# High-Resolution 160-GHz Imaging MIMO Radar Using MMICs With On-Chip Frequency Synthesizers

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Abstract—A high-resolution frequency-modulated continuous wave imaging radar for short-range applications is presented. A range resolution of about 1 cm is achieved with a bandwidth of up to 16 GHz around 160 GHz. In order to overcome losses and large tolerances on a printed circuit board (PCB), eight coherently coupled monolithic microwave integrated circuits (MMICs) are used, each with one transmit and receive antenna on-chip and each representing a single-channel radar system. The signals on the PCB are below 12 GHz, which facilitates fabrication and enables a design with low-cost substrates. The MMIC comprises a phase noise (PN)-optimized architecture with a fully integrated on-chip frequency synthesizer. Due to partly uncorrelated PN between the frequency synthesizer components, the noise level is increased in bistatic radar measurements between two different MMICs, which is explained by a thorough PN analysis. Timedivision multiplexing is used to realize a multiple-input multipleoutput system with a virtual array of 64 elements and an angular resolution better than 1.5° for the designed array. The positioning tolerances of the MMICs are included into the design resulting in a robust array design. The high-resolution radar performance is proven by imaging radar measurements of two exemplary scenarios.

*Index Terms*—Frequency-modulated continuous wave (FMCW) radar, imaging radar, mm-wave radar, multiple-input multiple-output (MIMO) radar, phase noise (PN), sparse antenna arrays.

# I. INTRODUCTION

**R**ADAR sensors are widely used for industrial, medical, security, and automotive applications [1]. There is a growing demand for high-resolution short-range imaging radar sensors measuring range, velocity, and angle in harsh environments. This leads to more information about the scene

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TABLE I OVERVIEW OF mm-WAVE IMAGING RADAR SYSTEMS

Source	$f_{\max}$	AoC	# Virt. ch.	BW	$\Delta \varphi / \Delta \theta$
[11]	77 GHz	no	16	2 GHz	12°/-
[12]	77 GHz	no	256	2 GHz	$6^{\circ}/6^{\circ}$
[13]	79 GHz	no	16	7 GHz	$2^{\circ}/2^{\circ}$
[14]	100 GHz	no	484	10 GHz	$0.3^{\circ}/0.3^{\circ}$
[15]	120 GHz	no	16	5 GHz	$3^{\circ}/21^{\circ}$
[16]	140 GHz	yes	12	20 GHz	$5^{\circ}/-$
This work	160 GHz	yes	64	16 GHz	$1^{\circ}/-$

under investigation, but requires a high absolute bandwidth and, therefore, pushes frequencies above 100 GHz.

The progress in the low-cost silicon–germanium (SiGe) technology provoked the development of radar monolithic microwave integrated circuits (MMICs) based on the frequency-modulated continuous wave (FMCW) principle. Above 100 GHz, it is desirable to integrate voltage-controlled oscillators (VCOs), amplifiers, mixers, and couplers on the MMIC. By additionally integrating antennas on-chip (AoCs), lossy and complicated high-frequency transitions to printed circuit board (PCB) antennas are avoided. In the past years, several single-channel radar systems fully integrated on an MMIC with AoCs were presented [2]–[7].

Imaging radars are well-known and numerous ones have been introduced [8]–[13]. Table I is ordered by the maximum operating frequency  $f_{\text{max}}$  and gives an overview of already presented mm-wave FMCW imaging radar systems and their theoretical angular resolution above 77 GHz. In [14]–[16], radar systems at 100 GHz or above are presented.

In [14] and [15], imaging radar systems are realized using an optimized 2-D sparse array. The system concept is based on separate transmit (TX) and receive (RX) modules with a frequency multiplier approach. The local oscillator (LO) is derived from a single phase-locked loop (PLL). The antennas are realized on a PCB due to the considerably lower frequency range at approximately 100 and 120 GHz in comparison with the 160 GHz in this paper.

In addition, Table I shows that only [15] and [16] are realized distinctly above 100 GHz. The system architecture

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Fig. 1. Block chart of the realized RF front end with the functional principle of a single-channel radar MMIC. Each MMIC is switched ON with a TXEN.

in [16] is based on a frequency multiplier with factor 8 and is comparable to [14] and [15], but it uses AoCs.

By combining several single-channel MMICs to one system, a 160-GHz imaging radar can be realized. The full integration of the whole radar system on the MMIC including the antennas results in antenna spacings of several wavelengths. In addition, the antenna array at frequencies above 100 GHz is sensitive to positioning tolerances, which demands to include them into the array design process. In comparison with previous systems, this paper contains two new key aspects.

First, the MMIC comprises a radar system with a phase noise (PN) optimized system architecture for single-channel applications. This paper discusses whether this architecture is also advantageous for multichannel applications. The investigation is based on a thorough PN analysis. Second, this paper presents an optimization approach allowing to integrate manufacturing tolerances into the array design.

This paper is organized as follows. After describing the system setup, a thorough PN analysis of the radar system is presented and proven by measurements in Section II. An adapted array design process including manufacturing tolerances is discussed in Section III. The limits concerning the angular performance of the realized radar system are studied in Section IV. Finally, the imaging capabilities of the developed radar are demonstrated in Section V in two typical nonstationary scenarios.

#### **II. SYSTEM ARCHITECTURE**

# A. RF System Architecture

The MMIC used for the multichannel imaging radar is described in [7]. The architecture is optimized for a low PN in a single-channel radar measurement. Therefore, it uses a low-frequency multiplier with factor 4 for the FMCW ramp oscillator (RO) and a low PN LO for the upconversion to the operating frequency at 160 GHz. The fed-in ramp signal from 8-to-12 GHz is multiplied by 4 to 32-to-48 GHz on the MMIC itself (see Fig. 1). The ramp signal from 8-to-12 GHz can be realized with an off-the-shelf fractional-N PLL and an external wideband VCO with a low PN.

The LO signal is generated on each MMIC by an on-chip integer-N PLL, which is used for upconversion of the ramp signal. The reference frequency of the on-chip PLL is a surface-acoustic wave (SAW) oscillator at 916 MHz with a PN



Fig. 2. Measured radiation pattern for the realized spherical-convex lens antenna at 154 GHz.

of -150 dBc/Hz at 100-kHz offset frequency. This frequency is increased by a factor of 128 resulting in an LO frequency of about 117 GHz. It is already shown in [7] that the PN density of the realized on-chip integer-N PLL is comparable to the swept fractional-N PLL (RO) due to the high-frequency reference and a VCO with a narrow bandwidth. Therefore, the LO only marginally contributes to the overall PN of the TX signal for offset frequencies between 100 kHz and 1 MHz.

The ramp signal from 8-to-12 GHz, multiplied by 4, is upconverted with the fixed-frequency LO to the radio frequency (RF) band from 149-to-165 GHz. The upconverted signal is splitted and fed to the TX and the RX path. Two dielectric resonator antennas (DRAs) are used, which are excited with short-circuited quarter-wavelength patches [7].

The single-channel MMIC is coherently extended to an eight-channel imaging radar system as depicted in Fig. 1. This is realized by distributing the ramp signal and the SAW reference to all eight radar MMICs. The antennas are aligned in a row in the *E*-plane (*xy* plane), and a dielectric spherical-convex lens made of Teflon [polytetrafluoroethylene (PTFE)] with  $\varepsilon_r$ =2.1, a diameter of  $D_L$ =38.6 mm, and a focal distance  $d_f = 2$  cm is used to focus the radiation pattern in the *H*-plane (*yz* plane). As the antennas are in one row, no gain reduction due to a parallax of an imperfect focusing lens has to be taken into account [17]. The measured radiation pattern of the lens antenna at the center frequency  $f_c = 154$  GHz is shown in Fig. 2. The lens design has a 3-dB-beamwidth of  $\varphi_{3dB} \approx 5^{\circ}$  in the *H*-plane, whereas no focusing is realized in the *E*-plane.

#### B. Noise Mechanism in Radar Systems

The noise level in the sampled intermediate frequency (IF) signal limits the achievable signal-to-noise ratio (SNR) and is determined by a superposition of thermal noise, quantization noise, and PN. The different noise sources are briefly discussed in general.

1) Thermal Noise: The thermal noise power density after the receiver is given by

$$w_{\text{th,in}} = k_{\text{B}}T_{\text{e}}G_{\text{RX}} = k_{\text{B}}(F-1)T_{0}G_{\text{RX}}$$
(1)

with the receiver noise figure *F*, Boltzmann's constant  $k_{\rm B}$ , the receiver noise temperature  $T_e$ , the receiver gain  $G_{\rm RX}$ , and the temperature  $T_0=290$  K [18].



Fig. 3. (a) Block chart of the PLL components affecting the PN of the TX signal. (b) Measured PN densities before and after the upconverting mixer.

2) *Quantization Noise:* The IF signal is digitized with an analog-to-digital converter (ADC). The maximum achievable SNR due to a quantization is dependent on the number of bits k and is given by SNR<sub>max</sub> =  $(6.02 \cdot k + 1.76)$  dB [19].

3) Phase Noise: For most of the PLL components like the loop filter (LF), the VCO, the phase–frequency detector (PFD), as well as the charge pump (CP), the thermal noise and the flicker noise are the main contributions to an emerging PN at the output of the frequency synthesizer [20]. For the following derivation, it is assumed that the PN sources are uncorrelated and can be linearly added up [21]. The PN is determined by

$$\mathcal{L}_{\Phi,\text{PLL}}(f) = \mathcal{L}_{\Phi,\text{Ref}}(f) + \mathcal{L}_{\Phi,\text{LF}} + \mathcal{L}_{\Phi,\text{VCO}}(f) + \mathcal{L}_{\Phi,\text{CP}}(f) + \mathcal{L}_{\Phi,\text{CP}}(f)$$
(2)

with the single sideband PN density  $\mathcal{L}(f)$ . The subscript Ref refers to the reference oscillator of the PLL. If the same LO is used for the TX and the RX signal path, the range correlation effect occurs [22], dependent on the time delay  $\tau$ within the radar channel. The residual PN density in the RX output spectrum  $\mathcal{L}_{\Delta\Phi}(f)$  can then be calculated by [22]

$$\mathcal{L}_{\Delta\Phi}(f) = 2\mathcal{L}_{\Phi}(f)(1 - \cos(2\pi f\tau)). \tag{3}$$

In (3),  $\mathcal{L}_{\Phi}$  describes the PN density of the TX signal, and *f* is the frequency offset from the carrier.

An increase in the noise level due to residual PN occurs locally in the vicinity of a very strong target. This reduces the detection sensitivity of weak targets. For normal radar operation, the leakage signal from the TX to the RX is a strong target, which is on the orders of magnitude larger than the received signal passing through the radar channel to the real target [23]. Typical leakage paths are transmit– receive coupler leakage, antenna reflections, or reflections at a radome or lens.

# C. Phase Noise Model of the 160-GHz Radar

The components affecting the PN of the TX signal are shown in Fig. 3(a). At the upconverting mixer, the PN densities of the multiplied ramp signal  $\mathcal{L}_{\Phi,RO,\times4}$  and the LO signal  $\mathcal{L}_{\Phi,LO}$  are linearly superimposed [24]. Fig. 3(b) shows the measured PN densities before ( $\mathcal{L}_{\Phi,LO}$ ,  $\mathcal{L}_{\Phi,RO,\times4}$ ) and after ( $\mathcal{L}_{\Phi,TX,RX}$ ) the upconverting mixer. The PN of the LO is



Fig. 4. Possible leakage paths in the multichannel radar.

measured after the divider output (Divout) and the PN of the ramp signal before the multiplier input. Afterward, they are deteriorated according to their frequency multipliers of 4 and 128 by 12 and 42 dB, respectively, and added up.

The decisive leakage paths to be considered for PN are shown in Fig. 4. One is the direct leakage between the two on-chip antennas and the other one is due to the reflection at the lens surface.

In a monostatic radar measurement, i.e., from  $MMIC_i \rightarrow MMIC_i$ , with  $i \in 1, ..., 8$ , the same LO is applied for both the transmission and the reception paths. This means that the range correlation effect occurs. The on-chip leakage path has a delay of 33 ps, which results in a PN cancellation factor of 74 dB at 1-MHz offset frequency [25].

For the lens leakage path, an upper bound of the maximum power coupling to the adjacent receive antennas can be approximated by the radar equation for extended targets. In addition, it is assumed that the DRA gain is 6 dBi and that the MMIC output power is  $P_{out}=-1$  dBm. By using Snell's refraction law, the ratio of the reflected wave intensity at the plane dielectric surface can be approximated [26]. For a TEM wave impinging perpendicularly on the air-Teflon interface, about 4% = -14 dB of the power is reflected and couples with a delay of 160 ps and a PN cancellation of 60 dB into the adjacent receivers. A summary of the noise budgets for the monostatic measurements and the system parameters is given in Table II. In addition, it contains the residual PN of both the lens and the DRA leakage for the frequency offset f=1 MHz. They are about 40 dB below the thermal noise power density. Hence, the SNR in the monostatic radar measurement is not degraded due to a residual PN within the leakage signal.

For a bistatic radar measurement from  $MMIC_i \rightarrow MMIC_j$ , with  $i, j \in 1, ..., 8, i \neq j$ , only the ramp and reference signals are the same for the transmitting and the receiving MMICs.

TABLE II NOISE BUDGETS FOR THE MONOSTATIC RADAR MEASUREMENT AT 1-MHz OFFSET FREQUENCY

TX power	-1  dBm
DRA gain	6 dBi
RX gain	-2  dB [7]
RX noise figure	25 dB [7]
Thermal noise power density	-151 dBm/Hz
TX signal PN density $\mathcal{L}_{\Phi,TX,RX}$ (1 MHz)	−87 dBc/Hz
Measured on-chip leakage attenuation $p_{i,i}$ {1} PN cancellation factor DRA leakage Residual PN power density DRA leakage	29 dB [25] -74 dB (33 ps) [25] -193 dBm/Hz
Lens leakage attenuation (radar eq. + Snell) PN cancellation factor lens leakage Residual PN power density lens leakage	51 dB -60 dB (160 ps) -201 dBm/Hz

Thus, the range-correlation effect is only valid for the ramp and the reference signal. It can be assumed that the residual PN density of both the signals is below the thermal noise power density as in the monostatic case.

In comparison with the monostatic case, different PLL components, such as LF, VCO, N-divider, PFD, and CP are involved in the bistatic system operation. Some of the components are affected by the common SAW reference oscillator at 916 MHz. Especially the PFD and the CP pulses are dependent on the common reference frequency. Therefore, there are PLL components with perfect correlation, partial correlation, and with no correlation. The common reference is correlated as it is the same for all on-chip PLLs. The PFD and the CP are partially correlated due to the direct influence of the reference oscillator on the PFD and CP cycle times [see Fig. 3(a)]. Components like the VCO, the LF, and the dividers are fully uncorrelated as there is no relation to the reference oscillator. The PN density at the receiver output spectrum in a measurement from  $MMIC_i \rightarrow MMIC_j$  can be written as

$$\mathcal{L}_{\Delta\Phi_{j,i}}(f) = \mathcal{L}_{\varphi,u_i}(f) + \mathcal{L}_{\varphi,u_j}(f) + 2(\mathcal{L}_{\varphi,\text{Reftramp}}(f) + \mathcal{L}_{\varphi,\text{PLL,corr}}(f)) \times (1 - \cos(2\pi\tau_{i,i}f)).$$
(4)

In (4),  $\mathcal{L}_{\varphi,u_{i,j}}(f)$  describes the PN density, which results from the uncorrelated PLL components (LF, VCO, and N-divider) and the uncorrelated parts of the PFD and CP in MMIC<sub>i</sub> and MMIC<sub>j</sub> due to their individual thermal and flicker noise.  $\mathcal{L}_{\varphi,\text{Ref+ramp}}(f)$  is the PN density of the reference oscillator and the ramp signal, which is the same for all MMICs and hence correlated, whereas  $\mathcal{L}_{\varphi,\text{PLL,corr}}(f)$  is the PN profile resulting from the correlated parts within the PFD and the CP.

To prove the concept of partial correlation in frequency synthesizers using the same reference, the divider outputs of the on-chip frequency synthesizers of two individual MMICs with the same reference are investigated with the measurement setup as given in Fig. 5(a). One of the signals is upconverted using a high-quality oscillator with a superior PN performance  $\mathcal{L}_{\Phi,\text{XCO}}$  in comparison with the PN of the divider outputs ( $\mathcal{L}_{\Phi,\text{Divout1,2}}$ ) of the on-chip PLLs. That means that the PN of the divider output is dominating and therefore mainly present after the upconverting mixer. This signal is afterward



Fig. 5. (a) Block chart of the measurement setup. (b) Measurement results of the residual PN in multiple frequency synthesizers with the same reference.



Fig. 6. Measured PN densities for the RO with and without reduced CP current and for the LO with and without reduced power-supply voltage.

downconverted, and the residual PN can be measured by using a signal source analyzer (SSA).

As can be seen in the measurement in Fig. 5(b), the PN is improved, which proves that there are correlated components due to the common reference input. Due to the uncorrelated PLL components in two different MMICs, it might happen that the residual PN density at the receiver output supersedes the thermal noise power density resulting in an increase of the system noise level.

The theory is now proven by radar measurements. Thus, the PN of both the ramp signal and the LO signal is deteriorated as depicted in Fig. 6. For the ramp signal, this is achieved



Fig. 7. Noise level for both the monostatic and the bistatic radar cases in comparison. (a) Noise level as a function of the ramp input and LO input PN performance. (b) Noise level dependent on the RX antenna distance. The noise level is dependent on the LO PN performance for the bistatic case. Furthermore, the noise level is mostly increased for the RX antennas spaced next to the active TX antenna as they are affected by the strongest leakage power. Therefore, the SNR for short-range targets is degraded in 56 of 64 virtual channels.

by a reduction in the CP current from 2.8 mA to 100  $\mu$ A. For the LO, this can be obtained by a reduction in the supply voltage from 1.8 to 1.5 V. Afterward, radar measurements are conducted and statistically evaluated by the standard deviation in the IF signal spectrum over 100 ramps and for each TX channel. The standard deviation is a measure of the noise level within the IF signal.

Fig. 7 shows the noise level—derived from the standard deviation—for both the monostatic and the bistatic radar cases in comparison. For the monostatic case, the noise level is not deteriorated due to a decreased PN performance of both the ramp and the LO signals. In the bistatic radar case in comparison, the noise level is increased by about 7 dB at an IF frequency  $f_{\rm IF} = 1$  MHz as depicted in Fig. 7(a). However, the bistatic radar case is not affected by a deterioration in ramp signal PN of approximately 25 dB [see Figs. 6 and 7(a)]. By deteriorating the PN performance of the LO, the standard deviation and therefore the noise level increase for a low IF in the bistatic case. This can be explained by the uncorrelated PN caused by the different on-chip PLLs.

This effect also holds for the other receive channels and depends on the MMIC distance among each other as can be seen in Fig. 7(b) for all eight virtual channels belonging to  $TX_2$ . The larger the RX antenna distance compared with the active TX antenna (see Fig. 8), the less power couples to it. For the largest leakage power, the noise level is increased by approximately 10 dB. Therefore, the increase in noise level is larger for the adjacent MMICs. The measurements show that RX<sub>3</sub>, which is next to TX<sub>2</sub>, perceives the largest increase in noise level.

# **III. ANTENNA ARRAY OPTIMIZATION**

Multiple-input multiple-output (MIMO) radars are typically realized with uniform linear arrays (ULAs) and an interelement spacing of  $\lambda/2$ . This approach has a maximum sidelobe level (SLL) suppression of -13.3 dB and avoids grating lobes within the angular range  $\pm 90^{\circ}$  [27]. By using fully integrated



Fig. 8. Realized array geometry. The MMICs are marked by rectangles.

radar MMICs with AoCs, it is usually not possible to realize an interelement spacing of  $\lambda/2$ , because PCB and MMIC interconnect constraints have to be fulfilled. In the following section, an optimization criterion, which considers an MMIC spacing of several wavelengths including manufacturing tolerances, is derived.

# A. Boundary Conditions

The above-mentioned boundary conditions concerning the alignment of the radar MMICs to realize the  $8 \times 8$  MIMO radar result from the following reasons. The MMICs comprise a TX and an RX antenna with a fixed antenna distance of  $0.87 \lambda$  onchip. In addition, the MMIC has different digital interconnects and two RF signals like the 916-MHz SAW reference and the ramp signal, which have to be fed to the MMIC. Due to PCB routing constraints, the MMICs can only be placed with a distance of several wavelengths next to each other. The spacing constraints were experimentally determined and are as follows.

1)  $TX \leftrightarrow TX: 4.1 \text{ mm} = 2.2 \lambda.$ 

2)  $RX \leftrightarrow RX$ : 2.8 mm = 1.5  $\lambda$ .

3)  $TX \leftrightarrow RX$ : 3.6 mm = 1.9  $\lambda$ .

Nevertheless, it is possible to achieve interelement antenna spacings of less than  $\lambda/2$  within the virtual array (see Section III-C).

# B. Design Criterion and Optimization Algorithm

The array alignment is optimized using a genetic algorithm (GA). The implemented algorithm is adapted from [28] and [29]. The allowed MMIC positions are discretized with a grid of 0.1  $\lambda$  within the simulation. Within this

grid, the TX antenna position is chosen randomly, and the corresponding RX antenna is placed 0.87  $\lambda$  with respect to the TX antenna. Afterward, the agreement to the constraints in Section III-A is verified.

In order to rank and compare the resulting array configuration, a maximum allowed array size and a minimum required angular resolution has to be specified. It is defined with the Rayleigh criterion [30]

$$\Delta \varphi = 1.22 \frac{\lambda}{d_{\rm V}} \quad . \tag{5}$$

The angular resolution in radiant is described by  $\Delta \varphi$ , and  $d_V$  is the virtual aperture size of the antenna array. For a desired angular resolution of 1°, the necessary virtual aperture size is  $d_V \approx 70 \ \lambda$  resulting in a physical aperture size of  $d_A \approx 35 \ \lambda$ .

The used DRAs provide a measured 6-dB beamwidth of about  $\pm 40^{\circ}$  [7]. Therefore, the fitness function  $f_n$  to rank the found arrays is evaluated within this angular range by

$$f_{\rm n} = \frac{1}{\text{ambiguity-free region}} \text{SLL}_{\rm max}$$
 . (6)

The ambiguity-free region is defined as the angular range for which the SLL is below 0.5=-6 dB. The ambuiguity-free region is evaluated by the ambiguity function (AF) [31], [32], which is defined by

$$AF(\varphi_i, \varphi_j) = \frac{|\mathbf{v}(\varphi_i)^H \mathbf{v}(\varphi_j)|}{\| \mathbf{v}(\varphi_i) \| \| \mathbf{v}(\varphi_j) \|} .$$
(7)

In (7),  $(\cdot)^H$  denotes the Hermitian operator,  $\| \cdot \|$  is the Euclidian norm of a vector, and  $\mathbf{v}(\varphi) = \mathbf{a}(\varphi) \otimes \mathbf{b}(\varphi)$  is the virtual steering vector, which is given by the Kronecker product  $\otimes$  of the transmitter and receiver steering vectors  $\mathbf{a}(\varphi)$  and  $\mathbf{b}(\varphi)$  [33].

In comparison with [29], the fitness function  $f_n$  is extended by a weighting with the maximum SLL<sub>max</sub> of the antenna pattern. Thus, arrays with a lower SLL get a higher rank resulting into more robust arrays.

# C. Robustness Analysis

At mm-wave frequencies around 160 GHz, the ratio between positioning tolerances and the wavelength is significant and cannot be neglected anymore. In order to realize a robust antenna design without ambiguities, it is necessary to include these tolerances into the array design process. For each array fulfilling the constraints, a Monte Carlo analysis consisting of 1000 randomly modified antenna arrays is added to the optimization algorithm. The antenna positioning variance is modeled by a random Gaussian process with a standard deviation of 50  $\mu$  m. The robustness is defined by the percentage of array realizations with full ambiguity-free region in the angular range  $\pm 40^{\circ}$ . The final array configuration has a simulated maximum SLL<sub>max</sub>=-9 dB. For the found antenna array, all the performed Monte Carlo simulations fulfilled the robustness criterion.

Figs. 8 and 9 show the realized array and the corresponding virtual antenna array. The realized antenna positions are measured using a measurement microscope. The values are summarized in Table III and show that the design specifications are met.



Fig. 9. Simulated virtual array. The virtual RX channels are marked by crosses. Several interelement spacings are lesser than  $\lambda/2$ .

TABLE III Overview of Measured Antenna Array

N <sub>{T,R}</sub> (#)	x <sub>R,sim</sub> (mm)	x <sub>R,phys</sub> (mm)	$\Delta x_{\rm R}$ (mm)	x <sub>T,sim</sub> (mm)	$x_{ m T,phys}$ (mm)	$\Delta x_{\mathrm{T}}$ (mm)
1	0	0	0	1.63	1.68	0.05
2	5.87	5.82	-0.05	7.50	7.50	0
3	22.17	22.15	-0.02	23.80	23.78	-0.02
4	34.98	34.98	0	33.35	33.33	-0.02
5	41.35	41.35	0	39.72	39.72	0
6	49.71	49.71	0	51.34	51.36	0.02
7	54.96	54.95	-0.01	56.59	56.59	0
8	65.16	65.13	-0.02	63.52	63.50	-0.02

# D. Array Calibration and Measured Array Performance

Phase and amplitude mismatches between the channels require a calibration. Each virtual channel is formed by a  $TX_k$ -RX<sub>l</sub> combination. For a single target in the far-field of the antenna array, the phase progression  $\Delta \phi_{kl}$  of the (k, l)th virtual antenna element relative to the (1, 1)th virtual antenna element behaves according to

$$\Delta \phi_{kl}(\varphi) = \frac{2\pi}{\lambda} \Delta x_{kl} \sin(\varphi) + \phi_{kl} \tag{8}$$

with the relative virtual antenna distance  $\Delta x_{kl} = x_{kl} - x_{11}$  concerning the virtual receive antenna (1, 1) and the constant phase offset  $\phi_{kl}$ .

Due to manufacturing tolerances, the ideal antenna positions from the simulations slightly differ from the physical realized ones (see Table III).

The array is calibrated in the angular range  $-60^{\circ} \le \varphi \le 60^{\circ}$  with a step size of 0.1°. Fig. 10 shows the measured relative phase progressions and the corresponding target power in dBm as a function of  $\sin(\varphi)$ .

From the slopes of the curves and by applying (8), the electrical virtual antenna positions coinciding with the electrical antenna behavior are determined. Mutual coupling and other nonidealities shift the physical antenna positions resulting in a larger deviation between ideal and electrical antenna positions (see Fig. 11). From now on, the measured electrical virtual antenna positions are used. For the simulated antenna array, there are 51 unique antenna array positions, whereas for the realized array, 64 virtual antenna elements are obtained due to manufacturing tolerances and coupling effects.

With the measured virtual antenna array steering matrix from Fig. 10(a) and (b), the AF in (7) can be calculated (see Fig. 12). There are no ambiguities within the azimuth angular range of  $\pm 60^{\circ}$ .

#### IV. IMAGING PERFORMANCE

# A. Measurement System

The radar system is realized in a modular PCB stack, with each functional part on a separate PCB to achieve modularity



Fig. 10. (a) Measured relative phase progression and (b) amplitude variations for one target placed at a distance of R=5 m exemplarily shown for the virtual channels belonging to RX<sub>8</sub>.



Fig. 11. Deviation of electrical from ideal virtual RX positions.



Fig. 12. Measured AF. The array has a full ambiguity-free region (<0.5) within [ $-60^\circ$ ,  $60^\circ$ ].

and to facilitate error diagnostics. The radar stack is subdivided into an RF PCB comprising the MMICs and the RF signal lines below 12 GHz, a PLL PCB for the ramp generation, an IF PCB with a variable amplification from 34 to 52 dB for eight differential channels, and a power supply PCB with a microcontroller ( $\mu$ C) as depicted in Fig. 13.

#### **B.** Phase Ambiguities

The used integer-N PLLs ensure that the relative output phases between the different VCOs are deterministic. Consequently, a phase synchronization is avoided and the underlying



Fig. 13. Photographs of (a) modular radar stack and (b) RF PCB zoomed-in view of an MMIC.

TABLE IV

O	VERVIEW	OF THE	RADAR	Settings
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Ramp duration T	100 µs
Ramp repetition time $T_r$	150 µs
Ramp start frequency $f_{\text{start}}$	8 GHz
Ramp stop frequency $f_{\text{stop}}$	10.5 GHz
RF bandwidth B	10 GHz
Center frequency $f_c$	154 GHz
Number of chirps $N_{\rm c}$	512

system architecture is capable of imaging. In comparison, fractional-N PLLs result in phase ambiguities [21].

# C. Angular Resolution

In order to demonstrate the imaging capabilities, radar measurements are conducted with the parameters summarized in Table IV. The angular resolution for the case of two targets with the same image intensity can be determined by (5). For the realized array and for the operated center frequency  $f_c=154$  GHz, the calculated angular resolution is 1.06°.



Fig. 14. Measured angular estimation performance of the realized antenna array for (a) one target at  $\varphi = -15^{\circ}$  and (b) two targets with an angular separation of  $\Delta \varphi = 1.5^{\circ}$ .

The measurements are evaluated using the maximum likelihood (ML) method, which correlates the measurement vector with the interpolated calibration vectors. Fig. 14(a) shows one single target at the angle  $\varphi = -15^{\circ}$ , whereas Fig. 14(b) shows the separability performance of two targets of the same image intensity and with an angular separation of 1.5°. Both the targets can be separated with a notch of 1.2 dB. The angular resolution of 1.5° is close to the theoretical limit of 1.06°.

# D. Time-Division Multiplexing

The radar system uses time-division multiplexing (TDM) to transmit orthogonal waveforms for MIMO. This reduces the maximum detectable Doppler frequency  $f_{D,max}$  by the number of transmit antennas, because the Doppler frequency  $f_D$  is only sampled by every eighth transmitted ramp [27]. Furthermore, an additional phase difference of  $\Delta \phi_{TDM}=2\pi f_D T_r$  due to the switching time occurs, which is corrected by an adapted discrete Fourier transform as in [34]. The time  $T_r$  denotes the ramp repetition time between two consecutive ramp segments.

#### V. RADAR MEASUREMENTS OF EXTENDED TARGETS

To evaluate the radar performance in realistic scenarios, two well-defined nonstationary scenarios are investigated. The first scenario consists of four metallic cylinders with diameter 1.6 cm mounted on a wooden plate with the dimensions 25 cm ×19 cm  $(l \times w)$ , whereas the second scenario is a metallic cuboid with the dimensions 45 cm×30 cm×30 cm  $(l \times w \times h)$ . Both the targets are rotating with a radial velocity  $v_r=50^{\circ}/s$ around their vertical axis. The measurement scenarios are depicted in Fig. 15. The virtual channels are noncoherently integrated. After separating the targets in the range–velocity domain (*R*-*v*-domain), an ML angle estimation is performed and the maximum value within the angular spectrum is determined and marked in an *x*-*y* diagram.

#### A. Scenario 1: Metallic Cylinders

The power range profile of the measured cylinders is depicted in Fig. 16. The results of the measurements are shown in Fig. 17. The orientation of the four cylinders and the



Fig. 15. Photographs of the measured scenarios. (a) First scenario: four metallic cylinders are mounted on a wooden plate and rotated around its vertical axis. (b) Second scenario: cylinders are replaced by a metallic cuboid. The rotation axis and the direction are indicated with red.



Fig. 16. Power range profile of scenario 1. The large absolute bandwidth enables to separate the cylinders in range.

distance to each other can be estimated with high accuracy. It is possible to separate the four targets independently both in range and velocity [see Figs. 16 and 17(a)]. The noise level is increased at v=0 m/s due to the decreased PN performance within the bistatic channels [see also Fig. 7(a)].

# B. Scenario 2: Metallic Cuboid

The measurements in Fig. 18 show that the dynamic range of the radar enables to detect diffracting sharp edges. The x-y diagram in Fig. 18(b) shows that it is even possible to determine the orientation of the cuboid assuming a prior knowledge about both side lengths.



Fig. 17. Imaging results consisting of (a) R-v diagram and (b) x-y diagram for the four rotating metallic cylinders around its axis. The orientation of the four cylinders can be determined.



Fig. 18. Imaging results consisting of (a) R-v diagram and (b) x-y diagram for a rotating metallic cuboid. Both edges and two direct reflections at the surface are measured and can be independently separated in range and velocity.

The measurement results in Figs. 17 and 18 verify the good imaging capability of the realized radar system.

# VI. CONCLUSION

In this paper, a 160-GHz high-resolution imaging radar system with simple single-channel radar MMICs at 160 GHz is shown. The MMICs use an on-chip PLL for frequency synthesis. A corresponding PN analysis is presented and proven by radar measurements. A monostatic radar measurement is limited by thermal noise, and the bistatic radar measurements are affected by the PN occurring due to the lens leakage, which increases the system noise level. This effect is dependent on the power coupling to the receivers and is more distinct for the adjacent receive antennas. In addition, a very robust array design including manufacturing tolerances is proposed resulting in a highly sparse antenna array with sidelobes below -9 dB, an ambiguity-free region of  $\pm 60^{\circ}$ , and a high angular resolution of 1.5°. The high-resolution imaging capabilities of the realized radar system are demonstrated using two exemplary nonstationary scenarios with the possibility to separate the targets independently in range and velocity.

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