, and Z directions corresponded to the range, azimuth,

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tions 5000 5000 6000 External Control X 0.358 0.369 0.329 External RMSE Z 0.469 0.431 0.797 layout (m) Total 0.661 0.677 1.881 Check X 6.645 3.831 2.980 Control points Y 2.649 0.895 1.430 points: 23 RMSE Z 7.375 4.297 2.792 Check (m) Total 10.274 5.826 4.327 points: 38 Time (s) 696.3 683.7 90.2 Number of itera- tions 3000 3000 475 Flat area RMSE Z 1.005 0.457 0.757 layout (m) Total 1.466 0.791 1.331 Check X 8.777 3.002 1.414 Control points Y 1.491 0.715 1.132	pointor o r	Number of itera-		3000	3000	690
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$		tions		5000	5000	020
$\begin{array}{c c c c c c c c c c c c c c c c c c c $		Control	Х	0.358	0.369	0.329
External layout RMSE (m) Z 0.469 0.431 0.797 layout (m) Total 0.661 0.677 1.881 Check X 6.645 3.831 2.980 Control points: 23 RMSE Z 7.375 4.297 2.792 Check (m) Total 10.274 5.826 4.337 points: 38 Time (s) 696.3 683.7 90.2 Number of itera- tions 3000 3000 475 Points: 38 Time (s) 0.916 0.366 0.458 points< Y		points	Y	0.299	0.370	1.672
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	External	RMSE	_Z	0.469	0.431	0.797
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	layout	(m)	Total	0.661	0.677	1.881
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	~ .	Check	Х	6.645	3.831	2.980
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	Control	points	Y	2.649	0.895	1.430
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	points: 23	RMSE	_Z .	7.375	4.297	2.792
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	Check	(m)	Total	10.274	5.826	4.327
Number of itera- tions 3000 3000 475 Control X 0.916 0.366 0.458 points Y 0.548 0.532 0.994 Flat area RMSE Z 1.005 0.457 0.757 layout (m) Total 1.466 0.791 1.331 Check X 8.777 3.002 1.414 Control points Y 1.491 0.715 1.132 point: 38 RMSE Z 10.029 3.253 1.581 Check (m) Total 13.410 4.616 2.404 points: 23 Time (s) 687.6 658.5 100.8 Number of itera- tions 3000 3000 533 Uniform RMSE Z 0.399 0.411 0.585 points Y 0.225 0.519 1.767 Uniform RMSE Z 0.399 0.431 1.188 layout	points: 38	Time	(s)	696.3	683.7	90.2
$\begin{array}{c c c c c c c c c c c c c c c c c c c $		Number of itera- tions		3000	3000	475
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$		Control	Х	0.916	0.366	0.458
$\begin{array}{c c c c c c c c c c c c c c c c c c c $		points	Y	0.548	0.532	0.994
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	Flat area	RMSE	Z	1.005	0.457	0.757
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	layout	(m)	Total	1.466	0.791	1.331
$\begin{array}{c c c c c c c c c c c c c c c c c c c $		Check	Х	8.777	3.002	1.414
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	Control	points	Y	1.491	0.715	1.132
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	point: 38	RMSE	Z	10.029	3.253	1.581
points: 23 Time (s) 687.6 658.5 100.8 Number of itera- tions 3000 3000 533 Control X 0.399 0.411 0.585 points Y 0.225 0.519 1.767 Uniform RMSE Z 0.399 0.431 1.188 layout (m) Total 0.607 0.790 2.208 Control points Y 1.528 0.501 0.882 points: 13 RMSE Z 8.954 3.609 0.725 Check (m) Total 12.392 5.121 1.329 points: 48 Time (s) 695.5 658.484 85.312 Number of itera- tione 3000 3000 446	Check	(m)	Total	13.410	4.616	2.404
Number of itera- tions 3000 3000 533 Control X 0.399 0.411 0.585 points Y 0.225 0.519 1.767 Uniform RMSE Z 0.399 0.431 1.188 layout (m) Total 0.607 0.790 2.208 Control points Y 1.528 0.501 0.882 points: 13 RMSE Z 8.954 3.609 0.725 Check (m) Total 12.392 5.121 1.329 points: 13 RMSE Z 8.954 3.609 0.725 Check (m) Total 12.392 5.121 1.329 points: 48 Time (s) 695.5 658.484 85.312 Number of itera- 3000 3000 446	points: 23 Ti		(s)	687.6	658.5	100.8
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$		Number of itera- tions		3000	3000	533
points Y 0.225 0.519 1.767 Uniform RMSE Z 0.399 0.431 1.188 layout (m) Total 0.607 0.790 2.208 Check X 8.430 3.599 0.680 Control points Y 1.528 0.501 0.882 points: 13 RMSE Z 8.954 3.609 0.725 Check (m) Total 12.392 5.121 1.329 points: 48 Time (s) 695.5 658.484 85.312 Number of itera- 3000 3000 446		Control	Х	0.399	0.411	0.585
Uniform RMSE Z 0.399 0.431 1.188 layout (m) Total 0.607 0.790 2.208 Check X 8.430 3.599 0.680 Control points Y 1.528 0.501 0.882 points: 13 RMSE Z 8.954 3.609 0.725 Check (m) Total 12.392 5.121 1.329 points: 48 Time (s) 695.5 658.484 85.312 Number of itera- 3000 3000 446		points	Y	0.225	0.519	1.767
layout (m) Total 0.607 0.790 2.208 Check X 8.430 3.599 0.680 Control points Y 1.528 0.501 0.882 points: 13 RMSE Z 8.954 3.609 0.725 Check (m) Total 12.392 5.121 1.329 points: 48 Time (s) 695.5 658.484 85.312 Number of itera- 3000 3000 446	Uniform	RMSE	Ζ	0.399	0.431	1.188
Check X 8.430 3.599 0.680 Control points Y 1.528 0.501 0.882 points: 13 RMSE Z 8.954 3.609 0.725 Check (m) Total 12.392 5.121 1.329 points: 48 Time (s) 695.5 658.484 85.312 Number of itera- 3000 3000 446	layout	(m)	Total	0.607	0.790	2.208
Control points Y 1.528 0.501 0.882 points: 13 RMSE Z 8.954 3.609 0.725 Check (m) Total 12.392 5.121 1.329 points: 48 Time (s) 695.5 658.484 85.312 Number of itera- 3000 3000 446		Check	Х	8.430	3.599	0.680
points: 13 RMSE Z 8.954 3.609 0.725 Check (m) Total 12.392 5.121 1.329 points: 48 Time (s) 695.5 658.484 85.312 Number of itera- tions 3000 3000 446	Control	points	Y	1.528	0.501	0.882
Check (m) Total 12.392 5.121 1.329 points: 48 Time (s) 695.5 658.484 85.312 Number of iterations 3000 3000 446	points: 13	RMSE	Z	8.954	3.609	0.725
points: 48 Time (s) 695.5 658.484 85.312 Number of itera- tions 3000 3000 446	Check	(m)	Total	12.392	5.121	1.329
Number of itera- tions 3000 3000 446	points: 48	Time (s)		695.5	658.484	85.312
tions		Number of itera- tions		3000	3000	446

TABLE 2. Results of adjustment experiments.

and elevation directions, respectively. Adjustment time is the running time of the adjustment calculation program written in C++ language. The configuration of the laptop used for the experiment was as follows: the operating system was Windows 10 21H2 x64, CPU was Intel i9-9880H, and memory size was 64 GB. To consider the account efficiency and accuracy in actual applications, the upper limit of the number of iterations was set to 3000 from experience. The results of the adjustment experiments are listed in Table 2.

The following are the analysis results of the data in Tab. 2:

1. Comparison of the adjustment accuracies of the three adjustment methods. In terms of checkpoints, the proposed method always obtains the optimal value in the X- and Z-directions and in the overall RMSE, achieving the highest accuracy. The four-parameter method had

a slightly higher accuracy in the Y-direction, but the overall adjustment accuracy was not as good as that of the proposed method. The seven-parameter method had the lowest overall adjustment accuracy because a certain degree of correlation exists between the orientation parameters selected by the seven- and four-parameter methods. In addition, the initial errors are inclined to compensate for each other in the two directions of the vertical track, that is, the range and vertical directions. This results in an increase in the errors in the X- and Z-directions for these two methods, affecting the accuracy of the adjustment and being not conducive to the convergence of the iterations. The correlation between the orientation parameters selected using the proposed method was weak, and the errors were more evenly distributed in the three directions. Although the accuracy in the Y-direction was slightly lower, the overall adjustment accuracy should be improved. In terms of control points, the seven- and four-parameter methods generally perform better in terms of RMSE than the proposed method, because a larger number of orientation parameters in the adjustment process can better match the control points. However, if we simultaneously refer to the RMSE of the checkpoints, we can identify an "overfitting" phenomenon where it is difficult to match the adjustment results with the checkpoints. Comparing the unadjusted results, it can be seen that the accuracy of the seven-parameter and fourparameter adjustment is worse in some cases. This is because the "overfitting" phenomenon will occur when the number and distribution of control points are not good. Reduced precision for checkpoints not participating in adjustments. However, the proposed method can always achieve higher totaling accuracy. Therefore, the proposed method is better than the seven- and four-parameter methods. Consequently, the proposed method achieved the highest adjustment accuracy.

2. The robustness and efficiency of the three adjustment methods were compared. The proposed method always has the shortest adjustment time and the lowest number of iterations. The four-parameter method requires a longer time, the seven-parameter method required the longest time and both reached the upper limit of the set number of iterations in all the cases. Reaching the set upper limit implies that the convergence condition is not strictly reached in the adjustment process, and the result is a forced output after a certain number of iterations, indicating that the algorithm is less robust. The reason is that the seven- and four-parameter methods choose a large number of orientation parameters with a certain degree of correlation, which is not conducive to the convergence of iterations, affecting the robustness and efficiency of the adjustment. The orientation parameters chosen by the proposed method have a weak correlation and are smaller in number, which is conducive to rapid convergence of the adjustment

iterations. Therefore, the proposed method achieved the highest robustness and adjustment efficiency.

3. In the experiments with different control-point layout plans, the results of the three adjustment methods showed the same pattern. The unilateral layout plan makes full use of the advantage of the averted vision of the SAR; however, the checkpoint accuracy is poorer because the control points are concentrated in the north, making it difficult to effectively control the south. The external layout plan simulates another inaccessible situation for which it is difficult to achieve effective control over a large area at the center. Moreover, because more control points were concentrated in the southeast, which significantly reduced the control in other directions, the overall accuracy was worse than that of the unilateral layout plan. The flat area layout plan used most control points, but the accuracy was unsatisfactory. This is because the control points were located in areas with a gentle slope, whereas the check points were all located in hilly areas with large undulations. Accurate terrain restoration is difficult to achieve using only the connection points in the absence of control points. Therefore, a certain number of control points must be deployed in these areas to obtain high-accuracy adjustment results in hilly areas with large undulations. The uniform layout plan has the least number of control points but always has the best accuracy and highest efficiency. This is because the control points in this plan cover most of the survey area and include different terrains, such as flat land and hills, which can better restore the entire survey area. Thus, when the control points are distributed more evenly in the survey area, only a small number of control points are required to achieve a higher adjustment accuracy. A comparison of the different layout plans showed that the highest adjustment accuracy could be obtained with relatively uniformly distributed control points in the survey area. In addition, the proposed method could still obtain a relatively high adjustment accuracy under three difficult conditions, further confirming the effectiveness of the proposed method.

C. NUMBER OF CONTROL POINTS EXPERIMENT

The results in Section B show that the proposed method can achieve good accuracy when the distribution of control points is uniform. In order to explore the influence of the number of control points on the adjustment results, an experiment on the number of control points was set up, as shown in Fig. 8. The proposed method is used to adjust the number of different control points, and the coordinate residuals of the check points are counted. The results are shown in Fig. 9.

As shown in Fig. 9, the optimal number of control points for this data set was 13, and the geometric model of the adjustment was then not sensitive to control point number. With the increase of control point number, the RMSEs decreased,



FIGURE 8. Adjustment tests with different control points numbers. (a) to (r) represent the number of control points varying from 5 to 22, respectively.

and achieved a smaller value and remained almost unchanged when the number of control points reached 13. It can be seen that the proposed method can obtain stable accuracy with a



FIGURE 9. RMSE of all 18 tests with different GCP numbers.

small number of 13 control points, which proves the effectiveness of the method for reducing the number of control points.

IV. CONCLUSION

In this paper, we proposed a robust block-adjustment method for UAV SAR images for a UAV SAR system with a small imaging width and poor platform stability. The method was based on the RD model and was designed to solve the three orientation parameters of the antenna phase center initial imaging moment, Doppler centroid, and proximity delay for block adjustment by analyzing the parameter correlation of the seven-orientation parameters. This method considers that the initial position adjustment solution is inclined to compensate for each other in the range and vertical directions. We used UAV SAR images of the Dengfeng area for the adjustment experiments. Subsequently, we compared the adjustment methods with different orientation parameter settings. The proposed method exhibited the highest accuracy and efficiency and was most likely to achieve the conditions for convergence. By comparing the adjustment results of different control-point layout plans, the proposed method achieved an adjustment accuracy better than 1 m with a small number of control points under the condition of a uniform layout plan and had certain practical value in some difficult situations. The robustness and effectiveness of the proposed method were confirmed.

Due to practical limitations, the experimental area was mainly concentrated in flat areas and hills, and experimental verification of this method in areas with highly undulating terrain was limited. Based on experience, the adjustment accuracy of the proposed method will be relatively low in areas with highly undulating terrain; however, the specific value still requires experimental verification. The distribution of control points in the experiment was not ideal, and sparse control points in some areas adversely affected the adjustment solution. However, this issue should be addressed in future studies.

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